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Monterey, California



THESIS

MARINE STEAM CONDENSER
DESIGN OPTIMIZATION

by

Thomas M. Buckingham

December 1983

Thesis Advisor:

R. H. Nunn

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Marine Steam Condenser
Design Optimization

by

Thomas M. Buckingham
Lieutenant, United States Navy
B.A., College of the Holy Cross, 1977

Submitted in partial fulfillment of the
requirements for the degree of

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December 1983

ABSTRACT

A surface-condenser analysis code was coupled with a constrained function minimization code to produce an automated marine condenser design and optimization package. The program, CONDIP, was based on the principles developed in ORCON1, a sophisticated computer code produced by the Oak Ridge National Laboratory. CONMIN, the optimization program, was developed at the Ames Research Center.

CONDIP is an extremely versatile design tool, incorporating a detailed analysis of the complex steam-side thermodynamic processes occurring at each row in the condenser. The additional capability of tube enhancement is also included. However, in coupling CONDIP with CONMIN numerous problems had to be overcome in order to make CONDIP capable of completing an analysis even when thermodynamic conditions in the condenser became infeasible. This had to be accomplished while ensuring continuity in all constraint and objective function evaluations. A series of test cases were conducted to evaluate and compare the importance of various objective functions and design criteria.

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I. INTRODUCTION

A. BACKGROUND

For many years the steam plant was unchallenged in its role as the primary type of marine propulsion systems. But recently gas turbines have become a desirable alternative despite the fact that they are less efficient than comparable steam plants. The primary advantage of gas turbines is their light weight and compact size. Thus, in order for the marine steam plant to survive, it is imperative that lighter, more compact and efficient steam plants be developed.

While there are numerous advanced concepts in all areas of steam propulsion which can be explored, one simple way to streamline the steam plant is the elimination of overdesign. Most overdesign is due to unnecessary safety factors used to offset lack of detailed knowledge about the thermal processes in the plant. Identification of the minimum safe design could significantly reduce unnecessary overdesign and result in the development of a smaller, more compact power plant.

B. METHCDOLOGY

In the United States the most prevalent criterion for the design and specification of surface condensers is based on the "square-root of V" relationship, as developed by the Heat Exchange Institute (HEI) [Ref. 1]. The HEI method was adopted by the Department of the Navy Bureau of Ships (now Naval Sea System Command) for specification of U.S. Naval condensers.

The HEI method is very simple in its approach, calculating the overall heat transfer coefficient as a function of cooling water velocity through the tubes, inlet coolant temperature, tube wall thickness and material, and fouling. The limitations of this method are apparent. Designs based on HEI are insensitive to shell-side conditions. Saturation steam pressure, temperature and enthalpy are assumed to be constant as steam passes through the bundle, whereas in reality there is a continual pressure drop as steam flow passes over the rows of tubes, with a corresponding decrease in saturation temperature. There is also no provision for any effects of condensate film, external tube enhancement, etc. on the shell-side of the bundle. In addition, the HEI method does not account for the presence and effect of non-condensable gases that inevitably contaminate a condenser.

With the capabilities of high speed computers now available, more comprehensive methods have been developed to account for the deficiencies of the HEI method. In particular, a radial flow computer code was developed to calculate the local heat transfer and thermodynamic properties on a row by row basis. Known as ORCON1, this code was developed by Oak Ridge National Laboratory (ORNL) under contract to the Office of Saline Water during the period from 1968-1970 [Ref. 2]. The program was based, in part, on the work performed by Eissenberg [Ref. 3]. Eissenberg's experimental results led to correction factors on the basic Nusselt equation to account for condensate inundation effects on tubes within a condenser bundle. Basically, ORCON1 divides the condenser into sectors and performs a row by row analysis within each sector, determining local heat transfer coefficients, heat flux, steam characteristics, the effect of condensate inundation and numerous other parameters at each row. ORCON1 is also capable of incorporating the effects of both tube-side and steam-side enhancement factors. Since

ORCON1 represented a much more comprehensive and detailed analysis of the condenser than the less exact HEI method, its results could be expected to be more precise.

Some work has been done at the Naval Postgraduate School to improve the capabilities of ORCON1. In his development of OPCODE2, Johnson [Ref. 5] added subroutines to ORCON1 which calculated tube-side pressure drops, corresponding pumping power and condenser volume. Nunn and Marto [Ref. 14] have incorporated the effects of vapor shear in an amended version of ORCON1 called MORCON. MORCON includes the correlations developed by Fujii [Ref. 4] to determine the effect of vapor velocity on the thermal resistance of the condensate film on the condenser tubes. In general, vapor shear effects tend to enhance the condenser heat transfer on the steam-side of the tube while condensate inundation tends to inhibit it.

The ability to represent numerically the actual thermodynamic processes occurring within the condenser has improved dramatically. However, the capability to couple these increasingly comprehensive and complex condenser design programs with an optimizing procedure has not made comparable progress. Optimization is a powerful tool which can help in reducing overdesign and achieving the goal of a safe compact condenser design.

There are currently numerous computer optimization programs available which can be coupled with general design programs of all types to numerically improve and ultimately determine the best design. The key is to properly write the design program so that it is compatible with the optimizer. Johnson [Ref. 5] developed a computer program called OPCODE1, based on the HEI method of condenser design, and was able to couple it with one such numerical optimizer. The results of OPCODE1 demonstrated how condenser designs can indeed be safely improved upon. It also revealed the

versatility of condenser design optimization as a powerful design tool. However, Johnson was unsuccessful in coupling OPCODE2 (his derivative of ORCON1) with an optimizer. This failure does not alter the fact that in order to fully appreciate more sophisticated condenser design analyses, such as that used in ORCON1, it is imperative that computer programs be developed which will be compatible with current numerical optimizers.

C. OBJECTIVE

There were two primary objectives of this thesis. The first objective was to develop a computer code which incorporates the basic condenser analysis of ORCON1 and the subsequent improvements made in MORCON and OPCODE2, but which will be capable of being coupled with a numerical optimizer to yield a complete, detailed design package. This design package can then be used as a tool in obtaining a much more reasonable conceptual design and for use in comparison studies. It would provide the naval architect the ability to optimize weight, volume, cost or any other potential design objective of the marine plant.

The second objective was to make this design package capable of determining the single best design rather than simply an improvement over the initial design. The key was to construct the program in such a way so the optimizer does not stop at some relative optimum, but continues the analysis until no further improvement can be realized. It is most desirable to be able to reach this single true optimum design regardless of initial design variable values.

II. NUMERICAL OPTIMIZATION

A. BACKGROUND

Nearly all design problems require either the minimization or maximization of a parameter. This parameter will be called the problem's objective function or design objective [Ref. 6]. For a given design to be feasible or acceptable, it must satisfy a set of design constraints which are either maximum or minimum limiting values for a pre-determined set of parameters or functions of parameters. For example, in any condenser design the outer diameter of a condenser tube can never be less than zero and there is normally some practical upper limit which also cannot be exceeded. These limits are design constraints on the tube outer diameter. In the design problem there is also a set of design variables which are parameters whose values can be changed within specified limits in order to minimize or maximize the design objective. For example, in minimizing the condenser volume an engineer may want to vary tube inner diameter, tube wall thickness and tube length. These three parameters would thus be examples of typical design variables.

For such complex design problems as the treatment of the condenser design in ORCON1, it is necessary to choose an optimization scheme which can handle the problem and provide a rational, rapid approach to design automation and optimization. An optimization program based on direct methods for solution of constrained problems [Ref. 11] was chosen for this research work.

B. CONSTRAINED FUNCTION MINIMIZATION (CONMIN)

Vanderplaats [Ref. 7] developed an optimization program, CONMIN, capable of optimizing a very wide class of engineering problems. CONMIN is a fortran program, in subprogram form, that optimizes a multi-variable function subject to a set of inequality constraints.

It is practical at this point to introduce three basic definitions and their respective conditions [Ref. 8].

DESIGN VARIABLES: Those parameters which the optimization program is permitted to change in order to improve the design. Design variables appear only on the right side of an equation, are continuous, and have continuous first derivatives.

DESIGN CONSTRAINTS: Any parameter which must not exceed specified bounds for the design to be acceptable. Design constraints may be linear or nonlinear, implicit or explicit, but they must be functions of the design variables. Design constraints appear only on the left side of equations.

OBJECTIVE FUNCTION: The parameter which is going to be minimized or maximized during the optimization process. The objective function may also be either linear or nonlinear, implicit or explicit, and must be a function of the design variables. The objective function usually appears on the left side of an equation. The only exception is if the objective function is also a design variable.

Assuming that the optimization process requires the minimization of a particular objective function the general optimization problem can be stated as:

Find the vector of design variables, \underline{X} ,
To minimize the objective function, $F(\underline{X})$,
Subject to the constraints:

$$G_j(\underline{X}) \leq 0.0 \quad j = 1, NCON \quad (\text{eqn 2.1})$$

$$VLB_i \leq \underline{X} \leq VUB_i \quad i = 1, NDV \quad (\text{eqn 2.2})$$

In the general problem, $G_j(\underline{X})$ are the constraint functions; there are NCON constraints and NDV design

variables; VLB_i and VUB_i are the lower bounds and upper bounds of the i -th design variable. If the equality condition is met, $(G_j(\underline{X})=0.)$, the constraint is active. If the inequality is met, $(G_j(\underline{X})<0.)$, the constraint is inactive. Finally, if the inequality of equation 2.1 is violated, $(G_j(\underline{X})>0.)$, that constraint is said to be violated. Because of numerical inaccuracies representing exact zero on the computer, the equality condition is represented by a band around the value $G_j(\underline{X})=(0.\pm CT)$ where CT is the constraint thickness.

Any design which satisfies the inequalities of equations 2.1 and 2.2, thus having no violated constraints, is said to be feasible. If the design violates any of these constraints it is said to be an infeasible design. The design which best minimizes the objective function while still remaining feasible is said to be optimal.

, CONMIN requires an initial set of values for the design variables \underline{X} to obtain an initial design which is either feasible or infeasible. If the initial design is feasible, CONMIN moves in a direction which will minimize the objective function. If the initial design is infeasible, CONMIN moves toward a feasible solution with minimal increase in the object function.

The optimization process proceeds in an iterative fashion. Johnson [Ref. 5] presents in greater detail the procedures utilized in CONMIN to search for the minimum objective value. In general, the methods used by CONMIN to determine search direction include the method of steepest descent, the method of conjugate direction, and the method of feasible directions. For further background concerning CONMIN and the numerical techniques utilized in optimization, consult Vanderplaats [Ref. 7], Fletcher and Reeves [Ref. 9], Zoutendijk [Ref. 10], and Vanderplaats and Moses [Ref. 12]. However, it is necessary to stress a few pertinent points which will aid in understanding how the program was developed in this thesis.

The optimization process begins by calculating the gradient of the objective function using a finite difference technique. A perturbation is applied to each of the design variables in a single forward step and the gradient vector is determined.

$$\Delta F(\underline{X}) = \begin{bmatrix} \frac{\partial F(\underline{X})}{\partial X} \\ \frac{\partial F(\underline{X})}{\partial X} \end{bmatrix}$$

The search direction is then calculated and is a function of this gradient and any active or violated constraints resulting from the applied perturbation. Subsequent search directions are a function of previous search directions, as well as current gradient information and any appropriate constraint factors. Obviously, the size of the perturbation and the size of the bandwidth about an active constraint will have a great deal of effect on the search direction and ultimate optimization process. This detail will be recalled later-on during the code development.

There are some limitations to CONMIN. The number of design variables (NDV) directly affects the computational time required to reach an optimum. Since the calculation of the gradient information required for each design variable at the beginning of each design iteration is found by using a single forward finite difference step, requiring a complete pass through the analysis portion of the program, there is an increase in CPU time as NDV increases. Also, as NDV increases, there is the corresponding rise in machine related numerical innacuracy. Vanderplaats [Ref. 6] recommends no more than twenty as a practical limit for the number of design variables.

It is quite possible that while design improvement may be obtained, the single best design optimum or true optimum

may not be reached. This is not an uncommon occurrence and there are several possible explanations. For example, the design problem may not be formulated properly or the analysis may be extremely complex and non-linear. However, a more common reason is that there are "relative optimums" between the initial design and the single true optimum. This concept of relative vs true optimum design can be better explained through an analogy. The search for the best optimum design can be likened to a blind man climbing to the top of a mountain. The blind man knows he is proceeding up the mountain by sensing the direction of ascent. However, the paths he takes may be limited by barriers or fences which will restrict the directions he can go. These fences represent constraints in the optimization problem. During the journey he may also encounter small crests and valleys. If the available paths lead the blind man up to one of these crests prior to reaching the mountain top, he will be confronted with a situation where he will sense no further rate of ascent and he will stop his journey. So although he has made progress from his initial starting point, the man did not achieve his ultimate goal of climbing the mountain.

During optimization, the search for a true optimum may proceed along a path on which the objective function assumes such relative optimum values. If the optimizer can not be made to "look beyond" these relative peaks, then the optimization will cease - at a design which may be an improvement over the initial one but short of the true optimum. This problem may be overcome by starting the design with several different initial design vectors, \underline{x} , until the same optimal design is repeated. Another alternative may be to increase the size of the finite difference so that the optimizer uses larger perturbations of \underline{x} thus looking beyond any small increases in the objective function which could stand in the way of further design progress. This second

alternative will be specifically addressed during the discussion of the code development.

C. CONTROL PROGRAM FOR ENGINEERING SYNTHESIS (COPES)

The optimizer, CONMIN, was written in subroutine form. Vanderplaats [Ref. 13] has developed a main program to simplify the use of CONMIN and aid in the design optimization process.

The user must supply an analysis subroutine called ANALIZ, which consists of three segments: input, analysis and output. COPES acts as an interface between ANALIZ and the optimizer CONMIN. Based on a flag from COPES (ICALC=1,2,3) ANALIZ performs the proper function. Figure 2.1 offers a simplified illustration of the interrelationship between COPES, ANALIZ and CONMIN.

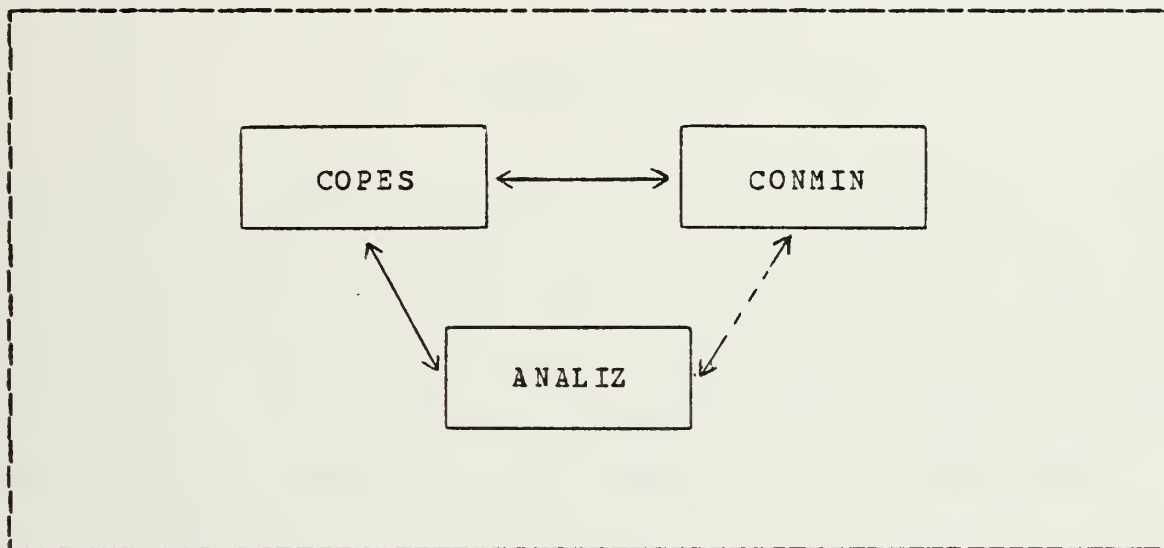


Figure 2.1 Flow Diagram For ANALIZ and COPES/CONMIN.

COPES currently provides four specific capabilities, two of which will be applied in this work:

1. Single analysis - just as if COPES/CONMIN were not used.
2. Optimization - minimization or maximization of a multi-variable function with corresponding constraints.

COPES requires certain initial data from the user in order to coordinate the optimization process. Initial values for the design variables as well optimizer control parameters are utilized by CONMIN to conduct its numerical analysis. There are a few optimizer parameters which are particularly important to the treatment of condenser designs. One is the finite difference step used in gradient calculations. Another is the normalization factor used in COPES evaluation of a constraint function. COPES utilizes the following expressions in determining constraint function violations:

$$\frac{BU - CFV}{SCAL1} \leq 0.$$

$$\frac{CFV - BL}{SCAL2} \leq 0.$$

where SCAL1 and SCAL2 are the normalization factors, BU and BL are the upper and lower limits of the constraint, and CFV is the constraint function value. It is intuitive that the normalization factor can play an important role in determining the size of the active region about a given design constraint. Both finite difference and constraint normalization will be recalled later during the code development.

The power of COPES is that it has simplified the procedures involved in using a sophisticated program such as CONMIN. The user is therefore freed from the unwanted role of systems analyst and can concentrate on the design analysis.

III. CONDENSER DESIGN IMPROVEMENT PROGRAM (CONDIP)

A. BACKGROUND

In the late 1960's, engineers at the Oak Ridge National Laboratory developed a sophisticated computer code under contract to the Office of Saline Water. This code, called ORCON1, [Ref. 2] was generated to aid in the analysis and parametric study of large, generally circular condensers. Much of ORCON1 was dependent on Eissenberg's research work [Ref. 3] on the effects of condensate rain on the shell-side convective heat transfer coefficient. Johnson [Ref. 5] took ORCON1 and made a few minor modifications to determine tube side pressure losses and volumetric calculations. Nunn and Marto [Ref. 14] further incorporated the correlations proposed by Fujii [Ref. 4] to determine the effects of shearing forces exerted by high vapor velocities on the condensate film and resulting shell-side heat transfer coefficient.

It was at this point that the development of CONDIP was begun. CONDIP was dependent primarily on the principles detailed in ORCON1 but also incorporated subsequent developments to the basic program. CONDIP was written, however, in such a way as to be compatible with the optimizer, CONMIN.

CONDIP analyzes a single or double pass, circular or semicircular condenser, with steam flowing radially inward on the shell-side of the tubes and variable salinity water flowing on the tube-side. An optional, rectangular air cooler bundle is provided-for as well as shell-side baffles. The circular bundle is normally divided into 30-degree sectors with symmetry about the central axis to reduce computational effort. Unless otherwise specified, tubes are

placed on a 60-degree equilateral, triangular pattern of concentric rows with the rows added from an inner void out to the outermost row. The void serves as a collection header for non-condensable gases prior to passage through an air cooler, if specified. As in ORCON1, CONDIP proceeds sector by sector, row by row through the condenser utilizing an average tube to represent the row segment, and calculates the following quantities in each sector:

- a) Steam pressure losses at the entrance of a sector.
- b) Total pressure of the steam/non-condensable gas mixture entering a row segment.
- c) Saturation pressure of steam entering a row segment.
- d) Saturation temperature of the steam entering a row segment.
- e) Steam flow entering the row segment.
- f) Velocity of the steam/non-condensable gas mixture at the minimum cross-section in the row segment.
- g) The fraction of non-condensable gas in the mixture by weight.
- h) The overall heat transfer coefficient for the average tube in the row segment.
- i) The steam-side condensing coefficient.
- j) The tube-side heat transfer coefficient.
- k) The shell-side film heat transfer coefficient composed of the non-condensable gas film and the condensate film.
- l) The shell-side friction factor.
- m) The shell-side pressure loss as steam passes over the row segment.
- n) The shell-side Reynolds number based on the mass flow at the minimum cross-sectional area in the row segment.
- o) The heat transfer rate per square foot of condenser tube.
- p) The mass flow rate of steam/non-condensable gas mixture at the minimum cross-section in the row segment.

- q) The mass flow of condensate produced as steam passes over the row segment.
- r) The cooling water temperature at the outlet end of the condenser.
- s) The coolant pressure loss on the tube-side.
- t) The average Reynolds number of the coolant through the tube.
- u) The heat transfer coefficient for the non-condensable gas film.
- v) The internal heat film heat transfer coefficient.
- w) The number of tubes per row segment.
- x) The cross-section area available for steam flow per row segment.
- y) The cumulative shell-side pressure drop.
- z) The LMTD based on inlet and outlet coolant temperatures and saturation temperature at each row segment.

In addition to the above parameters, the area-weighted overall heat transfer coefficients for the condenser, cooler and combined condenser are used to calculate the "back calculated" log mean temperature difference (LMTD). Steam exit-fraction, condenser volume, coolant pumping power and numerous other factors are calculated from the cumulative results of the row and sector analysis.

There are two significant contributions to the external film heat transfer coefficient which have a profound impact on the overall analysis. As mentioned earlier, Eissenberg [Ref. 3] corrected for condensate inundation effects on the external heat transfer coefficient with a series of empirical relations. He created a flooding factor F using the following relation:

$$F_n = .6F_d + (1 - .5647 F_d) n^S \quad (\text{eqn 3.1})$$

where F_d is a constant indicating the effect of tube spacing and orientation on condensate side drainage. With closely packed tubes, significant side drainage can occur in low velocity steam flow. Condensate generated on tubes above may, due to surface tension effects, proceed laterally to adjacent tubes rather than down. Thus F_d tends to approach 1.0 for closely packed, staggered tube bundles and zero for disperse bundle layouts. S is a constant ranging in value from:

$$(.07 < S < .25)$$

If the condensate rain is acting under the influence of gravity alone S approaches 0.25. But the influence of any steam velocity present begins to alter the rate and direction of condensate flow and correspondingly decrease S . Thus S is a function of vapor velocity and direction, as well as bundle geometry.

The condensate film coefficient for the average tube in the n -th vertical row is then calculated from the uncorrected heat transfer coefficient, h_o , as follows:

$$h_n = [nF_n - (n-1)F_{n-1}] h_o \quad (\text{eqn 3.2})$$

It is obvious that the determination of a corrected heat transfer coefficient is highly dependent on the choice of S and F_d in equation 3.1. S and F_d are extremely subjective constants and there does not exist a current analytical expression to determine them. Yet the choice of these constants can have profound impact on the condenser design. In CONDIP, as in ORCCN1 [Ref. 2], the following, relatively conservative values for S and F_d were used:

$$S = 0.2$$

$$F_d = 0.5$$

An additional important contribution to the external film coefficient is the effect of velocity shear forces on the condensate film. Fujii [Ref. 4] developed the following experimental correlations to correct the Nusselt number for the effect of velocity shear:

$$[Nu_m / Nu_o] = c_1 Nu_o^{(4a-1)} Re_L^{(.5-2a)} \quad (\text{eqn 3.3})$$

where Nu_m is the mean Nusselt number, Re_L is the two-phase Reynolds number (based on vapor velocity, tube outside diameter and kinematic viscosity of the condensate) and Nu_o is the standard Nusselt number for the zero shear case. The empirical constants c_1 and a lie within the following ranges:

$$1.13 < c_1 < 1.24$$

$$0.196 < a < 0.2$$

depending on how tube thermal conditions are described. In CONDIP the values for c_1 and a were:

$$c_1 = 1.24$$

$$a = 0.2$$

It should be noted that equation 3.3 is only valid in the range:

$$3.3 < (Re_L / Nu_o) < .28.$$

For smaller values of this parameter Fujii recommends the use of a slightly reduced value of the standard Nusselt number:

$$Nu_m = 0.96 Nu_o \quad (\text{eqn 3.4})$$

It is apparent that the vapor velocities commonly encountered in naval condensers can have an impact on the heat transfer coefficient.

B. CODE DEVELOPMENT

Johnson [Ref. 5] attempted to couple OPCODE2 (his version of ORCON1) with the optimizer CONMIN, but with little success. There were several reasons for this.

ORCON1 uses iterative techniques to solve for such quantities as condensate rate, steam mass flow rate and steam pressure through the sector. If unrealistic values are encountered, such as negative pressures or steam flow, or if the final steam exit-fraction exceeds a predetermined value, ORCON1 stops the analysis, returns to the beginning of the program and changes certain initial input parameters. The analysis begins again and the process is repeated until a satisfactory design is achieved. Thus ORCON1 has a limited capability to make design decisions to obtain a feasible design.

CONMIN, as do most optimizers, requires complete control in determining all iterative design variable values. As explained earlier, it uses perturbation techniques to calculate gradient information for each design variable and active design constraint, which it then uses to determine search directions. A perturbation of the design variable by CONMIN requires a complete, once-through analysis. If ORCON1 is coupled with CONMIN then any adjustment by ORCON1 will yield false gradient information to the optimizer and hinder, if not completely prevent, CONMIN from arriving at the optimum design. During program development it became apparent that the two programs were working independently against each other and that in its present state ORCON1 was incompatible with CONMIN.

In the formulation of CONDIP, it was necessary to locate and neutralize all the places where such design decisions are made. By removing the ability for CONDIP to make any design decisions it became totally passive and dependent on CONMIN for design variable changes.

However, once this was accomplished, another problem area was discovered. In the ORCON1 code and subsequently in CONDIP there are numerous thermal process and properties calculations that use logarithmic functions and other mathematical relationships which could produce singularities if the variables in the arguments approach zero or are negative. For example, saturation steam temperature is calculated from steam pressure using a logarithmic relationship. If, during a design analysis, saturation pressure approaches a negative value, this represents a clear violation of physical realities and of the limits of that property. Yet a computer cannot make that distinction so it tries to calculate the corresponding saturation temperature which, because of the logarithmic relationship, would be undefined. As just explained, ORCON1 with its built-in decision capability simply starts over when this situation is encountered. But CCNDIP, being completely dependent on CONMIN for design decisions, does not have that capability. Remembering that CONMIN requires a complete once-through analysis in order to collect enough information to make a design decision, it was necessary to somehow bypass such mathematical instabilities in order to keep the program operating. Yet the analysis still had to yield reasonable results from the given design in order to obtain meaningful gradients. This prompted the formulation of mathematical relationships to create "penalty" constraints which, if properly written, would indicate to the optimizer that a function or thermal property has violated its physical limits. However, not only would penalty constraints have to be defined, but a "fix up" or "correction" of the violated property or variable would be required in order to allow the analysis to continue. A good physical understanding of the inter-relationship between condenser parameters and the thermal processes resulting from the condenser design is

necessary so that the "fix up" of the violated property would still yield fairly accurate condenser information on which CONMIN could base its search for the optimized design.

For example, a condenser is usually designed around a given steam load. If the condenser has too many tubes, is too long, or coolant flow is too great, then the condenser will be overdesigned. There will be dry tubes within the condenser as all the steam is condensed before steam flow reaches the inner void. In CONDIP this means that zero or negative steam flow will be encountered in the analysis.

If, on the other hand, tube surface area is too small, coolant flow is inadequate or the condenser tube spacing is too tight and is choking the steam flow, then one of two things will happen. Either a quantity of uncondensed steam will make it completely through the condenser, or steam pressure loss in the condenser will cause steam pressure to drop below zero. In addition, there are two reasons why all the steam might not condense. It could be simply due to insufficient heat transfer surface area or it could be because the saturation temperature of the steam has dropped below the coolant inlet temperature. If the latter situation occurs, then there is no driving force for heat to transfer from the steam to the coolant. There is only one way that this situation can occur: if steam saturation pressure drops below some value indirectly determined by the coolant inlet temperature. In any event this condenser is certainly underdesigned and not capable of supporting the required steam load.

As stated earlier, the purpose of optimization is to obtain the best, feasible design. Thus, an understanding of the relationship between the physical characteristics of a given design and the subsequent thermal performances will certainly help in defining the appropriate penalty constraints and their corresponding limits. It will also

aid in the determination of appropriate "corrections" when those limits are violated. It is important to note that with the introduction of these penalty constraints, the definition of a feasible design is revised. A feasible design is now defined as one in which thermal properties and functions are not allowed to violate their physical limits, as well as other design constraints, anywhere in the condenser.

In CONDIP there are three basic thermal properties which could create the above mentioned problems if they fall below a certain value. They are steam saturation pressure, steam flow and steam temperature. Because of the direct relationship between steam saturation pressure and temperature, it was possible to deal with them simultaneously. The solutions that were developed to overcome the effects of these thermal violations determined the extent which CONDIP would optimize.

1. Steam Flow Effects

One source of mathematical instabilities within the program is if steam flow over the tubes falls to zero or below. It is intuitively obvious that steam flow can not physically fall below zero and that in order to keep the program running the steam flow rate must be kept greater than zero. However, correcting for this alone would certainly alter the results of that particular condenser design, perhaps even imply a feasible design.

To indicate to the optimizer an infeasible design was actually encountered - one in which steam flow had dropped below zero - the penalty function WTST was created. Since the condenser analysis is performed sector by sector and assuming there are J sectors in the condenser, then there had to be arrangements for J penalty constraints. This prompted the creation of the array, WTST(J), representing constraint penalty functions for each sector. The absolute

magnitude of these functions were directly related to both the severity of the steam flow violation and the number of dry tubes remaining in the sector. WTST(J) ranged in value from negative infinity to zero, where a value of zero represented a condenser design in which no flow violations occurred. Thus, WTST(J) were constrained functions whose lower limit was zero. For example, during an analysis, if steam flow was determined to fall below zero, then WTST for that sector would be given some negative value. In subsequent designs, as the number of dry tubes approached zero and better designs were obtained, then the magnitude of the penalty function approached zero indicating no constraint violation.

Since penalty constraints are entirely contrived relationships with no real physical basis, it is desirable to minimize their number to avoid the possibility of sending inaccurate signals to the optimizer.

To eliminate the need to use the WTST penalty functions as constraints, the values of WTST(J) were consolidated at the end of the sector analysis into the condenser steam exit-fraction constraint. Normally, steam exit-fraction ranges from zero to one and is simply equal to:

$$\frac{\text{steam leaving the condenser}}{\text{steam entering the condenser}}$$

By incorporating the WTST violations into steam exit-fraction, the exit-fraction was made a continuous function ranging in value from negative infinity to one. The negative steam exit-fraction represented a partially dry condenser and its magnitude was in direct proportion to the number of dry tubes. Thus instead of having to evaluate and calculate gradients for J number of WTST(J) constraints, the optimizer simply had to evaluate a previously defined and now expanded exit-fraction constraint.

It should be stressed that making steam exit-fraction continuous through zero was equally as important as eliminating the need for additional constraints. It can be reasonably assumed that for all practical condenser applications, exit steam fraction will always be limited to some positive number near zero. Here is where one applies the physical knowledge of the condenser and its relation to the thermal property of steam flow. As explained earlier, dry tubes represent an over-designed condenser. Thus the natural tendency is for the optimizer to alter those design variables so as to create a more compact condenser. As this occurs, steam exit-fraction will naturally increase. The upper active limit of that constraint will determine the optimum feasible design. While it is not necessary to have a lower limit for steam exit-fraction, it is very important for it to be a continuous smooth function especially in the region near the upper limit. It is therefore critical to properly define the penalty functions $WTST(J)$ in a way so as to provide a smooth transition from the negative, artificial values of negative steam exit-fraction to the real, positive values.

Since the steam flow penalty functions will not be used as constraints, the analytical results will provide gradient information to the optimizer. However, once steam flow has been determined to fall below zero, steam flow for that first dry row of tubes and all subsequent rows must be fixed up with dummy values to allow the program analysis to continue. How that "fix-up" is accomplished will ultimately determine the search direction for the optimizer.

Physically, once steam flow has gone to zero, there should be no further latent heat transfer, no subsequent condensate production, and further pressure losses should be only due to the flow of non-condensable gases. It is necessary to make the computer generated analysis reflect as

closely as possible these physical realities. Since the optimizer no longer has the penalty functions to use in calculating a search direction, other constraint values obtained from the analysis will dictate the next search direction. Gradients will also be calculated using these results and the determination of the next search direction will incorporate these gradients as well. In the case of negative steam flow, steam flow and condensate production over dry rows were given nominal values which were as small as the computer analysis would tolerate. These extremely small values closely approximate zero steam flow and generate results which resemble physical reality as closely as possible.

The following example is provided to better illustrate the logic used in CONDIP to handle negative steam flow. CONDIP determines flow rate through each row in each sector. During a sector analysis, CONDIP calculates the condensate generated at a given row and subtracts that value from the steam flow entering that row to calculate the steam leaving. The exiting steam flow rate is then checked to determine whether steam flow has gone to zero. If it has not, then the analysis continues. If it has, then the following two events occur.

The penalty function, $WTST(J)$, is calculated for that sector and dummy values are inserted for steam flow and condensate rate at the row where the violation occurred. For the remainder of the analysis condensate generation and subsequent steam flow calculations are bypassed and the remaining rows in the sector are fixed up with dummy values for steam flow and condensate. The analysis continues utilizing these dummy values in all appropriate heat transfer and pressure calculations. At the conclusion of the sector analysis, the values of the penalty functions, $WTST(J)$, of each sector are incorporated into the steam exit-fraction.

If the analysis revealed zero dry tubes then WTST(J) for all sectors would be zero and the steam exit-fraction would simply be calculated as:

$$\frac{\text{steam leaving the condenser}}{\text{steam entering the condenser}}$$

If, however, dry tubes were encountered in the condenser analysis then WTST(J) of some or all the sectors would be negative and dependent in magnitude on the number of dry tubes in each of the J sectors, as well as the severity of the steam flow violation. Steam flow leaving any sector which has gone dry would be zero and steam exit-fraction would be evaluated as:

$$\frac{\text{steam leaving any wet sectors}}{\text{steam entering the condenser}}$$

plus a weighted value of all the WTST(J) penalty function values. Using the relationships just described, it is apparent that steam exit-fraction: is negative if condenser tubes are dry; approaches zero as the design becomes feasible; and is greater than zero if there is steam leaving the condenser.

2. Steam Pressure and Temperature Effects

The other possible source of mathematical instability occurs when steam pressure falls below some preset limit. If pressure falls to zero, numerous mathematical singularities will be generated. Yet, before this situation can occur steam temperature will have already fallen below inlet coolant temperature causing singularities in the log mean temperature difference (LMTD) heat transfer calculation. Thus, the lower pressure limit which cannot be physically exceeded is not zero but the minimum saturation pressure established by the inlet coolant temperature. In CONDIP, this lower limit is given the variable name, PTLIM.

As steam flows through the condenser, pressure continually decreases due to friction losses and therefore, it is evaluated at each row in each sector. When the steam saturation pressure drops below PTLIM, indicating a physical violation of realistic limits, then the creation of a penalty function and a corresponding "fix up" of saturation pressure is required to allow the program to continue. The treatment of the problem was therefore analogous to the previous situation dealing with negative steam flow.

PTST(J) was the penalty function devised to indicate to the optimizer that the pressure limit, PTLIM, was violated in any of the J sectors. Values of these constraints ranged from negative infinity to zero, depending on the degree and location in the condenser of the violation. Since pressure is calculated on a row by row basis in each sector, the magnitudes of the pressure penalty functions were directly dependent not only on how much the calculated pressure dropped below PTLIM, but also on the number of rows remaining in the sector. Thus, as the condenser approached a feasible design the PTST(J) constraint values approached zero, indicating lessening violation of the minimum pressure.

As emphasized earlier, it is important to minimize the number of constraints, not only to avoid the possibility of sending confusing signals to the optimizer but also to reduce cost and improve program efficiency. This was accomplished here by inserting dummy values not only into the violated pressure variables but also associated thermal properties such as condensate generation, heat transfer coefficients and heat transfer rates for the row where the violation occurred and all subsequent rows in the sector. The dummy values were chosen such that realistic gradient information would be sent to the optimizer. The proper choice of "fix-up" values for these variables resulted in

the elimination of penalty functions as design constraints, and provided sufficient information to determine subsequent search directions.

It is necessary to understand the influence that steam pressure and temperature exert on the overall condenser analysis. With this knowledge it will be easier to predict the physical designs which could cause violations of the pressure limit. PTLIM is violated due to excessive steam flow pressure losses. As explained earlier, these large pressure losses would result from large steam velocities that are found in condensers which are too tightly designed. Thus, the particular condenser design is incapable of handling the required steam load, implying an infeasible design. Understanding this relationship will aid in choosing the appropriate "fix-up" values which will indicate to the optimizer that when the pressure limit is violated an infeasible condenser has been designed.

Physically, When steam temperature falls below coolant inlet temperature (PTLIM is violated) there is no heat transfer from the steam to the coolant and no additional steam is condensed. These physical realities must be reflected in the condenser analysis. Therefore, in subsequent rows, condensation and heat transfer rates were set equal to zero. Since there is no further condensation, the steam exit-fraction is equal to the steam flow at the point of violation divided by the total flow into the condenser. Thus PTLIM indirectly determines the exit steam fraction of the infeasible design. This relationship between exit-fraction and the PTLIM violation is what makes the penalty constraints obsolete. If PTLIM is violated early in the steam's passage through the condenser, steam exit-fractions will be large, violating its upper constraint limit and thus reflecting an underdesigned condenser. As the condenser design improves, then exit-fractions will decrease.

Physically, this can only be accomplished if the condenser design "opens up", reducing pressure losses in the condenser. Consequently, as condenser designs become larger, steam exit-fractions decrease and the condenser is driven towards a feasible design.

The following example illustrates the logic employed by CONDIP to handle steam pressure and temperature violations within the analysis. Condenser inlet flow is divided by the number of sectors in the condenser. Condenser inlet saturation pressure is determined by the steam inlet temperature. Entrance pressure losses are calculated and subtracted from the inlet pressure. The resulting pressure is checked against PTLIM and a violation at this point indicates a totally infeasible condenser in which no steam is condensed. Steam exit-fraction will thus be equal to one. If the saturation pressure is greater than PTLIM the analysis continues row by row through the sector. Pressure losses over each row are calculated and subtracted from the row inlet pressure to determine pressure into the next row of tubes. If this next-row steam pressure is determined to fall below PTLIM, then a thermal violation has occurred requiring "fix up". Subsequent rows are made to indicate zero condensate generation and zero heat transfer. Steam flow over the remaining rows is maintained at a constant value, which will subsequently be used to determine steam exit-fraction. Pressure variables over the remaining rows are given small positive values just large enough to allow the analysis to continue. Although all heat transfer and condensate calculations will be bypassed, the analysis must be allowed to continue so that pressure losses will continue to be calculated based on the steam flow at the point of violation. This is important since steam flow adjustments to the sectors are based on certain pressure comparisons between the sectors. The cumulative sum of all row pressure losses

in each of the sectors must be equal to within some tolerance. If they are not then steam flow into each of the sectors is altered so that the exit pressures from each sector converge to some common value. Thus, an accurate reflection of true pressure losses is important to this calculation.

The value of the steam exit-fraction is again determined to be the single constraint necessary to drive subsequent condenser designs to a feasible optimum configuration. The pressure penalty constraints proved to be superfluous information, but the corresponding variable "fix-up" was critical in the determination of search direction.

C. LIMITATIONS

During the development of CONDIP, it became apparent that steam exit-fraction would become the key constraint during optimization of any objective function. A feasible design implies that steam exit-fraction is a small positive number perhaps somewhere between zero and 0.1 percent. As explained earlier, violations of either steam flow or pressure physical limits resulted in penalty functions and variable "fix-up" which were later directly or indirectly incorporated in the calculation of steam exit fraction. Thus any feasible design, let alone the optimum one, centers on the limits placed upon this design constraint. Any number of design variable combinations will yield a feasible design, and each design variable affects steam exit-fraction differently. The intertwined, complex calculations used to ultimately determining exit-fraction are done by sector and row with each design variable repeatedly playing a factor. For example, the profound effect of both vapor shear and condensate inundation on the shell-side heat transfer coefficient and consequently steam exit-fraction, is

indirectly determined by numerous design variables. However, their effects are impossible to predict. The cascading effect of the thousands of calculations performed during the course of a design analysis is to ultimately create a single, highly non-linear variable in the form of the steam exit-fraction, upon which design decisions will be made.

As more design variables were involved in the analysis, the optimizer had difficulty determining their often conflicting effects on both the objective function and the steam exit-fraction. A small perturbation of each of the design variables independently would yield gradients indicating design improvement. But when these gradients were evaluated simultaneously to actually determine the direction of the subsequent design, their combined effect would actually indicate either no improvement of the objective function or a violation of the steam exit fraction design constraint. The end result would be that the optimization process would stall as no feasible search direction could be obtained. Larger perturbations to the design variables were required to properly evaluate their relative effects on the objective function and any active or violated constraints. This would enable the optimizer to overcome either small inconsistencies or discontinuities in the objective function and the constraint functions which would otherwise prevent the optimizer from reaching the optimum design. This was accomplished during data input by changing the normalized finite difference step from 0.01 to about 0.1. Increasing the finite difference is not without its drawbacks. As the optimum objective value is approached, the optimizer overlooks the subtle effects of small changes in the design variables because of the relatively large perturbations. Thus, depending on the initial design variables, the optimizer will improve the design to some point near, but necessarily not at, the optimum.

When a design becomes feasible, steam exit-fraction will always become an active constraint. But the stated goal is not in achieving a feasible design but in driving the design to a feasible optimum. However, this iterative process can not be accomplished at the expense of violating a constraint and it was here that further complication was introduced. The initial impetus in any optimization process is to first obtain a feasible design. However, once the very small steam exit-fractions are obtained that are necessary for a feasible design, the exit-fraction becomes extremely sensitive to any further design variable changes. Thus any effort to further improve the current design could easily cause exit fraction to increase. Even slight increases would be perceived as violations of the constraint limit and thus prevent further optimization from the first feasible design. There are two possible solutions to this problem. Either increase the upper limit on the exit-fraction constraint or redefine the constraint. COPEs formulates the general constraint function in such a way as to allow the user to increase the active region about the constraint limit. This is accomplished here by increasing the normalization factor in the following expression for the exit-fraction constraint function:

$$\frac{BU - EXITFR}{SCAL1} \leq 0.$$

where BU is the upper constraint bound, EXITFR is the exit-fraction constraint value and SCAL1 is the normalization factor for this constraint. Increasing the normalization factor reduces the optimizer's sensitivity to constraint violations by enlarging the range of constraint values in which the constraint is active. This enlarges the region of feasibility and allows the optimizer more flexibility in altering design variables by reducing the risk of violating

the constraint. The overall effect is that the design optimization can continue but at the expense of accurate constraint limits. The normalization factor used effectively for the exit-fraction in CONDIP analysis was approximately 0.1.

One of the stated objectives was to create a robust program which would consistently yield the single best optimum design independent of the initial design and not get hung up on a relative optimum. As it was explained earlier, although relative optimums represented design improvement, they also indicated the inability of the optimizer to locate the single best or true optimum design. However, the objective was achieved for only three design variables. When more than three design variables were used, the optimum designs became loosely dependent on the initial input, although not in any predictable way. This is not to say that the condenser design did not optimize. By incorporating the finite difference and scaling normalization on exit-fraction as described above, final designs did yield objective function values which were continually within about ten percent of the true optimum regardless of the initial design. However, there was just no guarantee that the single, best optimum design could be consistently obtained. In summary, the reasons why CONDIP did not consistently optimize to the single, true optimum were: the extreme non-linearity of the steam exit-fraction, the need for a large finite difference gradient, and the need for a normalization factor for the exit-fraction upper limit constraint.

While the optimum design solutions obtained from CONDIP may be sufficient, there are several ways to improve the results and increase the chances of obtaining the best possible design. The easiest way is to try several initial input values until the user is satisfied that the best solution has been obtained. The problem with this approach is

that it is both costly and time consuming. A second recommendation is to couple an extremely simplified version of the condenser analysis with the optimizer to obtain a educated guess as to what the optimum design should be. The results of this analysis could then be used as input for CONDIP. OPCODE1, which utilizes the HEI methods in its analysis, is a likely candidate. The advantage of this approach is that a quicker, cheaper analysis can be used to obtain a rough idea of the anticipated optimum design. CONDIP can then use these design results to obtain even better and more accurate solutions, faster. There is still no guarantee, however, that the true optimum will be solved. Perhaps with the development of more robust and versatile optimizers, ones which uses numerical techniques and methods that are better suited to this type of problem than CONMIN, more precise solutions can be obtained. However, there is little more that can be done to simplify the analysis of the steam exit-fraction and subsequently linearize the problem.

IV. DESCRIPTION OF THE MAIN AND SUPPORTING SUBROUTINES

A. MAJOR SUBROUTINES

The following section contains a brief description of the major subroutines in CONDIP. The appropriate flow diagrams are also provided to better illustrate and complement the explanations. For further information concerning the various subroutines and functions see the CONDIP listing in Appendix C and ORCON1 [Ref. 2].

1. ANALIZ

This subroutine basically arranges CONDIP in a standardized form which is compatible with COPES/CONMIN. COPES uses a variable flag, ICALC, to coordinate the optimization process with ANALIZ. Utilizing this flag ANALIZ then calls the input, analysis and output portions of CONDIP as required. When COPES sets ICALC equal to one ANALIZ reads in all initial input. This is the only time any input can be entered. When ICALC equals two, COPES works with CONMIN to optimize the design. ANALIZ makes available the analysis portion of the program to be used repeatedly by CONMIN. When COPES sets ICALC equal to three, the optimization is complete and ANALIZ calls all applicable output subroutines. Figure 4.1 illustrates the flow process for ANALIZ.

2. INPUT

This subroutine enters all initial input of data by which the initial design is determined. The resultant design may be either feasible or infeasible, subject to the limitations previously discussed, so it is not critical what values are initially assigned to the design variables.

However, the initial input is screened to prevent the introduction of totally unrealistic values of variables into the program. For example, initial tube thickness, tube inner diameter, tube number and tube length are all checked to ensure that their values are greater than zero. If any of the screened initial inputs do not satisfy the minimum requirements, then the program exits prior to entry into the optimizer. The limits of the design variables and constraints will prevent similar situations from occurring during the analysis. Figure 4.2 presents the flow diagram for the INPUT subroutine.

3. OUT3

This subroutine simply prints all the initial values entered in the INPUT subroutine.

4. ORCON

This subroutine calculates the bundle geometry, flooding factors and such coolant flow parameters as pressure loss, flow rate, and pumping power. There are two options available to determine bundle geometry, each with certain advantages and disadvantages.

Option1: The number of rows is entered as a constant and the tube number is determined based on pitch, tube outer diameter, and row spacing. The advantage of this method to determine bundle geometry is that it allows the user to linearly vary pitch and/or tube inner diameter by row. The disadvantage is that tube number is a dependent variable. The optimizer is therefore limited in determining the optimum design by the specified number of rows.

Option2: The number of tubes can be used as a design variable while the number of rows is determined by tube number, row spacing, pitch and tube outer diameter. There

is more flexibility in this method of condenser design but it is not possible to linearly vary tube pitch and inner diameter. The condenser bundle is generated from a specified inner void out, and all the appropriate condenser geometry is determined for one of the identical 30 degree sectors. Overall bundle volume is then calculated as is the ratio of tube hole area to tube sheet area.

Once the basic condenser geometry has been determined, the code then proceeds through an algorithm to calculate baffle location based on an input value specifying the number of baffles desired in the condenser bundle. After this has been completed, flooding factors are determined. That is, the number of tubes in a vertical row above the central tube in each row is calculated. This is done for each of the six sectors on one side of a circular bundle. Symmetry is assumed for the other side. These flooding factors are later used in calculations to determine the effect of condensate inundation on shell-side heat transfer coefficients.

Finally, ORCON calculates coolant mass flow, coolant velocity, header pressure difference and pumping power based on the type of coolant flow input received. The flow chart in Figure 4.3 is a simple illustration of the logic used in ORCON. From ORCON, the subroutine SECALC is called.

5. SECALC

This subroutine determines all the parameters of each of the sectors in the condenser by row. The first calculation made in SECALC is the determination of the cooler geometry, if there is one. Entrance pressure losses into the condenser bundle are calculated for each sector and saturation pressure is checked to ensure that it is greater than PTLIM. From this point, much of the remaining

subroutine is comprised of two do-loops with one nested inside the other. The outer loop cycles through however many sectors are in the condenser model. The inner loop cycles through the rows in the sectors. Pressure, temperature, mixture velocity, steam flow and condensate flow are calculated at each sector row. The subroutine HETTRN is called repeatedly to provide the necessary heat transfer information. Pressure and steam flow is checked continually at each row to ensure that neither falls below its predetermined lower limits. In the event that either situation occurs, the appropriate penalty function and fix-up procedure is implemented to enable the analysis to continue. As previously discussed, these values are chosen to reflect as accurately as possible real conditions which would occur when steam flow or pressure violate their physical limits.

Once all the sectors have been analyzed the cumulative steam-side pressure losses from each sector are compared. Steam pressure at the inner void must be uniform, therefore the sector pressure losses are required to be equal within some allowable tolerance. If they are not, then the distribution of inlet steam flow into each sector is altered to force the pressure losses to converge to a single value. Once steam flow to the sectors has been adjusted, the sector and row analysis in SECALC is repeated until the pressure losses within each sector approach a common value. After the pressure comparison has been satisfied, certain overall condenser parameters are calculated such as steam exit-fraction, bundle heat load, and steam-side pressure drops.

Finally, if a cooler is required, the subroutine COOLEX is called. Otherwise the condenser analysis is complete. The flow diagram for SECALC is presented in Figure 4.4.

6. HETTRN

This subroutine is called repeatedly in SECALC to solve for all shell and tube-side heat transfer properties for each row in each sector of the condenser. In particular, values for the overall heat transfer coefficient and log mean temperature difference are utilized by SECALC in computing condensate production and heat transfer rate at each row of tubes.

On entering this subroutine, a series of estimates for certain row variables are calculated. Based on an assumed initial value for the overall heat transfer coefficient, the exit coolant temperature and corresponding film temperature are calculated. Utilizing these temperatures, the LMTD, thermal resistances, individual heat transfer coefficients and numerous other heat transfer parameters are then calculated. Finally, another value for the overall heat transfer coefficient is determined based on the above-mentioned analysis, and this final value is subsequently compared to the initial value. If they are not in agreement, within a specified degree of tolerance, then the initial value for the overall heat transfer coefficient is updated and the entire process is repeated until the initial and final values converge. This iterative process is necessary as temperature dependent heat transfer coefficients, film temperature drops, and exit coolant temperatures are all being calculated simultaneously.

Note that it is in HETTRN that the concepts of vapor shear and condensate inundation are incorporated. Heat transfer coefficients are corrected for both effects based on the calculations presented earlier. Also note that since steam temperature is never allowed to drop below inlet coolant temperature in the calling subroutine SECALC, resultant LMTD calculations in HETTRN will not yield singularities.

Once all the heat transfer variables have been determined, control is returned to SECALC where the appropriate results are utilized and stored. The appropriate flow diagram for HETTRN presented in Figure 4.5.

7. COOLEX

This subroutine solves for all the necessary parameters required in the cooler analysis. The cooler is assumed to be of rectangular cross-section with the height of the cooler not to exceed the difference between the condenser inner and outer radii. The values used for tube pitch and tube diameters in the cooler are the same as the innermost row of the condenser bundle.

Steam exits the condenser bundle, collects in the inner void and enters the bottom row of the cooler. The steam then proceeds vertically up through the cooler. The physical location of the cooler is not a prerequisite to the subsequent design, although it is expected that the cooler will be placed within the condenser bundle, thus the limit on cooler height.

The first calculation in COOLEX determines the steam velocity at minimum cross-section in the first row of tubes, VLCMAX. VLCMAX is directly proportional to the amount of steam and non-condensable gas entering the cooler as well as the cooler geometry. Therefore the constrained limits for VLCMAX will play a major factor in the overall condenser design.

Subsequent row analysis is treated identically as in SECALC. However, all pertinent heat transfer data are calculated directly within COOLEX, making it independent of HETTRN. Steam pressure and steam flow are checked at each row to ensure that the appropriate limits are not violated and all thermodynamic parameters are calculated. At the conclusion of COOLEX, cooler performance variables such as

heat load, exit-fraction steam, steam pressure losses and overall heat transfer coefficients are calculated and control is returned to SECALC. The flow diagram for the COOLEX subroutine is illustrated in Figure 4.6.

8. OUT2

This subroutine prints the overall condenser bundle results including heat load, steam exit-fraction, overall heat transfer coefficient, overall condenser LMTD and bundle volume. Normally, OUT2 is called once after the initial design is analyzed and again after the optimum design has been determined. Final design variable values such as tube number, coolant flow, tube pitch, tube wall thickness and tube inner diameter are also printed.

9. OUT2C

This subroutine prints the cooler results as well as the combined cooler/condenser results. This subroutine is called from the subroutine OUT2 and is called only if a cooler is required and subsequently designed. Therefore, these results will always be printed in conjunction with OUT2 output.

10. OUT3

This subroutine prints a very detailed output of the condenser and cooler results by row and sector. Nearly all the thermodynamic and heat transfer properties are presented, thus providing a rather complete picture of conditions everywhere in the condenser. This is extremely helpful in determining, for example, where additional heat transfer enhancement would be most beneficial, or where baffles should be best located to reduce the effects of condensate inundation.

B. SUPPORTING SUBROUTINES

The following is a brief description of supporting functions and subroutines called frequently by the main subroutines.

1. DFSVTY: This subroutine returns the value of the mutual diffusivity of the steam and non-condensable gas present.
2. XTR: This function subroutine transforms the calculated data, received in the argument list, to the log values and performs a linear regression on two or more points using the model.
3. AMUFN: This function subroutine calculates the viscosity of the non-condensable gas in $\text{lbm}/(\text{ft}\cdot\text{sec})$.
4. BMUFN: This function subroutine calculates the viscosity of a saline solution in the range of 0-24 percent concentration and temperatures of 40-210 °F in $\text{lbm}/(\text{hr}\cdot\text{ft})$.
5. CPAFN: This function subroutine returns a value for the heat capacity of the inert, non-condensable gas mixed in with the steam in units of $\text{Btu}/(\text{lbm}\cdot\text{mol}\cdot^\circ\text{F})$.
6. CPFN: This function subroutine calculates the specific heat of a saline solution units of $\text{Btu}/(\text{lbm}\cdot^\circ\text{F})$.
7. CPSEFN: This function subroutine calculates the heat capacity of steam in $\text{Btu}/(\text{lbm}\cdot\text{mol}\cdot^\circ\text{F})$.
8. HFGFN: This function subroutine returns a value for the latent heat of vaporization of water in Btu/lbm .
9. PRSDRP: This subroutine returns the shell-side pressure drop across a row of tubes in psia .
10. PSATFN: This function subroutine calculates saturation

pressure of steam as function of temperature. Pressure is returned in units of psia.

11. ROEPN: This function subroutine calculates the density of a saline solution of concentration range 0-24 percent and temperature range of 40-300 °F. Density is returned in units of lbm/(cu.ft.).

12. SKBFN: This function subroutine calculates the thermal conductivity of a saline solution of concentration range 0-24 percent and a temperature range of 40-300 °F. Thermal conductivity is in (Btu)/(hr-f-°F).

13. TSATFN: This function subroutine returns the value for steam temperature in °R given a pressure in psia.

14. VGFN: This function subroutine calculates the specific volume of steam as a function of temperature and pressure. It has units of (cu.ft.)/lbm.

15. SWITCH: This function subroutine reverses the order of a stored array.

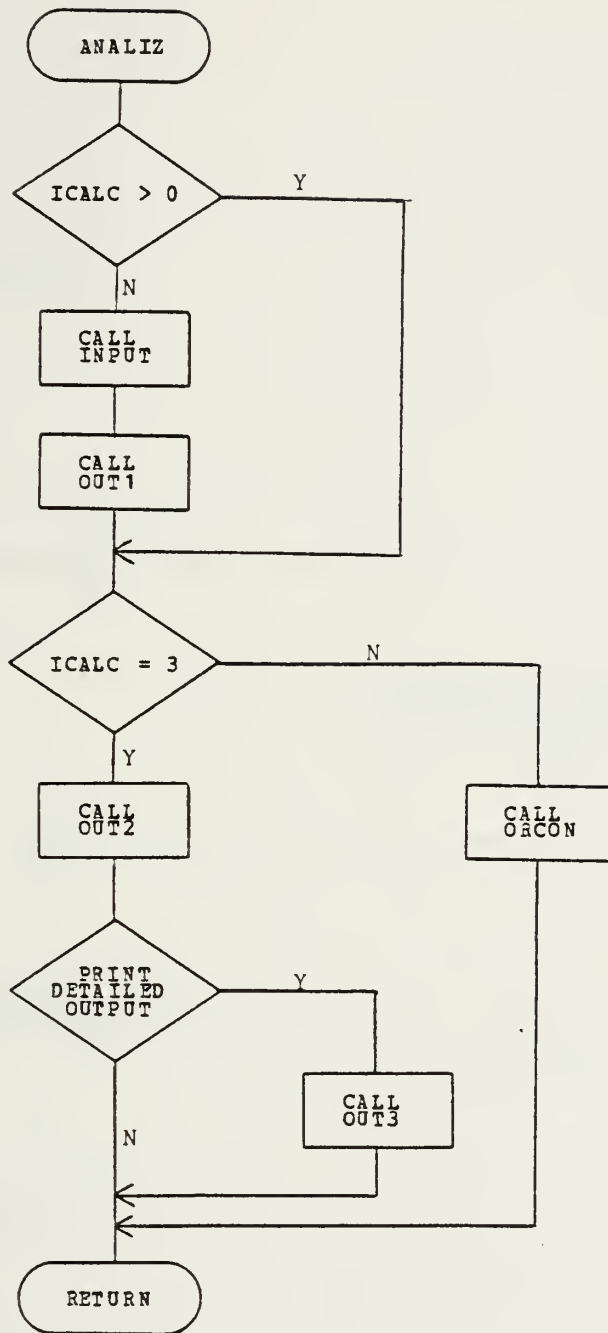


Figure 4.1 Flow Diagram for the ANALIZ Subroutine.

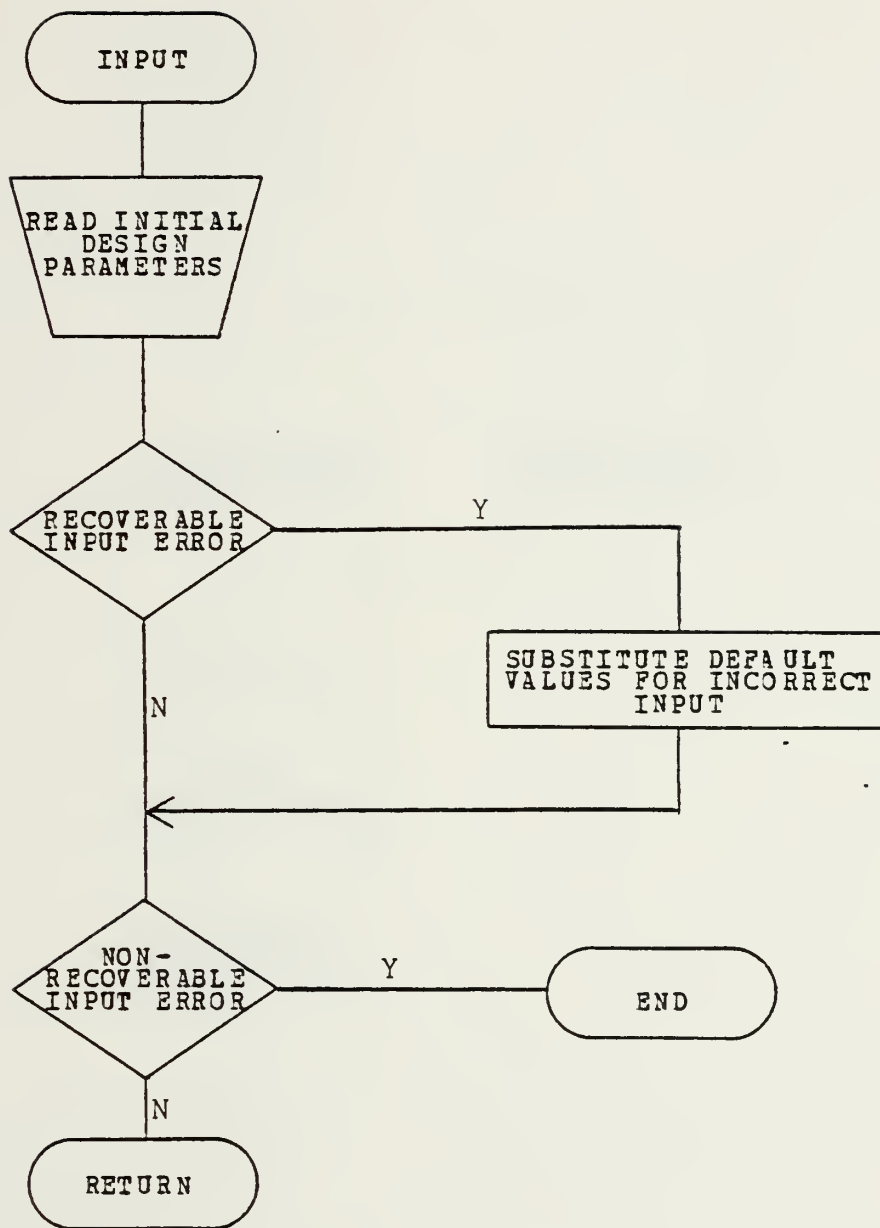


Figure 4.2 Flow Diagram for the INPUT Subroutine.

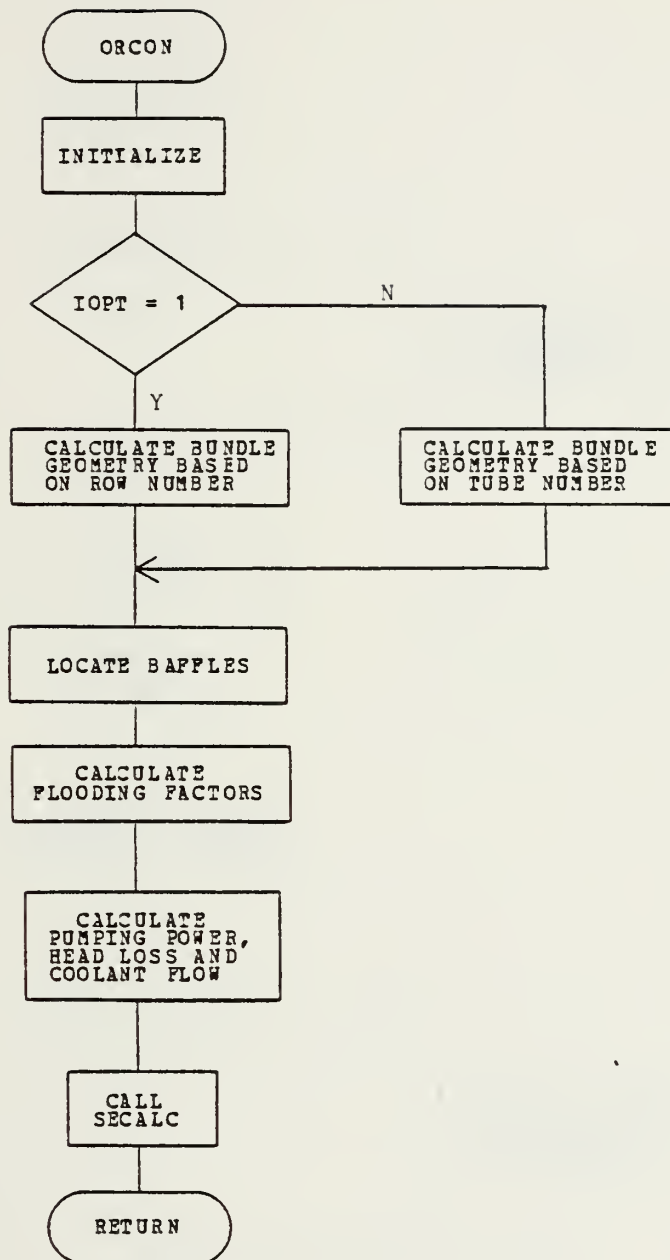


Figure 4.3 Flow Diagram for the ORCON Subroutine.

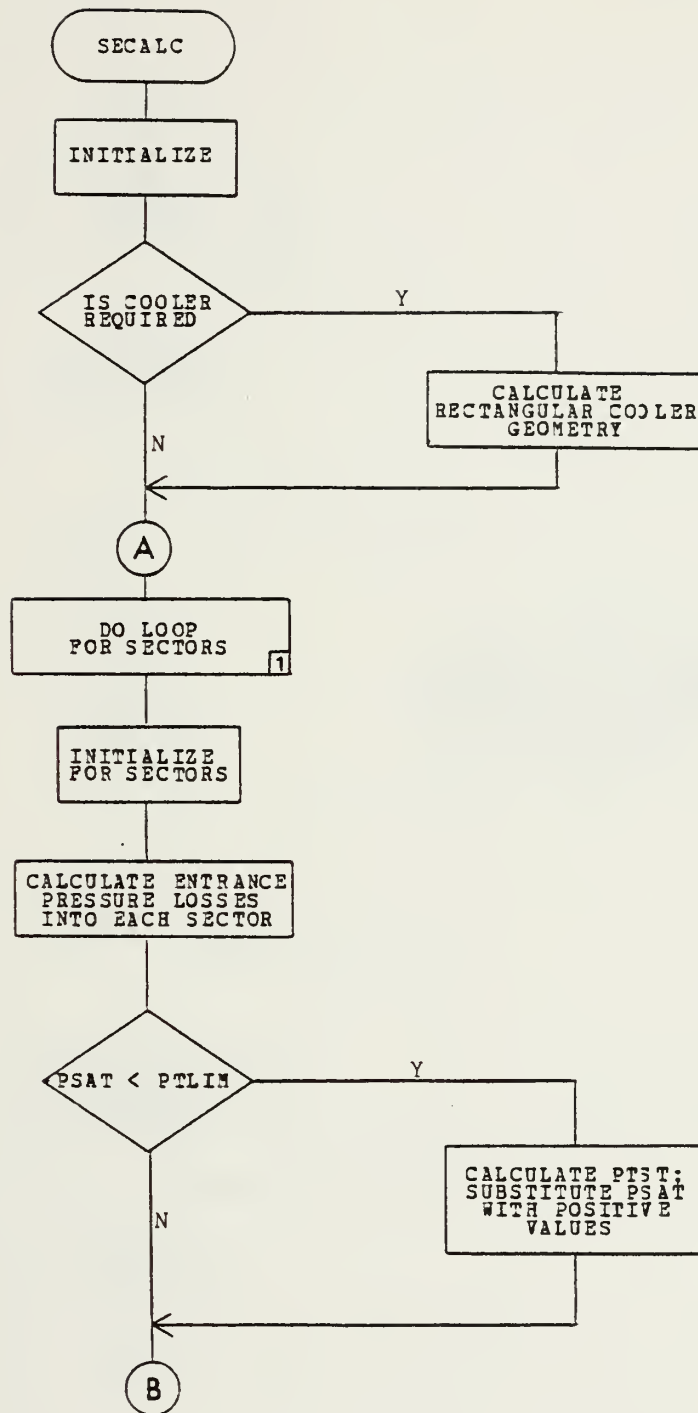
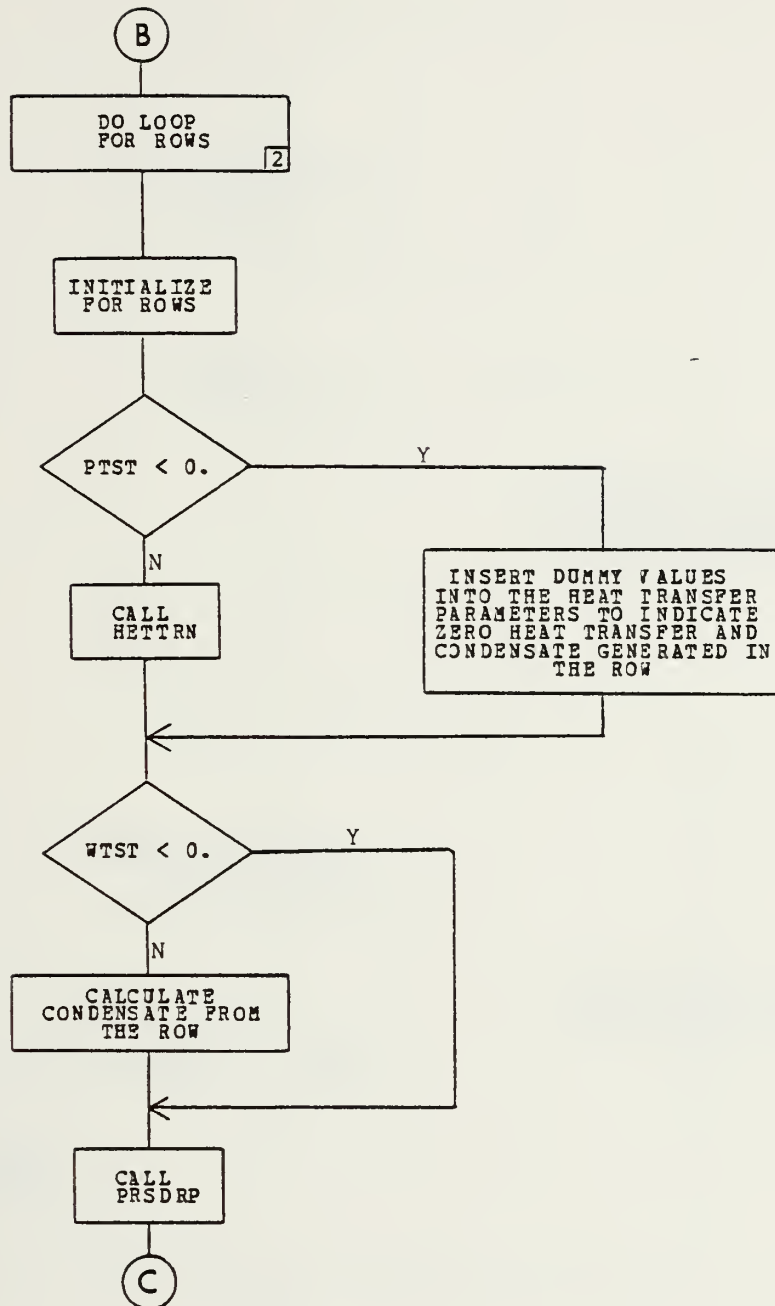
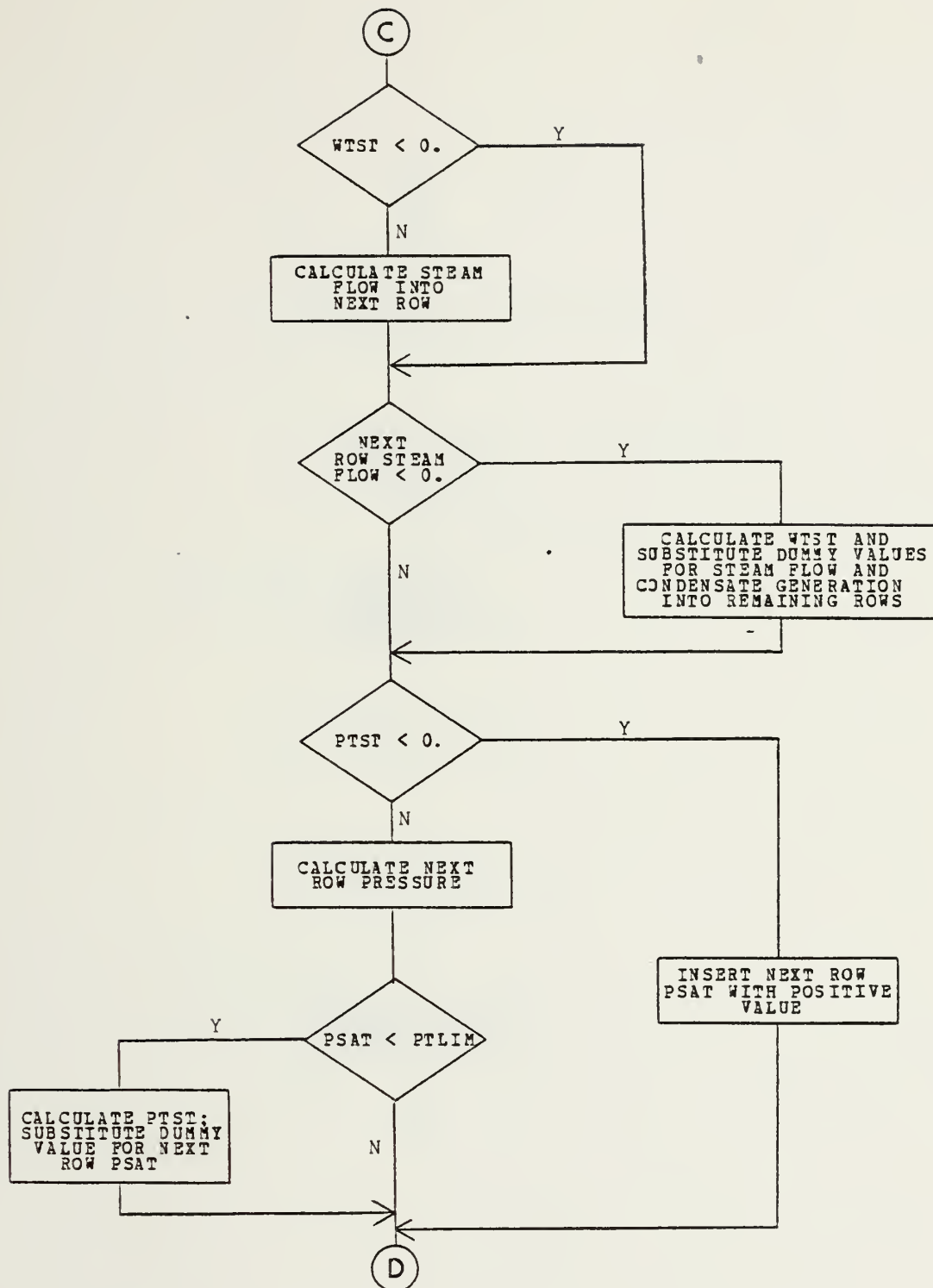


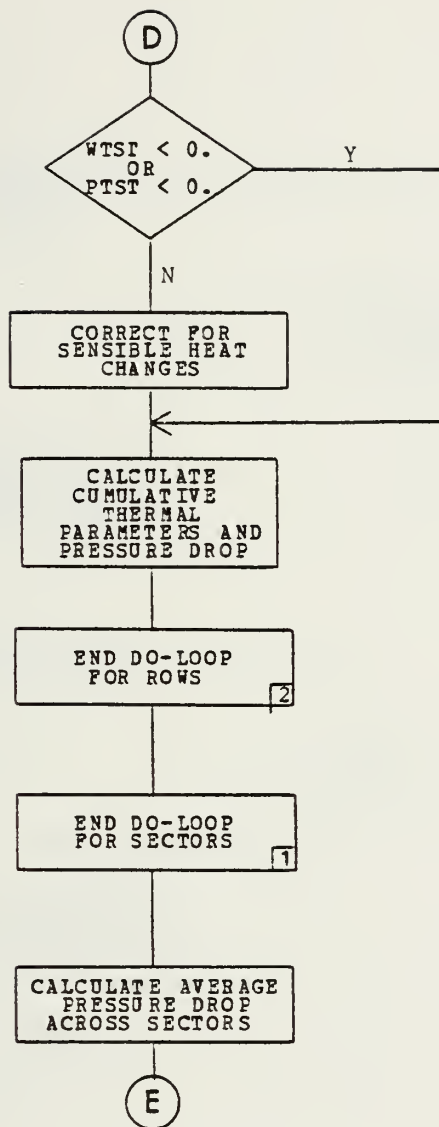
Figure 4.4 Flow Diagram for the SECALC Subroutine.



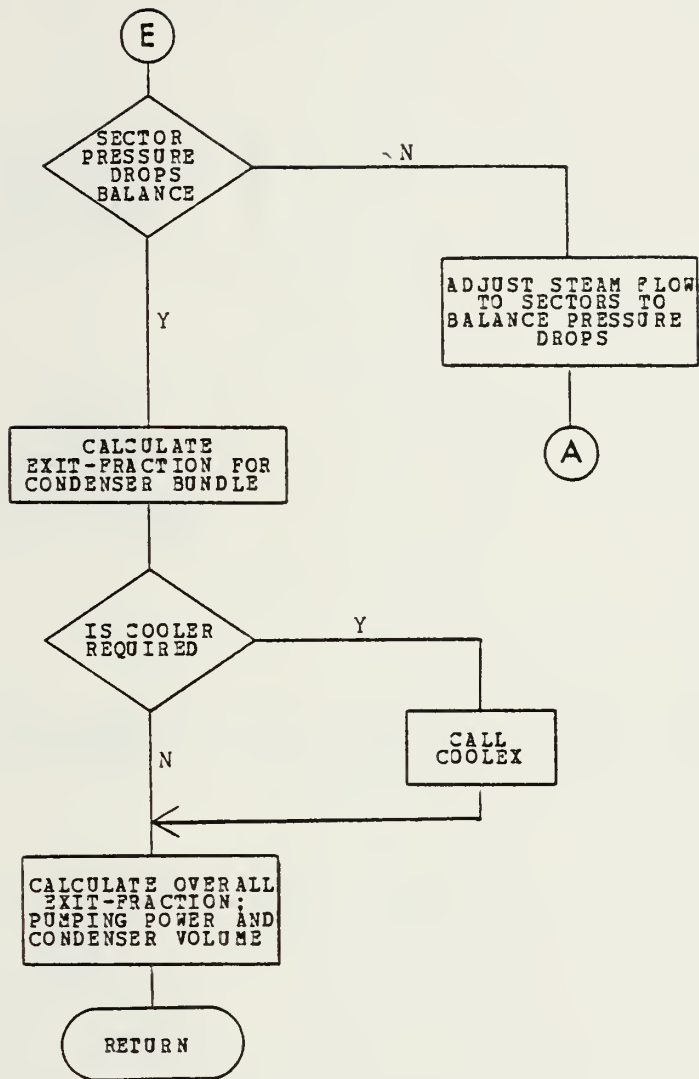
SECALC Flow Diagram (continued)



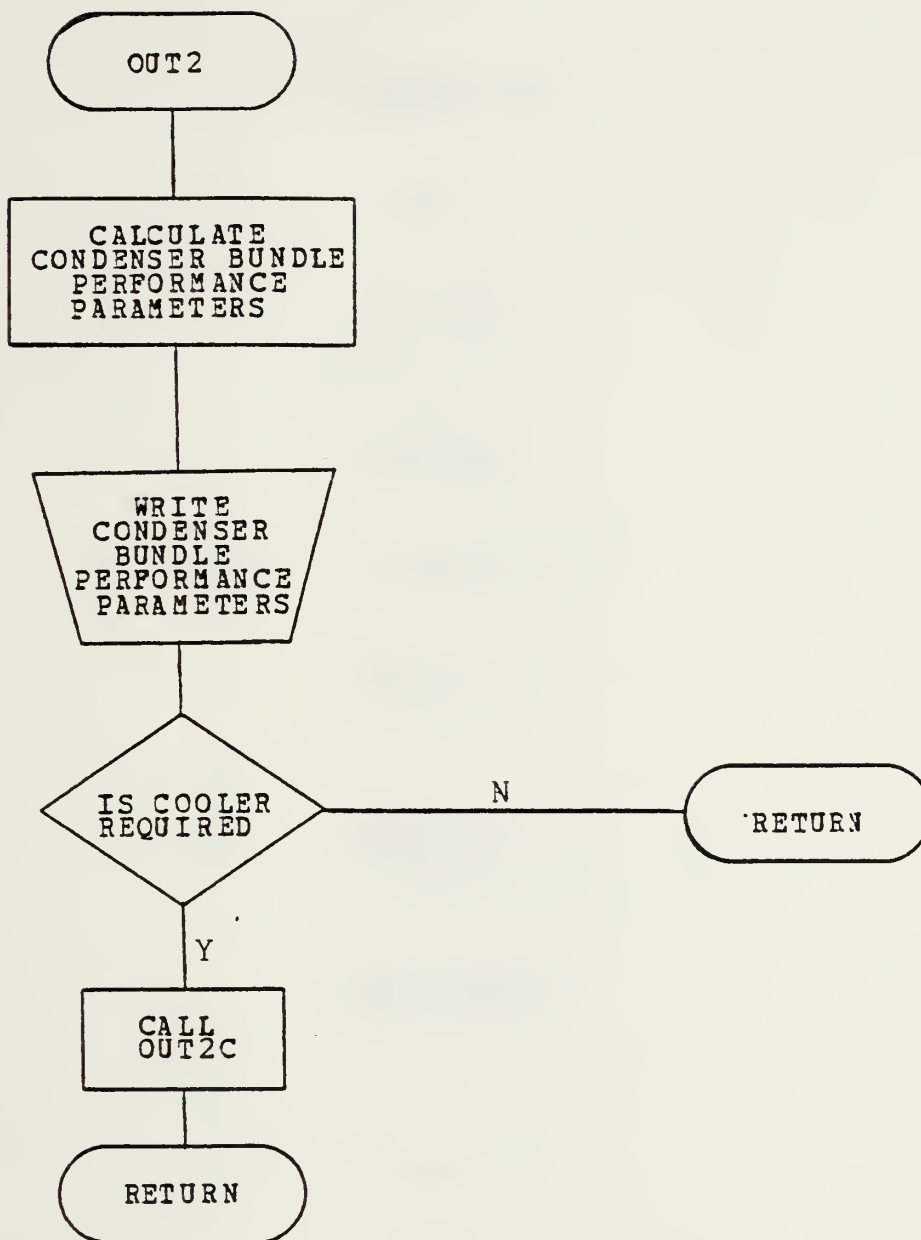
SECALC Flow Diagram (continued)



SECALC Flow Diagram (continued)



SECALC Flow Diagram (continued)



Flow Diagram for the OUT2 Subroutine

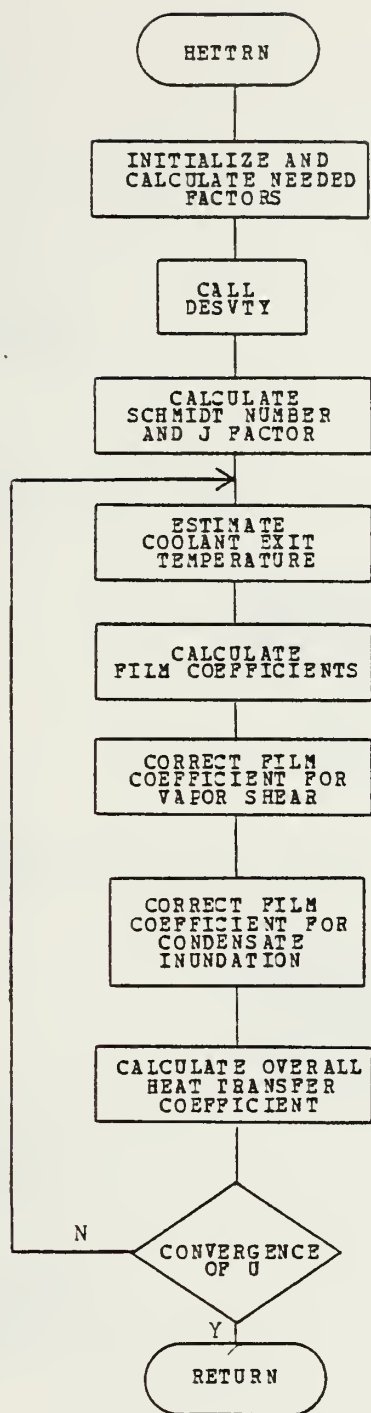


Figure 4.5 Flow Diagram for the HETTRN Subroutine.

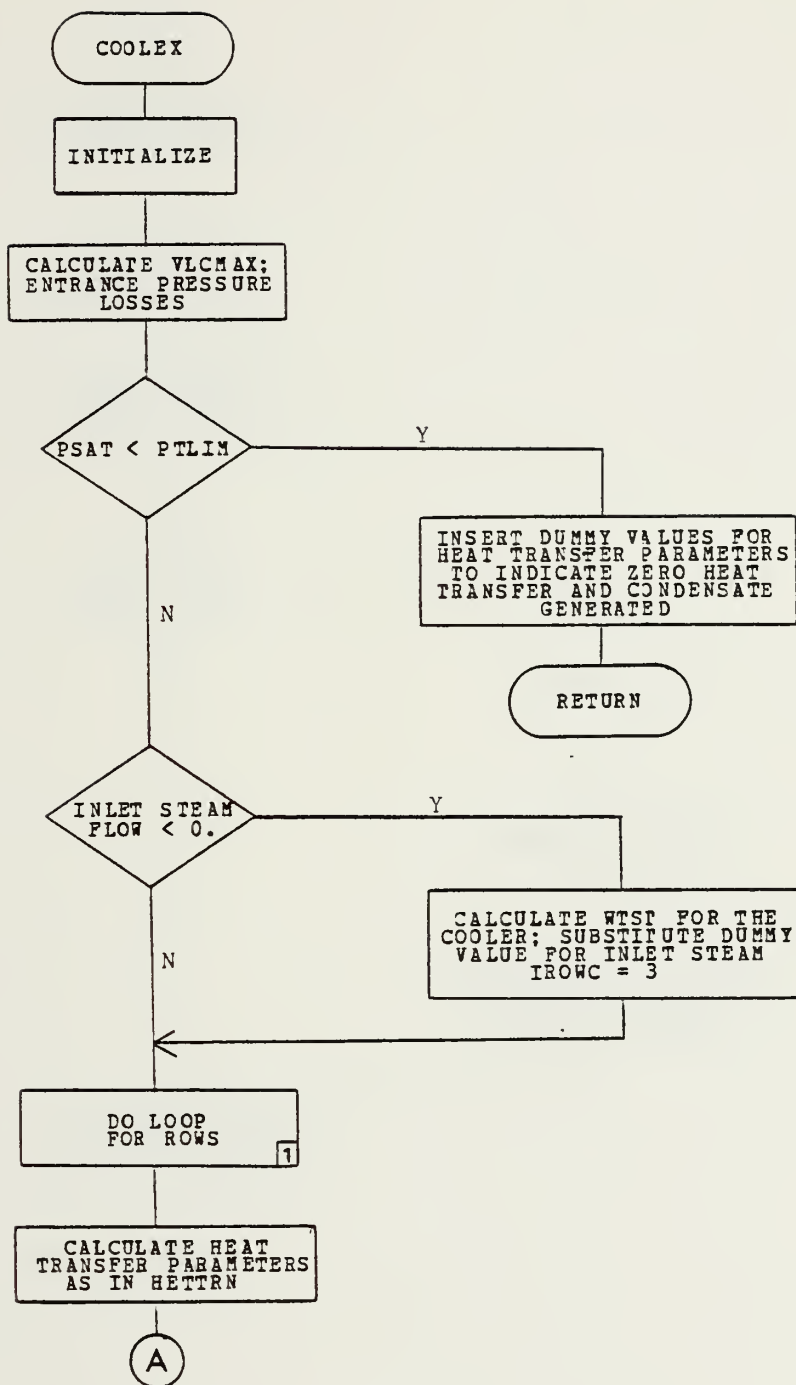
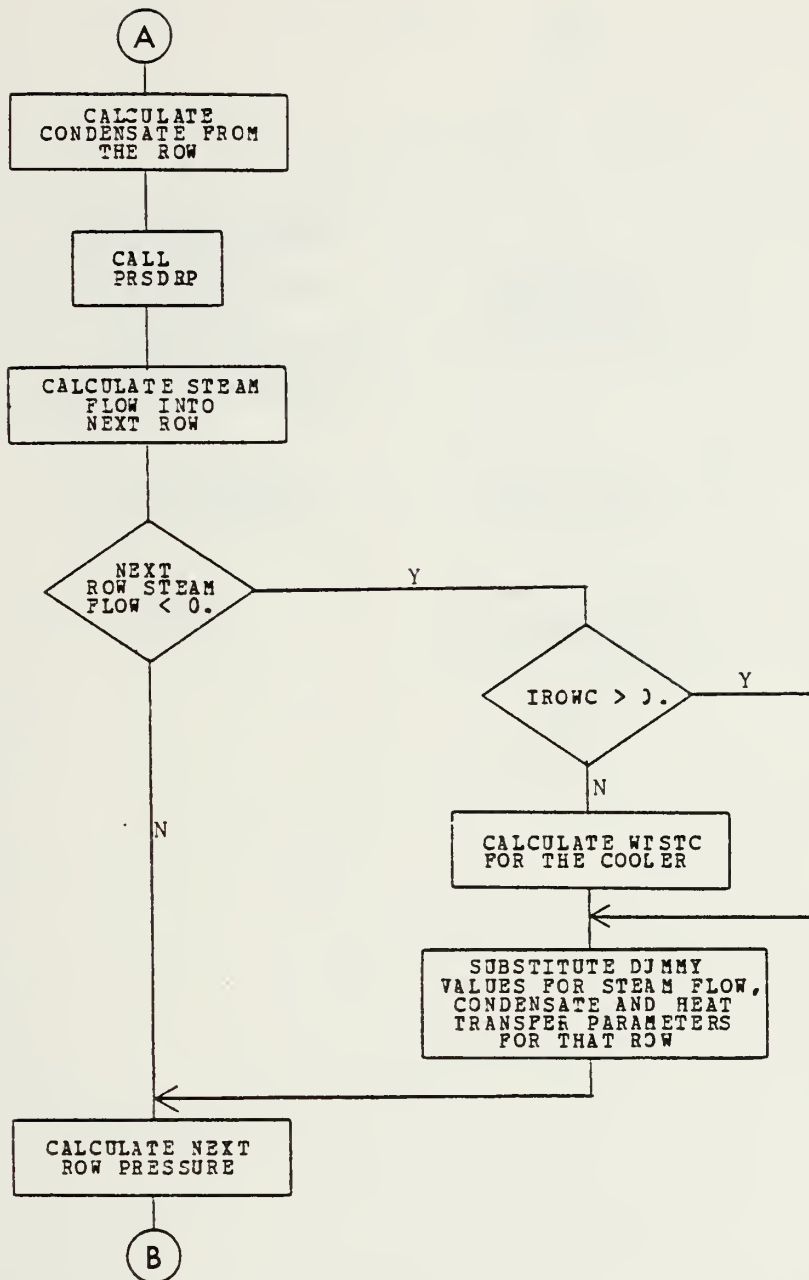
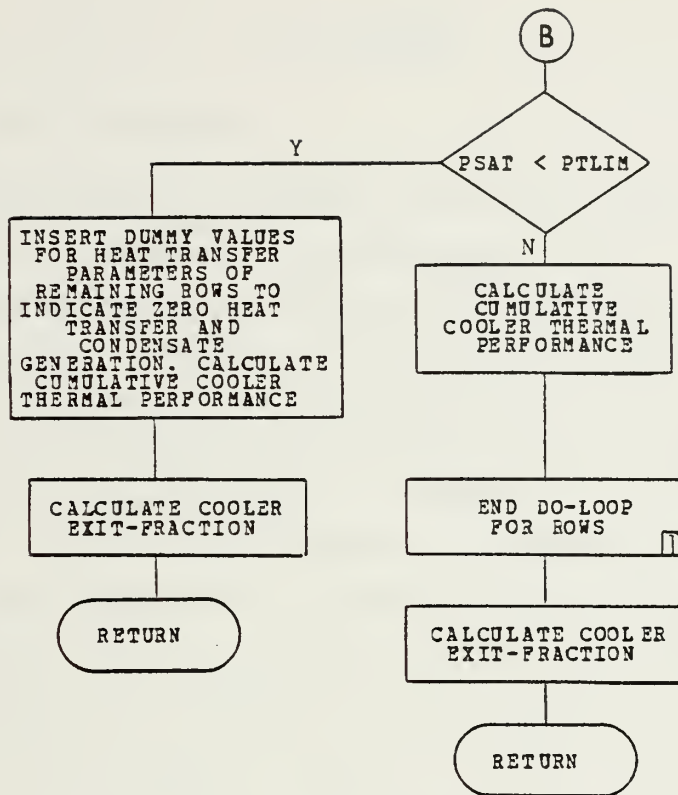


Figure 4.6 Flow Diagram for the COOLEX Subroutine.



COOLEX Flow Diagram (continued)



COOLEX Flow Diagram (continued)

V. RESULTS

A. CONDIP VERIFICATION

It was desirable to verify the single pass capability of CONDIP (i.e. without optimization) as a predictor of condenser performance by comparison with actual experimental data. However, complete and accurate data on condenser design and corresponding performance is not always readily available. Lynch [Ref. 18] encountered this same problem in attempting to verify ORCON1. However, he did manage to locate some actual experimental condenser data, obtained during a test conducted to determine the general performance of the DDG-37 class propulsion machinery [Ref. 16]. The test took place at the Naval Boiler and Turbine Laboratory and was conducted primarily to determine the performance of the turbine and reduction gears. Some limited condenser data was taken as a by-product. The various measurements were obtained as described below:

1. Steam flow measurements were made by weighing the condensate.

2. Cooling water inlet and outlet temperatures were measured by thermometers installed in the inlet and discharge lines.

3. The heat load is calculated based on total steam flow into the condenser multiplied by the difference between inlet steam and condensate enthalpies.

4. Circulating water flow was determined from a heat balance around the condenser. The total heat load was divided by the circulating water heat capacity and temperature rise to obtain flow rate.

5. Condenser inlet pressure was determined by pressure instruments located above the condenser inlet flanges.

6. Non-condensable gas flow was measured by a flowrator.

7. Pressure at the air ejector was measured directly. This pressure and the inlet pressure determined the pressure drop across the condenser.

It should be pointed out that this data were not recorded with the care that normally accompanies scientific data collection. Neither the instruments nor the techniques employed were particularly accurate. The possibility that this observed data are in error casts a cloud over the credibility of the corresponding condenser performance, which was calculated based on those values. However, for a lack of better alternatives, this data and the resulting condenser analysis will be used to determine the reliability of CONDIP.

The DDG-37 condenser geometric design variables obtained from the technical manual [Ref. 15] and input parameters corresponding to a full speed run are presented in Table I. An attempt was made to repeat the design using CONDIP. The results of CONDIP's proposed design as well as the experimental performance are presented for comparison in Table II. Percentages were calculated to quantify the differences between the actual and hypothetical performance.

Before elaborating on the results of this verification, some notable differences between the two designs must be clarified. First, a specific fouling factor was not determined at the time of the experiment and was therefore not provided. A somewhat realistic cleanliness factor of 87.5 percent (fouling factor of .0002) was utilized.

Second, in the DDG-37 condenser the rectangular cooler appears to be inserted directly into the condenser bundle.

However, in order to accomodate the cooler, the bundle must expand or distort. In addition, a void of some dimensions must be provided-for in the center of the bundle to collect any uncondensed steam and non-condensable gases. Diagrams in [Ref. 15] indicate that the DDG-37 condenser is indeed nearly elliptical in shape with bundle axes of 5.67 and 7.17 feet. Although it is apparent that a void does exist, exact dimensions can not be readily determined from the available information.

CONDIP approximates this design by creating separately a circular condenser bundle and a rectangular cooler, the height of which cannot exceed the difference between the outer and inner bundle radii. A circular void of pre-determined size is provided-for when determining the condenser bundle geometry. Subsequent volume calculations are performed on the condenser bundle and cooler separately and the overall condenser volume is computed as simply their sum.

Although CONDIP does not exactly duplicate the geometric configuration of the DDG-37 condenser, it was possible to manipulate certain initial design variables in order to cause CONDIP to develop an approximately equivalent configuration. These variables were chosen because, for small changes in their values, there is a rather significant change in the bundle geometry with relatively small effects on the overall condenser performance. Since there are no specific dimensions provided for the inner void in the DDG-37 technical manual, it was picked to be one of the design variables to be adjusted. Row spacing was also adjusted because it satisfied the conditions described above. Through trial and error a combination of row spacing and inner void radius were determined, from which CONDIP yielded a geometric design similar to the DDG-37 condenser and that satisfied condenser requirements specified in

[Ref. 17]. In this particular case, the void diameter and row spacing were determined to be 1.1 feet and 1.35 inches respectively. This arrangement enabled the condenser model to closely approximate the tube sheet area ratio of the actual DDG-37 condenser. This manipulation, however, must be interpreted as another source of error and innacuracy when comparing CONDIP's condenser performance with the experimental results.

Lastly, condenser designs in [Ref. 15] reveal that three different tube patterns were employed in the DDG-37 condenser. In addition, two different values for tube pitch were used - a pitch of 1.4 in the condenser bundle and a pitch of 1.3 in the cooler. This situation cannot be duplicated in CONDIP. Therefore a constant pitch of 1.40 and a uniform tube pattern were utilized throughout the condenser.

The design approximations utilized in CONDIP to try to geometrically simulate the actual DDG-37 condenser introduce significant uncertainty into subsequent design comparisons. This, coupled with the fact that the data collected is also suspect, would imply that it is rather difficult to verify CONDIP's analysis with the information available. It should also be noted that CONDIP is sensitive to even small variance in either the data collected (i.e. steam inlet temperature) or the approximated design variables. However, despite the above-mentioned problems associated with equating CONDIP's condenser to the DDG-37 condenser, the experimental data obtained from the DDG-37 condenser still provide the best available base upon which to make a reasonable determination of CONDIP's capabilities and limitations.

In comparing the results in Table II, it is immediately clear that there is significant difference between certain condenser performance parameters predicted by CONDIP and the corresponding experimentally derived condenser performance. Already, much has been said about the numerous geometric

approximations used to model the DDG-37 condenser. But questionable data and geometric manipulations may not completely explain the 13 percent exit-fraction and general poor performance generated in CONDIP's analysis. Its values for the average overall heat transfer coefficient and heat rejected were significantly lower than the experimental results. One source of the problem lies in the actual heat transfer analysis performed in the code. Lynch [Ref. 18] graphically illustrated how sensitive this analysis is to the effects of condensate inundation. In particular, by making small changes - within the allowable ranges - in the constants used in Eissenberg's correlations for condensate rain, significant improvement could be realized in the overall heat transfer characteristics of the condenser. CONDIP's results, when compared to the experimental data support the argument that the values currently used in the inundation correlations are rather conservative in nature, and cause the overall analysis to yield a poor performance for the given steam load and condenser design.

Therefore, in order to present CONDIP with a fair test to determine its credibility as a design predictor, some additional work must be first accomplished. A condenser geometrically identical to the general model created in CONDIP should be constructed with complete and accurate data acquisition systems to establish a thorough data base from which to compare. Also, more research should be performed on the effects of condensate inundation and velocity shear to obtain more precise correlations in determining their overall effects on the film heat transfer coefficients.

One last additional point should be mentioned. In comparing the steam-side pressure drops through the condenser, it was shown that CONDIP's pressure losses were nearly 72 percent larger than the actual physical measurements. However this radical difference is mainly due to the

high pressure losses experienced at the entrance of the cooler as a large volume of steam tried to force its way through the small available area. Therefore, the significance of this large disagreement in results is relatively minor and can be treated simply as a consequence of the more important heat transfer limitations in the comparison run.

Although the goal of verifying CONDIP as a design predictor has proven elusive, it was still possible to demonstrate its capabilities through comparison studies. Therefore, the remaining emphasis in this thesis is to show the ability of CONDIP (in combination with the optimizer COPES/CONMIN) to take an initial design with a given framework of constraints and design variables, and obtain better designs based on a desired objective function.

B. EXPLANATION OF THE CASE STUDIES

The following case studies were devised to best exercise the capabilities of CONDIP. They were made as realistic as possible so as to simulate the problem of condenser design and specification confronting the engineer during the early stages of power plant design. The condenser performance returned by CONDIP during the verification run and contained in Table II will serve as a baseline for comparing the results of each case study. The baseline condenser performance is based on the design parameters from the DDG-37 condenser listed in Table I. It was stated earlier that CONDIP's optimization results are slightly sensitive to the initial design if more than three design variables are used. Since all the cases involve eight or more design variables it would be best, for the purposes of comparison, to start from the same initial design in all cases. Therefore, the initial design variables used for the verification run and contained in Table I will be utilized as the baseline

design. Although many of these initial design variables will be allowed to change during optimization, certain basic condenser requirements will not. They include: steam flow into the condenser, inlet steam saturation pressure and temperature, cooling water injection temperature, the fraction of non-condensable gases in the steam, the tube fouling factor, and the tube material. It should be noted that although there was an initial value for row spacing given in Table I, row spacing was not used as a design variable during any of the optimizations. Instead, the program used the default method of row spacing calculation available in the code where the rows are spaced such that a 60-degree equilateral triangle pattern of concentric rows is obtained. Row spacing is therefore dependent on tube pitch and tube outer diameter by the following relation:

$$RSPA = (SDDO * ODOI) * .866 \quad (\text{eqn 5.1})$$

where RSPA is row spacing, SDDO is tube pitch, and ODOI is tube outer diameter in inches.

There are a few key points to be kept in mind when comparing the results of the case studies with the baseline. First, the baseline design is an infeasible and inadequate design. Its performance indicates that it is not capable of supporting the required steam load by returning a steam exit-fraction in excess of 13 percent. So any gains in the objective function that were realized in the case studies is even more remarkable since it is a necessary condition that the optimum design be a feasible design, defined as having an exit-fraction not greater than 1 percent. Second, the percent change referred to when analyzing the results is calculated based on the baseline design. Thus the baseline serves as a uniform frame of reference. Next, it should be noted that because of the large number of design variables

and constraints, intuition on how an optimized result will turn out is not always applicable. Finally, it will be easier to understand the effects of the various design parameters by keeping in mind the following, very basic, heat transfer correlation:

$$Q = U * A * LMTD \quad \text{(eqn 5.2)}$$

where Q is the rate of heat released as the steam condenses; U is the overall heat transfer coefficient; A is the heat transfer surface area; and $LMTD$ can be interpreted as the thermal driving force between the steam and the coolant. Q is directly dependent on steam flow and pressure into the condenser, the percentage of that steam that is condensed, and any subcooling of the condensate. For a given steam load and a very small exit-fraction Q is nearly constant as the optimized results in all the case studies will indicate.

1. Constraint Framework for CONDIP

In order to simulate an actual trade-off study, the constraints and their respective limits were kept constant for all the case studies. The condenser was to be designed with a maximum bundle diameter of ten feet, a maximum and minimum tube outer diameter of 0.625 and 1.25 inches respectively, a steam exit-fraction of not more than 1 percent, a maximum cooler inlet velocity ($VLCMAX$) of 200 feet/second, and a ratio of tube sheet hole area to total tube sheet area of less than 0.30.

The constraint on bundle diameter was chosen somewhat arbitrarily. It seems unlikely that this limit would be realistically exceeded, although certainly space requirements would dictate the exact configuration. Tube outer diameter is dependent on the values for tube wall thickness and tube inner diameter. Thus the limits imposed on tube

outer diameter represent realistic restrictions on the possible combinations of inner diameter and wall thickness. These restrictions are based loosely on anticipated tube structural and strength requirements and correspond to values of normally available tubes [Ref. 19]. The maximum limit of 200 feet/second for VLCMAX was also a somewhat arbitrary but realistic limit. It is assumed that steam velocities often exceed that value in the condenser bundle.

It is recalled that steam exit-fraction will play a significant role in the determination of the final optimum design. The baseline exit-fraction of 13 percent predicted by CONDIP for the DDG-37 condenser is unsatisfactory. Therefore a more reasonable upper limit of 1 percent was placed on this constraint. Although CONDIP will return a much more conservative design if 1 percent vice 13 percent is used as the upper limit, the subsequent design will be much more credible.

Finally, the amount of tube sheet material that can be removed by drilling for the installation of condenser tubes is specified at 24 percent of the total tube sheet area in [Ref. 17]. This area ratio limit represents a structural limit imposed to ensure that the tube sheets do not fail due to heat and pressure stresses in the condenser. However, CONDIP does not take into account the space between the condenser and tube shell normally used in area ratio calculations as blank tube sheet area. For this reason and to allow more flexibility in the design analysis, the constraint limit was set at 30 percent.

In summary, the general design constraints and the associated upper and lower bounds were:

$$0.625 \leq \text{tube outer diameter (inch)} \leq 1.25$$

$$1.0 \leq \text{bundle diameter (feet)} \leq 10.0$$

$$\text{steam exit-fraction (\%)} \leq 1.0$$

$$\text{VLCMAX (ft/sec)} \leq 200.0$$

$$\text{area ratio} \leq .30$$

These design constraints and associated bounds were used in all the case studies except where specifically modified.

2. Design Variable Framework for CONDIP

At least eight design variables were used in all the case studies. They include tube inner diameter, tube wall thickness, tube pitch, the number of tubes in the condenser, tube length, the inner void radius, the percent of the tubes in the cooler, and cooling water velocity. Side constraints were placed on all of these variables to correspond to either realistic physical limits or available standardized materials.

Tube wall thickness was not allowed to fall below 0.022 inches (BWG 24) or exceed .109 inches (12 BWG), sizes normally available commercially. Tube inner diameter was restricted to values between .407 and 1.206 inches so as to yield tube outer diameters within the limits specified earlier.

Tube pitch is defined as the ratio of the center to center spacing between adjacent tubes in a row to the tube outer diameter. Tube pitch is an accurate measure of how closely packed the tube bundle is. Generally accepted values for pitch lie in the range of 1.3 to 1.7. However, to provide more latitude in the design process this design variable was allowed to vary in the range between 1.1 to 2.0.

There is no guidance available as to the allowable range for tube length in the condenser. Since the lower limit was not expected to be crucial, it was set randomly at 1.0 feet. The upper limit of 25.0 feet was a realistic limit considering the size of the tube diameter being worked with. Inner void radius and the percent of tubes in the cooler were chosen to be design variables simply to enhance the

flexibility of the code in designing the condenser model. The bounds for both variables were entirely arbitrary with only common sense as the determining factor. The upper and lower limits on the percent of tubes in the cooler was established as 10.0 and 2.0 percent respectively. The upper and lower bounds on the inner void radius was set at 1.0 and 0.1 feet.

Cooling water velocity generally ranges from three to nine feet per second in value for all common tube materials, except titanium which has an upper limit of 15 feet per second. Exceeding these upper limits risks excessive tube erosion and material damage. Finally the number of tubes was permitted to vary between 1000 and 8000 tubes for the purpose of improving design flexibility. It is extremely unlikely that, for most propulsion applications, tube number would fall below 1000. The upper limit was simply chosen as a realistic cutoff point in terms of complexity, cost and maintainability.

In summary, the general design variables and the associated side-constraints were:

$$0.407 \leq \text{tube inner diameter (inches)} \leq 1.206$$

$$0.022 \leq \text{tube thickness (inches)} \leq 0.109$$

$$2.0 \leq \text{percent of tubes in cooler} \leq 10.0$$

$$0.10 \leq \text{inner void radius (feet)} \leq 1.0$$

$$3.0 \leq \text{coolant velocity (ft/sec)} \leq 9.0$$

$$1.0 \leq \text{tube length (feet)} \leq 25.0$$

$$1000 \leq \text{tube number} \leq 8000$$

$$1.1 \leq \text{tube pitch} \leq 2.0$$

As in the case of design constraints, these design variables and their respective limits were used consistently in all the case studies unless otherwise specified.

C. CASE STUDIES USING CONDIP

1. Case One

The objective of this case was to minimize condenser volume. The final results of the optimization along with the initial parameters is listed in Table III.

These results show a 16 percent decrease in condenser volume with a corresponding 24 percent increase in pumping power. The source of the improvement can be understood by noting the following:

1) Tube wall thickness was reduced from 0.049 to 0.022, the minimum side-constraint, thus allowing tube inner diameter to increase while maintaining a minimum tube outer diameter.

2) The number of tubes shrank slightly as did tube length, resulting in a smaller heat transfer surface area.

3) Tube pitch increased markedly, causing a reduction in steam pressure losses which then ensured that high values for steam saturation pressure and temperature would be maintained throughout the condenser. The large pitch also reduced steam velocities, allowing the cooler inlet velocity limit to be satisfied. Row spacing decreased from the initial value of 1.35 inches, thus decreasing condenser volume.

4) Cooling water velocity increased to the maximum allowable value of 9 ft/sec which correspondingly resulted in larger head losses and coolant flow, causing overall pumping power to increase.

As cooling water velocity increased and tube wall thickness decreased, then their respective thermal resistances were diminished. The cumulative effect was to improve the overall heat transfer coefficient. LMTD rose primarily as a result of the higher steam temperatures throughout the condenser. It is apparent by looking at equation 5.2 that increasing the driving forces for heat

transfer, such as the overall heat transfer coefficient and LMTD, allows the heat transfer surface area to decrease. This resulted in a similar reduction in condenser volume.

The constraint limits that prevented further design improvement were the upper bound on the cooling water velocity, the upper limit on the tube sheet area and the upper limit on VLCMAX.

2. Case Two

The objective of this case was to minimize the pumping power required to overcome the tube-side head losses and drive the cooling water through the condenser tubes. The final results of the optimization are presented along with the initial design in Table IV.

The results indicate a dramatic 90 percent reduction in required pumping power with an equally large 120 percent increase in condenser volume. The major factors involved in the design improvement along with their relative effects are briefly explained below:

1) Tube inner diameter increased 27 percent while tube thickness remained relatively unchanged. Thus tube outer diameter was caused to increase.

2) The number of tubes in the condenser rose significantly, along with tube length. This, coupled with the enlarged tube outer diameter resulted in nearly doubling the heat transfer area.

3) Tube pitch increased 29 percent, which allowed steam saturation pressure and temperature to be maintained at consistently large values in the condenser. This had a benefiting effect on the associated LMTD calculation. The large pitch also helped satisfy the steam velocity limit into the cooler. The tube spacing decreased from the initial value of 1.35, but by a smaller amount than the previous case because of the large values for tube pitch and outer diameter.

4) Cooling water velocity dropped to the minimum allowable limit of 3 ft/sec. This had the effect of reducing tube-side head losses and coolant flow through the condenser. Consequently, pumping power was drastically reduced.

The combined effect of all these changes can again be put in perspective by looking at equation 5.2. For the given steam and corresponding heat load, the heat transfer area increased drastically, allowing both LMTD and the overall heat transfer coefficient to decrease. A smaller overall heat transfer implies a smaller convective tube-side contribution which in turn permits coolant velocity to reduce to its lowest allowable value. The LMTD decrease is explained by the fact that cooling water was spending more time in the tubes, thus causing the average cooling water temperature to rise. However, the subsequent reduction in LMTD was minimized by the fact that a high steam temperature was maintained in the condenser.

There were no active constraints in this design outside of cooling water velocity which prevented further design improvement. However, the penalty paid in terms of a huge condenser volume, appears prohibitive.

3. Case Three

The objective of this case was to minimize pumping power while holding condenser volume constant at the initial value of 432 cubic feet. This was a particularly interesting test case as the results in Table V bear out. The required pumping power was reduced by nearly 38 percent with no change in volume. The effects of the design changes which resulted in the design improvement are presented below:

1) Tube inner diameter increased noticeably. However, the effects of this increase on tube outer diameter was minimized by a large drop in tube wall thickness. Thus, tube outer diameter remained relatively unchanged.

2) The number of tubes experienced a minor reduction, while tube length increased. The overall effect was to increase heat transfer surface area.

3) Tube pitch again rose by nearly 25 percent, causing steam saturation temperature and pressure to maintain a nearly constant value throughout the condenser. This had a beneficial effect on the LMTD between the steam and the cooling water. The larger pitch also had the additional effect of reducing steam velocity thus allowing the subsequent design to satisfy the upper limit on steam velocity into the cooler (VLCMAX). The combination of tube pitch and tube outer diameter resulted in a reduction in row spacing from the initial value of 1.35

4) Cooling water velocity decreased by about 21 percent. This effect was manifested in subsequent pressure head, coolant flow and pumping power calculations.

Looking at equation 5.2 we see the same general pattern emerging as in Case 3, but with more subtlety in the changes. Heat transfer increased, but not at the expense of volume. Cooling water velocity was allowed to decrease while the overall heat transfer coefficient actually rose. One explanation is that as the tube wall got thinner its thermal resistance got smaller which more than offset the loss of convective heat transfer contribution from the coolant. The LMTD dropped slightly due to the higher average coolant temperature of the coolant in the tubes.

The constraints which became active and prevented further improvement in the design include tube sheet area ratio as well as tube wall thickness. However, tube wall thickness was particularly crucial because of its related effect on heat transfer.

4. Case Four

The objective of this case was to minimize condenser volume while holding pumping power constant at the initial value of 55.7 horsepower. The results of this case can be found in Table VI. Chosen to contrast the results in Case 4, the relative improvement in this design objective was not nearly so impressive. Condenser volume shrank by only 12 percent. An explanation of the causes and effects is provided below:

1) Tube inner diameter increased, with a corresponding decrease in tube wall thickness to yield the minimum allowable tube outer diameter.

2) The number of tubes decreased noticeably, tending minimize bundle volume. Note, there was only slight increase in tube length. The overall effect was to similarly reduce heat transfer surface area as condenser volume decreased.

3) Tube pitch again increased significantly, having the same effects on steam pressure, temperature and steam velocity into the cooler as discussed earlier. A large tube pitch benefits the LMTD between the steam and the cooling water. Row spacing was again a factor in reducing condenser volume as before.

4) Cooling velocity decreased slightly as did head loss. But overall coolant flow increased due to an increase in tube inner diameter. The net effect was to maintain pumping power.

Again, referring to equation 5.2 , it is clear that the slight decrease in heat transfer area was offset by the slight rise in LMTD resulting from higher condenser steam temperatures. The significant improvement in overall heat transfer coefficient, therefore, is what makes the heat balance work. The large decrease in tube wall thickness and corresponding reduction in thermal resistance contributed heavily to this improvement.

There were several constraint limits which prevented any additional objective optimization. They include the minimum tube wall thickness, tube sheet area ratio and steam velocity entering the cooler.

5. Case Five

The objective of this case was to minimize condenser volume while exercising CONDIP's capabilities to linearly vary tube pitch and tube inner diameter by row. Thirty-five rows were used, which was the identical number as the initial design. Tube pitch and tube inner diameter of both the outermost and innermost rows served as design variables in this case. However - because tube number is now a dependent variable based on the number of rows, tube pitch, and tube diameter - it could not be used as a design variable. The optimized results of this analysis along with the initial design are presented in Table VII.

The results of this test case indicate a condenser volume which is 20 percent smaller than the initial design as compared to a 16 percent decrease in Case 1. The basic reasons and explanations as to why volume was able to be reduced remain fundamentally the same as in Case 1. Attention will therefore be focussed on the effects of linearly varying pitch and tube diameter. Final tube pitch ranged in value from 1.75 in the inner row to 1.44 in the outer row. Similarly, tube inner diameter ranged from .729 in the inner row to .583 in the outer row.

It is believed that smaller pitch and inner diameter were used in the outer row because of the higher available steam saturation pressure and temperature. The resulting higher steam velocities enhanced the beneficial effects of vapor shear on the external heat transfer coefficient thereby improving heat transfer on the outer rows. As steam pressure decreased, then tube pitch and tube inner diameter

increased to compensate and extract all the available heat from the steam. Consequently, steam velocity decreased and was able to satisfy to the limit imposed on the cooler entrance velocity. The end result is a condenser geometry that makes complete use of the available resources and conforms to the geometry to take advantage of the thermal conditions in the condenser. The big limitation with this approach is that the number of rows is held constant. Thus the subsequent condenser is designed around that value and the subsequent optimum design is a function of the number of rows specified.

TABLE I
Input Design Data

PARAMETER	VALUE
Total number of tubes	5230
Tube length (feet)	10.3
Tube inner diameter (inches)	0.527
Tube wall thickness (inches)	0.049
Tube outer diameter (inches)	0.625
Tube mat'l thermal conductivity btu/(ft-hr-°F)	26.0
Tube pitch	1.38
Percent of tubes in the cooler	7.0
Steam inlet flow (lbm/hr)	161,961
Fraction of non-condensable gas (ppm)	37.1
Steam inlet pressure (psia)	1.294
Steam inlet temperature (°F)	110.52
Coolant inlet velocity (ft/sec)	8.473
Coolant inlet temperature (°F)	75.66
Fouling factor	.0002
Inner void diameter (feet)	1.1
Row spacing (inches)	1.35

TABLE II
CONDIP Verification Results

PARAMETER	EXPERIMENT RESULTS	CONDIP RESULTS	CHANGE (%)
Heat transfer area (sq.ft.)	8805	8814	+0.10
Overall heat transfer coefficient btu/(hr-sq.ft.-°F)	635.2	547.9	-9.5
Log mean temperature difference (°F)	28.24	28.62	+1.3
Coolant temperature rise (°F)	10.61	9.90	-6.7
Coolant mass flow rate (10 ⁷ lbm/hr)	1.503	1.540	+2.5
Condenser volume (cu.ft.)	-----	432.3	-----
Bundle diameter (ft)	5.7 7.2	7.17	-----
Shell-side pressure drop (psia)	0.751	1.29	+71.8
Steam exit-fraction (% of input)	-----	13.3	-----
Heat rejected (10 ⁸ btu/hr)	1.595	1.451	-9.03
Area ratio	0.291	0.266	-8.59

TABLE III
Volume Minimization

PARAMETER	BASELINE RESULTS	OPTIMIZED RESULTS	CHANGE (%)
Total number of tubes	5230	5117	-2.2
% of tubes in cocler	7.0	7.01	+0.1
Tube length (ft)	10.3	9.92	-3.7
Tube inner diam. (in)	.527	.582	+10.4
Tube wall thick. (in)	.049	.022	-55.1
Tube outer diam. (in)	.625	.626	+0.2
Tube pitch	1.4	1.73	+23.6
Void diameter (ft)	1.10	1.34	+21.8
Bundle diameter (ft)	7.17	6.59	-8.1
Condenser volume (cu.ft.)	432.3	362.2	-16.2
Area ratio	0.266	0.300	+12.7
Coolant inlet vel. (ft./sec)	8.473	9.00	+6.2
Coolant mass flow rate (10^7 lbm/hr)	1.540	1.955	+26.9
Head loss (ft H ₂ O)	7.35	7.18	-2.3
Pumping power (hp)	55.69	68.99	+23.9
Coolant temperature rise (°F)	9.90	9.00	-9.1
Log mean temperature difference (°F)	28.62	29.13	+1.8
Heat transfer area (sq.ft.)	8814.	8320.	-5.6
Average overall heat transfer coefficient btu/(hr-sq.ft.-°F)	574.9	689.9	+20.0
Steam exit-fraction (% of input)	13.3	0.0	-100.
Heat rejected (10^8 btu/hr)	1.451	1.672	+15.2

TABLE IV
Power Minimization

PARAMETER	BASELINE RESULTS	OPTIMIZED RESULTS	CHANGE (%)
Total number of tubes	5230	6393	+22.2
% of tubes in cocler	7.0	7.4	+5.7
Tube length (ft)	10.3	13.34	+29.5
Tube inner diam. (in)	.527	.667	+26.6
Tube wall thick. (in)	.049	.0455	-7.1
Tube outer diam. (in)	.625	.758	+21.3
Tube pitch	1.4	1.812	+29.4
Void diameter (ft)	1.10	1.13	+2.7
Bundle diameter (ft)	7.17	9.25	+29.0
Condenser volume (cu.ft.)	432.3	964.4	+123.1
Area ratio	0.266	0.277	+4.1
Coolant inlet vel. (ft./sec)	8.473	3.00	-64.6
Coolant mass flow rate (10^7 lbm/hr)	1.540	1.067	-30.7
Head loss (ft H ₂ O)	7.35	1.11	-84.9
Pumping power (hp)	55.69	5.84	-89.5
Coolant temperature rise (°F)	9.90	16.3	+64.6
Log mean temperature difference (°F)	28.62	24.82	-13.3
Heat transfer area (sq.ft.)	8814.	16,921	+92.0
Average overall heat transfer coefficient btu/(hr-sq.ft.-°F)	574.9	393.8	-31.5
Steam exit-fraction (% of input)	13.3	1.0	-92.5
Heat rejected (10^8 btu/hr)	1.451	1.654	+14.0

TABLE V
Power Minimization With Volume Constant

PARAMETER	BASELINE RESULTS	OPTIMIZED RESULTS	CHANGE (%)
Total number of tubes	5230	5062	+3.2
% of tubes in cocler	7.0	6.7	-4.3
Tube length (ft)	10.3	11.39	+10.6
Tube inner diam. (in)	.527	.594	+12.7
Tube wall thick. (in)	.049	.022	-55.1
Tube outer diam. (in)	.625	.638	+2.1
Tube pitch	1.4	1.753	+25.2
Void diameter (ft)	1.10	1.14	+3.6
Bundle diameter (ft)	7.17	6.73	-6.1
Condenser volume (cu.ft.)	432.3	431.6	-0.2
Area ratio	0.266	0.297	+11.7
Coolant inlet vel. (ft./sec)	8.473	6.74	-20.5
Coolant mass flow rate (10^7 lbm/hr)	1.540	1.508	-2.1
Head loss (ft H ₂ O)	7.35	4.69	-36.2
Pumping power (hp)	55.69	34.77	-37.6
Coolant temperature rise (°F)	9.90	11.6	+17.2
Log mean temperature difference (°F)	28.62	27.63	-3.5
Heat transfer area (sq.ft.)	8814.	9631.4	+9.3
Average overall heat transfer coefficient btu/(hr-sq.ft.-°F)	574.9	627.3	+9.1
Steam exit-fraction (% of input)	13.3	0.0	-100.
Heat rejected (10^8 btu/hr)	1.451	1.669	+15.0

TABLE VI
Volume Minimization With Power Constant

PARAMETER	BASELINE RESULTS	OPTIMIZED RESULTS	CHANGE (%)
Total number of tubes	5230	4867	-6.9
% of tubes in cocler	7.0	7.2	+2.9
Tube length (ft)	10.3	10.92	+6.0
Tube inner diam. (in)	.527	.581	+10.2
Tube wall thick. (in)	.049	.022	-55.1
Tube outer diam. (in)	.625	.625	0.0
Tube pitch	1.4	1.739	+24.2
Void diameter (ft)	1.10	1.20	+9.1
Bundle diameter (ft)	7.17	6.41	-10.6
Condenser volume (cu.ft.)	432.3	378.4	-12.4
Area ratio	0.266	0.299	+12.4
Coolant inlet vel. (ft./sec)	8.473	8.27	-2.4
Coolant mass flow rate (10^7 lbm/hr)	1.540	1.701	+10.5
Head loss (ft H ₂ O)	7.35	6.67	-9.3
Pumping power (hp)	55.69	55.80	+0.2
Coolant temperature rise (°F)	9.90	10.34	+4.4
Log mean temperature difference (°F)	28.62	28.38	-0.8
Heat transfer area (sq.ft.)	8814.	8697.	-1.3
Average overall heat transfer coefficient btu/(hr-sq.ft.-°F)	574.9	677.4	+17.5
Steam exit-fraction (% of input)	13.3	0.0	-100.
Heat rejected (10^8 btu/hr)	1.451	1.672	+15.2

TABLE VII
Volume Minimization With Linear Variations

PARAMETER	BASELINE RESULTS	OPTIMIZED RESULTS	CHANGE (%)
Total number of tubes	5230	5348	+2.3
% of tubes in cocler	7.0	7.3	+4.3
Tube length (ft)	10.3	8.7	-15.5
Tube inner diam. (in)	.527	*.729 .583	----
Tube wall thick. (in)	.049	.022	-55.1
Tube outer diam. (in)	.625	*.773 .627	----
Tube pitch	1.4	*1.75 1.44	----
Void diameter (ft)	1.10	1.28	+16.4
Bundle diameter (ft)	7.17	6.8	-5.2
Condenser volume (cu.ft.)	432.3	345.8	-20.0
Area ratio	0.266	0.298	+12.0
Coolant inlet vel. (ft./sec)	8.473	9.0	+6.2
Coolant mass flow rate (10^7 lbm/hr)	1.540	2.48	+61.0
Head loss (ft H ₂ O)	7.35	5.84	-20.5
Pumping power (hp)	55.69	71.1	+27.7
Coolant temperature rise (°F)	9.90	7.1	-28.8
Log mean temperature difference (°F)	28.62	30.14	+5.3
Heat transfer area (sq.ft.)	8814.	7746.	-12.1
Average overall heat transfer coefficient btu/(hr-sq.ft.-°F)	574.9	713.1	+24.0
Steam exit-fraction (% of input)	13.3	0.5	-96.2
Heat rejected (10^8 btu/hr)	1.451	1.665	+14.7

* Inner row values followed by outer row values.

VI. CONCLUSIONS

The intent of this research was to create a detailed condenser analysis code capable of being coupled with a numerical optimizer and to test the program to prove its versatility. An additional objective was to validate the analysis with existing data. The results of the test cases were presented in Chapter Five; the resulting conclusions are summarized here.

A. There were significant difficulties encountered in formulating the complex condenser design analysis, CONDIP, in a way that was compatible with the optimizer COPES/CONMIN. However, the majority of those problems were overcome resulting in the creation of a program which, when combined with the optimizer, is capable of taking any initial design, no matter how impractical or infeasible, and solving for an optimum solution based on a set of pre-determined constraints and design variables. There are still some minor limitations as to the degree of optimization, but the final design is usually within 10 percent of the single best optimum. In addition, the test cases indicate that as many as ten design variables and six constraints can be used simultaneously in the design optimization with CONDIP.

B. The test cases demonstrated the effectiveness of CONDIP as a design tool for not only the conceptual design of a condenser, but also in evaluating comparison studies based on any number of design variable combinations. The number of possible combinations of design objectives, design variables and design constraints implies limitless possibilities to be explored and evaluated.

C. An attempt was made to verify CONDIP with existing data with inconclusive results. Part of the blame can be placed on the rather inadequate quality and quantity of the data, but the general performance of CONDIP's condenser indicates a weakness in the analysis. As stated earlier, the source of the this weakness may be found in the correlations used for condensate inundation. The constants used in the expression for correcting shell-side heat transfer coefficients are based somewhat on conjecture. Yet they play a significant role in the overall condenser performance. Despite this limitation, the ability to optimize CONDIP's detailed analysis is a significant step forward over using the traditional and limited HEI method.

D. CONDIP incorporates features that further increase its appeal as a design tool. By possessing the ability to linearly vary pitch and tube diameter, a better understanding of how to improve condenser performance based on its configuration is realized. The capability to incorporate shell-side tube enhancement is another added plus. The possibilities that can now be investigated are limitless.

VII. RECOMMENDATIONS

In addition to the insight that this investigation has given into the generation of automated condenser design programs, it has specifically addressed the shortcomings and pitfalls which may be encountered along the way and offered possible solutions to overcome them. Presented below are recommendations for furthering the development of CONDIP as a completely versatile and accepted design program.

A. Since the weak link and the most significant unknown in condenser analysis is the effect of condensate rain in typical condenser environments, subsequent research should be devoted to investigating this phenomenon and developing more precise analytic correlations. In particular, the effects of velocity and flow direction on the condensate film should be attended to.

B. Perhaps in conjunction with the above, a test condenser should be constructed which is geometrically similar to the model proposed in CONDIP in order to physically observe and record the condenser performance. This data could then be used to either verify CONDIP or strengthen some of its analysis. In addition, this condenser should be built such that the tube bank can be arranged in any combination of pitch, tube diameter and row spacing to fully appreciate the effects of these variables.

C. A series of sensitivity studies should be conducted on CONDIP to fully exercise its capabilities and determine the relative effects of various design variables on condenser performance. Tradeoff studies similar to those performed in this research would be most beneficial to fully understand condenser behavior.

D. Additional subroutines should be created which would allow tube enhancement to be a design variable. This involves developing correlations between heat transfer enhancement and associated frictional losses. This type of relationship can be developed for both tube-side and shell-side enhancement.

E. Finally, it is recommended that additional refinement be performed on the code to increase its capability and flexibility. One such way is to somehow allow pitch, tube diameter and tube wall thickness to vary linearly by row while still allowing the number of tubes to be a design variable. The options available are limitless.

APPENDIX A

GLOSSARY

While it would most beneficial to present a complete glossary of all the variables used in CONDIP, the sheer number makes it difficult to present a comprehensive list. However, CONDIP makes liberal use of comment cards to define as many variables as possible to make the code easier to follow. Therefore, the computer listing in Appendix C is available for reference. A list of the possible design variables and constraints is provided here along with its corresponding position in the GLOBCM common block for easy reference in writing the appropriate COPES data cards. In addition, it will be specified whether these variables can be used as design constraints or design variables.

1. ALST: The length of the condenser and cooler tubes in feet. ALST is to be used only as a design variable.
2. DELWLP: The pressure difference between the inlet and outlet coclant headers of the condenser bundle in psi. DELWLP is to be used only as a design variable.
3. DELWPC: The pressure difference between the inlet and outlet coolant headers of the condenser bundle in psi. DELWPC is to be used only as a design constraint.
4. GFLOW: The mass flow rate of the coolant in lbm/hour. GFLOW cannot be used as a design variable simultanously with VELBI. Otherwise it can be used as a design variable or a design constraint.
5. SIDI: The tube inner diameter of the innermost row of

the condenser bundle in inches. SIDI is to be used only as a design variable.

7. SIDO: The tube inner diameter of the outermost row of the condenser bundle in inches. If there is no linear variation of tube inner diameter then this variable represents the tube inner diameter of the entire condenser bundle. SIDO is to be used only as a design variable.

8. PHP: The coolant pumping power in horsepower. PHP is to be used only as a design constraint.

9. RSPA: The spacing between concentric rows in the condenser bundle in inches. RSPA is to be used only as a design variable.

10. RADINS: The inner void radius of the condenser bundle in feet. RADINS is to be used only as a design variable.

11. REWI: The tube-side Reynolds number of the coolant in the innermost row of the condenser bundle. REWI is to be used only as a design constraint.

12. REWO: The tube-side Reynolds number of the coolant in the outermost row of the condenser bundle. If there is no linear variation of tube inner diameter then this variable represents the tube-side Reynolds number of the entire condenser bundle. REWO is to be used only as a constraint.

13. SDDI: Tube pitch (tube spacing/tube outer diameter) of the innermost row of the condenser bundle. SDDI is to be used only as a design variable.

14. SDDO: The tube pitch of the outermost row of the condenser bundle. If there is no linear variation of tube pitch then this variable represents the tube pitch for the entire condenser bundle. SDDO is to be used only as a design variable.

15. SLDI: Ratio of tube length to tube outer diameter of the outermost row of the condenser bundle. SLDI is to be used only as a design constraint.

16. SLDO: Ratio of tube length to tube outer diameter of the outermost row of the condenser bundle. If there is no linear variation of tube pitch then this variable represents the tube pitch for the entire condenser bundle. SLDO is to be used only as a design constraint.

17. VELBI: The velocity of the coolant in feet/sec. VELBI cannot be used as a design variable simultaneously with GFLOW. Otherwise it can be used as either a design constraint or as a design variable.

18. XW1: The ratio of tube thickness to tube inner diameter. XW1 can be used only as a design variable.

19. XW2: Tube thickness in inches. XW2 is to be used only as a design variable. XW2 and XW1 cannot be used simultaneously.

20. VOL1: The overall condenser and cooler volume in cubic feet. VOL1 is to be used only as a design constraint.

21. VOL2: The volume occupied by the tube bank, excluding the volume of the inner void, in cubic feet. VOL2 is to be used only as a design constraint.

22. TNOTOT: The total number of tubes in the condenser and cooler combined. If Option 1 is being used then TNOTOT is to be used only as a design constraint. If Option 2 is being used then TNOTOT is to be used only as a design variable.

23. BNDRAD: The condenser bundle in feet. BNDRAD is to be used only as a design constraint.

24. ARATIO: The ratio of the total cross-sectional area of

the tubes (based on the tube outer diameter) to the tube sheet area. ARATIO is to be used only as a design constraint.

25. ODII: The tube outer diameter of the innermost row in inches. ODII is to be used only as a design constraint.

26. ODOI: The tube outer diameter of the outermost row in inches. If there is no linear variation of tube inner diameter then this variable represents the tube outer diameter of the entire condenser bundle. ODOI is to be used only as a design constraint.

27. VLCMAX: The maximum allowable steam velocity into the cooler. VLCMAX can be used only as a design constraint and only when a cooler is being designed in the system.

28. PRCCLR: The percent of the total number of tubes in the cooler. PRCCLR can be used only as a design variable.

APPENDIX B

USERS MANUAL FOR CONDIP

This appendix describes the data cards that are necessary in order to couple any design program with COPES/CONMIN. Also described are cards illustrating data input required by CONDIP to initiate analysis. Thus, the data is divided into the COPES/CONMIN program section and the CONDIP-based condenser design program section.

The COPES data is segmented into "blocks" for convenience. All formats are alphanumeric for title, end and stop cards; F10 for real data; and I10 for integer data. The formatted input may be overridden by inserting commas between data entries. Comment cards may be inserted anywhere in the data stack prior to the end card and are identified by a dollar sign (\$) in column 1. The COPES data stack must terminate with an end card containing the word "END" in column 1-3. It should be noted that information pertaining only to single analysis and optimization is presented here. Information concerning the other options available in COPES along with further explanation of COPES capabilities can be found in [Ref. 13].

The analysis data is also segmented into blocks for convenience and they begin immediately following the "END" card in the COPES data. No comment cards are permitted here, and the analysis data stack must terminate with the word "STOP" in columns 1-4. This is where the initial design values are placed for entry into CONDIP.

Default values are recommended for use in the following COPES data cards unless otherwise noted. It is recommended that these values in the COPES data blocks be used until the user becomes familiar with the program. In addition a

sample data input is illustrated in figure B.1 at the end of this appendix.

DATA BLOCK A

DESCRIPTION: COPES Title Card

FORMAT: 20A4

1	2	3	4	5	6	7	8
TITLE							

REMARKS:

- 1) This line is available for a brief description.

DATA BLOCK B

DESCRIPTION: COPES Program Control Parameters

FORMAT: 7I10

1	2	3	4	5	6	7	8
NCALC	NDV						

FIELD

CONTENTS

- | | | |
|---|--------|--|
| 1 | NCALC: | Calculation control |
| 0 | | Read input and stop. Data of blocks A-B is required. Remaining data is optional. |
| 1 | | One cycle through the program. Data of blocks A-B is required. Remaining data is optional. |
| 2 | | Optimization. Data of blocks A-I is required. Remaining data is optional. |
| 2 | NDV: | Number of independent design variables in optimization or optimum sensitivity study. |

REMARKS:

- 1) Field 1 determines program execution
- 2) Fields 3-8 are to be left blank for the CONDIP application of COPES/CONMIN.

DATA BLOCK C

DESCRIPTION: COPES Integer Optimization Control Parameters

FORMAT: 8I10

1	2	3	4	5	6	7	8
IPRINT	ITMAX	ICNDIR	NSCAL	ITRM	LINOBJ	NACMX1	NFDG

FIELD

CONTENTS

- 1 IPRINT: Print control used in optimization program, CONMIN.
0 No print during optimization.
1 Print initial and final optimization information.
2 Print above plus function value and design variable values at each iteration.
3 Print above plus constraint values, direction vector and move parameter at each iteration.
4 Print above plus gradient information.
5 Print above plus each proposed design vector, objective function and constraints during the one-dimensional search. required. Remaining data is optional.
- 2 ITMAX: Maximum number of optimization iterations allowed. DEFAULT = 20.
- 3 ICNDIR: Conjugate direction restart parameter. DEFAULT = NDV+1.
- 4 NSCAL: Scaling parameter. GT.0 - Scale design variables to order of magnitude one every NSCAL iterations. LT.0 - Scale design variables according to scaling values input. DEFAULT = No scaling.
- 5 ITRM: Number of subsequent iterations which must satisfy relative or absolute convergence criterion before optimization process is terminated. DEFAULT = 3.
- 6 LINOBJ: Linear objective function identifier. If the optimization objective is known to be a linear function of the design variables, set LINOBJ = 1. DEFAULT = Non-Linear.
- 7 NACMX1: one plus the maximum number of active constraints anticipated. DEFAULT = NDV+2.

DATA BLOCK C (Continued)

<u>FIELD</u>	<u>CONTENTS</u>
8	NFDG: Finite difference gradient identifier.
0	All gradient information is computed by finite difference.
1	Gradient of objective is computed anaytically. Gradients of constraints are computed by finite difference.
2	All gradient information is computed analytically

REMARKS:

1) The value of NSCAL = 0 is suggested and ITRM = NACMX1 = 0 should be used.

2) The value of IPRINT may be reduced when the user becomes familiar with the optimization output.

3) The default values will be used if the card is either left blank or a value of zero is entered.

4) Because of the complexity of the problem it is necessary to have a large value for ITMAX so the problem will not be terminated prematurely. Recommended value is ITMAX = 40

5) The complexity of the condenser analysis ensure that no function can be considered linearly dependent on any combination of variables. This justifies using the DEFAULT value for LINOBJ.

DATA BLOCK D

DESCRIPTION: COPES Floating Point Optimization Program
Parameters

FORMAT: 8F10

1	2	3	4	5	6	7	8
FDCH	FDCHM	CT	CTMIN	CTL	CTLMIN	THETA	PHI

FIELD

CONTENTS

- | | | |
|---|---------|---|
| 1 | FDCH: | Relative change in design variables in calculating finite difference gradients.
DEFAULT = 0.01 |
| 2 | FDCHM: | Minimum absolute step in finite difference gradient calculations. DEFAULT = 0.001. |
| 3 | CT: | Constraint thickness parameter.
DEFAULT = -0.1. |
| 4 | CTMI: | Minimum absolute value of CT considered in the optimization process.
DEFAULT = 0.004 |
| 5 | CTL: | Constraint thickness parameter for linear and side constraints. |
| 6 | CTLMIN: | Minimum absolute value of CTL considered in the optimization process.
DEFAULT = 0.001 |
| 7 | THETA: | Mean value of push-off factor in the method of feasible directions.
DEFAULT = 1.0 |
| 8 | PHI: | Participation coefficient, used if one or more constraints are violated.
DEFAULT = 5.0. |

DATA BLOCK D (continued)

FORMAT: 2F10

1	2	3	4	5	6	7	8
DELFUN	DABFUN						

FIELD

CONTENTS

- | | | |
|---|---------|--|
| 1 | DELFUN: | Minimum relative change in objective function to indicate convergence of optimization process. DEFAULT = 0.001. |
| 2 | DABFUN: | Minimum absolute change in objective function to indicate convergence of the optimization process.
DEFAULT = 0.001 times the initial objective value. |

REMARKS:

1) Note that data for Data Block D is entered on two separate cards. A blank card indicates the default value is to be used.

2) If the NDV is greater than 3, the recommended value for FDCH is between 0.05 and 0.10.

DATA BLOCK E

DESCRIPTION: Total Number of Design Variables, Design Objective Identification and Sign on Design Objective.

FORMAT: 2I10, F10

1	2	3	4	5	6	7	8
NDVTOT	IOBJ	SGNOPT					

FIELD

CONTENTS

- | | | |
|---|---------|---|
| 1 | NDVTOT: | Total number of variables linked to the design variables. NDVTOT must be greater or equal to NDV. This option allows two or more parameters to be assigned to a single design variable. The value of each parameter is the value of the design variable times a multiplier which may be different for each parameter.
DEFAULT = NDV. |
| 2 | IOBJ: | Global variable number associated with objective function in optimization or optimum sensitivity analysis. |
| 3 | SGNOPT: | Sign used on objective of optimization to identify whether function is to be maximized or minimized. +1.0 indicates maximization; -1.0 indicates minimization.
DEFAULT = -1.0 |

REMARKS:

1) Currently there are not any variables in CONDIP which are linked to any of the design variables. Therefore the DEFAULT value is used for NDVTOT.

DATA BLOCK F

DESCRIPTION: Design Variable Bounds, Initial Values, and Scaling Factors.

FORMAT: 4F10

1	2	3	4	5	6	7	8
VLB	VUB	X	SCAL				

FIELD

CONTENTS

- | | | |
|---|-------|---|
| 1 | VLB: | Lower bound on the design variable. |
| 2 | VUB: | Upper bound on the design variable. |
| 3 | X: | Initial value of the design variable.
If X is non-zero, this will supercede the
value initialized by subroutine ANALIZ. |
| 4 | SCAL: | Design variable scale factor. Not
used if NSCAL \geq 0 in Block C |

REMARKS:

- 1) There must be one separate data card for each design variable. Therefore there will be NDV data cards.
- 2) For all applications with CONDIP, initial values for the design variables will be entered through the INPUT subroutine called in ANALIZ.

DATA BLOCK G

DESCRIPTION: Design Variable Identification

FORMAT: 2I10,F10

1	2	3	4	5	6	7	8
NDSGN	IDSGN	AMULT					

FIELD

CONTENTS

- | | | |
|---|--------|---|
| 1 | NDSGN: | Design variable number associated with the variable. |
| 2 | IDSGN: | Global variable number associated with the variable. |
| 3 | AMULT: | Constant multiplier on the variable. The value of the variable will be the value of the design variable, NDSGN, times AMULT. DEFAULT = 1.0. |

REMARKS:

1) There must be one separate card for each of the NDVTOT design variables. These data cards must follow the same order as the corresponding design variable parameter cards in Block F.

DATA BLOCK H

DESCRIPTION: Number of Constrained Parameters.

FORMAT: I10

1	2	3	4	5	6	7	8
NCONS							

FIELD

CONTENTS

1 NCONS: Number of constraint SETS in the
 optimization problem.

REMARKS:

1) If two or more adjacent parameters in the Global common block have the same limits imposed, these are part of the same constraint set.

DATA BLOCK I

DESCRIPTION: Constraint Identification and Bounds.

FORMAT: 3I10

1	2	3	4	5	6	7	8
ICON	JCON	LCON					

FIELD

CONTENTS

- | | | |
|---|-------|--|
| 1 | ICON: | First Global number corresponding to the constraint set. |
| 2 | JCON: | Last Global number corresponding to the constraint set. DEFAULT = ICON. |
| 3 | LCON: | Linear constraint identifier for this set of constrained variables.
LCCN = 1 indicates linear constraints.
DEFAULT = 0 = Nonlinear constraint. |

REMARKS:

- 1) In CONDIP there is only Global number and thus one constraint that comprise a constraint set. Therefore the DEFAULT value is used for JCON.
- 2) All the constraints in this analysis are nonlinear. The DEFAULT value was therefore used for LCON as well.
- 3) This is the first card of a two card set which must be read together.

DATA BLOCK I (Continued)

FORMAT: 4F10

1	2	3	4	5	6	7	8
BL	SCAL1	BU	SCAL2				

FIELD

CONTENTS

- | | | |
|---|--------|---|
| 1 | BL: | Lower bound on the constrained variables.
Value less than $-2.0E+15$ is assumed
unbounded |
| 2 | SCAL1: | Normalization factor on lower bound.
DEFAULT = Max of ABS(BL) or 0.1. |
| 3 | BU: | Upper bound on the constrained variables.
Value greater than $+2.0E+15$ is assumed
unbounded. |
| 4 | SCAL2 | Normalization factor on upper bound.
DEFAULT = Max of ABS(BU) or 0.1. |

REMARKS:

1) The normalization factor can usually be defaulted, with the notable exception of exit-fraction. the normalization factor used for this constraint is usually ten times the upper bound.

DATA BLOCK P

DESCRIPTION: COPES Data 'END' Card.

FORMAT: 3A1

1	2	3	4	5	6	7	8
END							

FIELD

CONTENTS

1 The word 'END' in column 1-3.

REMARKS:

- 1) This card must appear at the end of the COPES data.
- 2) This ends the COPES input deck.

DATA BLOCK AA

DESCRIPTION: Geometry Option

FORMAT: I5

1	2	3	4	5	6	7	8
IOPT							

FIELD

CONTENTS

- | | | |
|---|------|--|
| 1 | IOPT | Two condenser geometric options |
| | 1 | IOPT = 1. Number of condenser rows is a input as a constant; the number of tubes is a dependent variable. Use data blocks EE, FF, and GG. |
| | 2 | IOPT = 2. Number of tubes is allowed to be an independent variable and the number of rows is a dependent variable. Use data blocks HH and II. DEFAULT value is IOPT = 2. |

REMARKS:

1) Data is right-justified and blanks will be interpreted as zeros.

2) If IOPT = 1, a smaller finite difference (FDCH) can be utilized in data Block D. This is because with this option the design analysis is less sensitive to the problems discussed earlier. Recommended using the DEFAULT value of 0.01 for FDCH.

DATA BLOCK BB

DESCRIPTION: Condenser Orientation

FORMAT: 1I5,3F10

1	2	3	4	5	6	7	8
ISEC	SECWID	PHI	PRCCLR				

FIELD

CONTENTS

1	ISEC	The number of sectors in the condenser.
2	SECWID	Sector width in degrees of arc.
3	PHI	Symmetry angle measure from the vertical.
4	PRCCLR	The percent of the tubes in the cooler.

REMARKS:

- 1) Data BB is required, no matter what geometry option is chosen.
- 2) The only limitation on ISEC and SECWID is that their product is less than 360 degrees. If the product is exactly 360 degrees, certain trigonometric functions will return a singularity.
- 3) PHI is that angle from the vertical that cuts the condenser in half.
- 4) Data is right-justified and blanks will be interpreted as zeros.

DATA BLOCK CC

DESCRIPTION: Void Size, Tube Length, Row Spacing

FORMAT: 3F10

1	2	3	4	5	6	7	8
RADINS	ALST	RSPA					

FIELD

CONTENTS

1	RADINS	The inner void radius; feet.
2	ALST	The tube length; feet.
3	RSPA	Concentric row spacing about the void; inches.

REMARKS:

1) Data CC is required, no matter what geometry option is chosen.

DATA BLOCK DD

DESCRIPTION: Tube Material Parameters

FORMAT: I5,4F10

1	2	3	4	5	6	7	8
IWALL	XW	TUBESW	SKW	FOUL			

FIELD

CONTENTS

1	IWALL	A Flag indicating the tube thickness specification. 1 IWALL = 1. Tube thickness is input as ratio of tube thickness to tube inner diameter. 2 IWALL = 2. Tube thickness is input in inches.
2	XW	The input for wall thickness, dependent on the value for IWALL.
3	TUBESW	Specific weight of the tube material; lbm/(cu.ft.)
4	SKW	Tube material thermal conductivity; (btu-ft)/(sq.ft.-hr-°F)
5	FOUL	Tube fouling factor.

REMARKS:

- 1) Data DD is required, no matter what geometry option is chosen.
- 2) Data is right-justified and blanks will be interpreted as zeros.

DATA BLOCK EE

DESCRIPTION: Number of Rows

FORMAT: I5

1	2	3	4	5	6	7	8
NOROWS							

FIELD

CONTENTS

1	NOROWS	The number of concentric rows in the condenser bundle built around the center void
---	--------	--

REMARKS:

- 1) Data EE is used only when IOPT = 1
- 2) Data is right-justified and blanks will be interpreted as zeros.

DATA BLOCK FF

DESCRIPTION: Tube Inner Diameter

FORMAT: 1I5,2F10

1	2	3	4	5	6	7	8
MDIAM	SIDO	SIDI					

FIELD

CONTENTS

1	MDIAM	A flag to indicate whether tube inner diameter will linearly vary by row through the condenser bundle. 1 MDIAM = 1. Tube inner diameter is uniform through the condenser bundle. 2 MDIAM = 2. Tube inner diameter varies linearly through the bundle by row.
2	SIDO	Tube inner diameter of the outer row; inches.
3	SIDI	Tube inner diameter of the inner row; inches.

REMARKS:

- 1) Data FF is used only when IOPT = 1
- 2) Data is right-justified and blanks will be interpreted as zeros.
- 3) Cooler tubes use the inner diameter of the innermost bundle row.
- 4) The DEFAULT value is MDIAM = 1

DATA BLOCK GG

DESCRIPTION: Tube Pitch

FORMAT: 1I5,2F10

1	2	3	4	5	6	7	8
MPITCH	SDDO	SDDI					

FIELD

CONTENTS

1	MPITCH	A flag to indicate whether tube pitch will linearly vary by row through the condenser bundle. 1 MPITCH = 1. Tube pitch is uniform through the condenser bundle. 2 MPITCH = 2. Tube pitch varies linearly through the bundle by row.
2	SDDO	Tube pitch of the outer row;
3	SDDI	Tube pitch of the inner row;

REMARKS:

- 1) Data GG is used only when IOPT = 1
- 2) Data is right-justified and blanks will be interpreted as zeros.
- 3) Cooler tubes use the pitch of the innermost bundle row.
- 4) The DEFAULT value is MPITCH = 1

DATA BLOCK HH

DESCRIPTION: Tube Inner Diameter and Tube Pitch

FORMAT: 2F10

1	2	3	4	5	6	7	8
SIDO	SDDO						

FIELD

CONTENTS

1	SIDO	Tube inner diameter for the entire condenser; inches
2	SDDO	Tube pitch for the entire condenser

REMARKS:

- 1) Data HH is used only when IOPT = 2.
- 2) Data is right-justified and blanks will be interpreted as zeros.
- 3) In the calculations, SIDI is set equal to SIDO and SDDI is set equal to SDDO. This avoids the need for two systems of nomenclature for each geometry option.

DATA BLOCK II

DESCRIPTION: Total Number of Tubes in the Condenser

FORMAT: F12

1	2	3	4	5	6	7	8
TNOTOT							

FIELD

CONTENTS

1	TNOTOT	The total number of tubes in the condenser (cooler and the bundle).
---	--------	---

REMARKS:

- 1) Data II is used only when IOPT = 2.
- 2) Data is right-justified and blanks will be interpreted as zeros.

DATA BLOCK JJ

DESCRIPTION: Inlet Steam Mixture.

FORMAT: 15,2F10

1	2	3	4	5	6	7	8
JGAS	WSI	WNCIR					

FIELD

CONTENTS

1	JGAS	A Flag indicating the type of non-condensable gas entering the system. 1 JGAS = 1. This indicates that the gas is air. 2 JGAS = 2. This indicates that the gas is carbon dioxide. 3 JGAS = 3. This indicates that the gas is a mixture of the two.
2	WSI	Steam flow rate entering the condenser; lbm/hr.
3	WNCIR	Ratio of the non-condensable gas flow to inlet steam flow; lbm/hr.

REMARKS:

- 1) Data JJ is required, no matter what geometry option is chosen.
- 2) Data is right-justified and blanks will be interpreted as zeros.

DATA BLOCK KK

DESCRIPTION: Inlet Temperatures.

FORMAT: 2F10

1	2	3	4	5	6	7	8
STBI	STSAT1						

FIELD

CONTENTS

- | | | |
|---|--------|--|
| 1 | STBI | Coclant inlet temperature; °F. |
| 2 | STSAT1 | Inlet steam saturation temperature;
°F. |

REMARKS:

- 1) Data KK is required no matter, what geometry option is chosen.
- 2) Data is right-justified and blanks will be interpreted as zeros.

DATA BLOCK LL

DESCRIPTION: Cooling Water Parameters

FORMAT: 2F10

1	2	3	4	5	6	7	8
IFLOW	X5						

FIELD

CONTENTS

1	IFLOW	A control flag for cooling water specifications.
1		IFLOW = 1. Input pressure drop across cooling water headers in psia.
2		IFLOW = 2. Input cooling water velocity in ft/sec.
3		IFLOW = 3. Input coolant flow in lbm/hr.
2	X5	Actually input the value for flow into this variable. The specification for flow to be determined by IFLOW

REMARKS:

- 1) Data LL is required, no matter what geometry option is chosen.
- 2) Data is right-justified and blanks will be interpreted as zeros.
- 3) X5 acts as a temporary all-purpose storage variable for whatever expression for coolant flow is used.

DATA BLOCK MM

DESCRIPTION: Internal Enhancement Regions

FORMAT: I5

1	2	3	4	5	6	7	8
NEI							

FIELD

CONTENTS

1	NEI	Number of internal enhancement regions. A value between 1 and 6.
---	-----	---

REMARKS:

- 1) Data MM is optional. There must be NEI subsequent data cards providing the necessary parameters for each region.
- 2) enhancement can only be used if IOPT = 1.
- 3) Data is right-justified and blanks will be interpreted as zeros.
- 4) This value was zero for all runs.

DATA BLOCK NN

DESCRIPTION: Internal Enhancement Parameters

FORMAT: 2I5,3F10

1	2	3	4	5	6	7	8
NRNI	NETI	ENI	BC	BE			

FIELD

CONTENTS

1	NRNI	Row number of first row in internal enhancement region.
2	NETI	Number of tubes in each internal enhancement region.
3	ENI	Internal heat transfer enhancement factor.
4	BC	Coefficient in internally enhanced tube coolant pressure drop calculation.
5	BE	Exponent in coolant pressure drop calculation.

REMARKS:

1) Data NN is optional. However, if NEI is greater than zero then there must NEI "NN" data cards to provide the necessary data for each enhancement region.

2) enhancement can only be used if IOPT = 1.

3) Data is right-justified and blanks will be interpreted as zeros.

4) These values are constant for entire run and cannot be changed by the optimizer.

5) This value was zero for all runs.

DATA BLOCK 00

DESCRIPTION: External Enhancement Regions

FORMAT: I5

1	2	3	4	5	6	7	8
NEE							

FIELD

CONTENTS

1	NEE	Number of external enhancement regions. A value between 1 and 6.
---	-----	---

REMARKS:

- 1) Data 00 is optional. There must be NEE subsequent data cards providing the necessary parameters for each region..
- 3) Data is right-justified and blanks will be interpreted as zeros.
- 4) This value was zero for all runs.

DATA BLOCK PP

DESCRIPTION: External Enhancement Parameters

FORMAT: 2I5,2F10

1	2	3	4	5	6	7	8
NRNE	NETE	ENO	ENH				

FIELD

CONTENTS

1	NRNE	Row number of first row in external enhancement region.
2	NETE	Number of tubes in each external enhancement region.
3	ENI	External heat transfer enhancement factor.
4	ENH	Steam-side pressure drop factors.

REMARKS:

1) Data PP is optional. However, if NEE is greater than zero then there must NEE "PP" data cards to provide the necessary data for each enhancement region.

2) Data is right-justified and blanks will be interpreted as zeros.

3) Enhancement can only be used if IOPT = 1.

4) These values are constant for an entire run and cannot be changed by the optimizer.

5) This value was zero for all runs.

DATA BLOCK QQ

DESCRIPTION: Baffle Options

FORMAT: I5

1	2	3	4	5	6	7	8
IBAF							

FIELD

CONTENTS

1	IBAF	A flag to be used to determine baffle number and location.
---	------	--

REMARKS:

- 1) Data QQ is optional.
- 2) Data is right-justified and blanks will be interpreted as zeros.
- 3) These values are constant for an entire run and cannot be changed by the optimizer.
- 4) Additional specified baffles were not used in any of the runs. This value was zero for all runs.

DATA BLOCK RR

DESCRIPTION: Baffle Location

FORMAT: I5

1	2	3	4	5	6	7	8
JBAF							

FIELD

CONTENTS

1 JBAF An array containing baffle locations

REMARKS:

- 1) Data RR is optional.
- 2) Data is right-justified and blanks will be interpreted as zeros.
- 3) These values are constant for an entire run and cannot be changed by the optimizer.
- 4) Additional specified baffles were not used in any of the runs. This value was zero for all runs.

DATA BLOCK SS

DESCRIPTION: Detailed Printout

FORMAT: I5

1	2	3	4	5	6	7	8
IPRT							

FIELD

CONTENTS

1	IPRT	A flag to generate a detailed output of the condenser analysis (OUT3)
---	------	--

REMARKS:

- 1) Data SS is optional.
- 2) Data is right-justified and blanks will be interpreted as zeros.
- 3) These values are constant for an entire run and cannot be changed by the optimizer.
- 4) This value was zero for all runs.

VOLUME MINIMIZATION - CIRCULAR CONDENSOR

```

2,8
1,40,,9,,,20
.10

1,20,-1.
1.0,2.0E+15
1000.,8000.
1.1,2.
3.,9.0
.022,,109
.407,1.206
.1,1.
2.,10.
1,1
2,22
3,14
4,17
5,19
6,7
7,10
8,29
5
4
-2.0E+15,,,01,.01
23
.5,,5.
24
.1,,,30
26
.625,,1.25
28
-2.E+15,,200.
END

      2
      12  29.99   180.         7.0
0.55      .049   10.3         26.0   .0002
      2
      .527      1.40
5230.0
      1  161970. .0000371
76.66      110.52
      2  8.473
      0
      0
      0
      0

```

Figure B.1 Sample Data Input.

APPENDIX C
CONDIP LISTING

The following Appendix contains a complete listing for CONDIP. An effort has been made to make the program as readable as possible through liberal use of comment cards.

C	READ (5,400) IWALL,XW,TUBESW,SKW,FOUL	CON00970
C	IF ((IOPT.GE.2).OR.(IOPT.LT.1)) GO TO 10	CON00580
	READ (5,380) NROWS	CON00590
	READ (5,370) MDIAM,SIDO,SIDI	CON01000
	READ (5,370) MPITCH,SDDO,SDDI	CON01010
10	GO TO 20	CON01020
	READ (5,360) SIDO,SDDO	CON01030
	READ (5,390) TNOTOT	CON01040
	MPITCH=1	CON01050
	MDIAM=1	CON01060
		CON01070
20	IF (MDIAM.EQ.1) SIDI=SIDO	CON01080
	IF (MPITCH.EQ.1) SDDI=SDDO	CON01090
	IW=IWALL	CON01100
	IF (IWALL.NE.2) IW=1	CON01110
	XW1=0.	CON01120
	XW2=0.	CON01130
	IF (IW.EQ.1) XW1=XW	CON01140
	IF (IW.EQ.2) XW2=XW	CON01150
	READ (5,350) JGAS,WSI,WNCIR	CON01160
	READ (5,410) STBI,STSAT1	CON01170
	READ (5,420) IFLOW,X5	CON01180
	IFL=IFLOW	CON01190
	GFLOW=0.	CON01200
	DELBWP=0.	CON01210
	IF (IFLOW.EQ.1) DELWP=X5	CON01220
	IF (IFLOW.EQ.2) VELBI=X5	CON01230
	IF (IFLOW.EQ.3) GFLOW=X5	CON01240
	READ (5,430) NEI	CON01250
	IF (NEI.EQ.0) GO TO 40	CON01260
	IF ((NEI.LT.0).AND.(NEI.GT.6)) GO TO 120	CON01270
30	DO 30 I=1,NEI	CON01280
40	READ (5,440) NRNI(I),NETI(I),ENI(I),BC(I),BE(I)	CON01290
	READ (5,430) NEE	CON01300
	IF (NEE.EQ.0) GO TO 60	CON01310
	IF ((NEE.LE.0).OR.(NEE.GT.6)) GO TO 120	CON01320
	DO 50 I=1,NEE	CON01330
50	READ (5,450) NRNE(I),NETE(I),ENO(I),ENH(I)	CON01340
60	READ (5,430) IBAF	CON01350
	IF ((IBAF.LE.-2).OR.(IBAF.GT.ISEC)) GO TO 120	CON01360
	IF ((IBAF.EQ.0).OR.(IBAF.EQ.-1)) GO TO 80	CON01370
70	DO 70 I=1,IBAF	CON01380
80	READ (5,430) JBAF(I)	CON01390
	CONTINUE	CON01400
	READ (5,430) IPRT	CON01410
		CON01420
		CON01430
		CON01440


```

C$-----
C INP 3
C
C TEST FOR RECOVERABLE INPUT ERRORS
C
C ZZZ INP-3
C-----
IF ((IOPT.EQ.1).OR.(IOPT.EQ.2)) GO TO 90
WRITE (6,460) IOPT
WRITE (6,510)
IOPT=2
IF ((IOPT.EQ.2) GO TO 110
IF ((MDIAM.EQ.2).OR.(MDIAM.EQ.1)) GO TO 100
WRITE (6,460)
WRITE (6,480) MDIAM
WRITE (6,510)
MDIAM=1
IF ((MPITCH.EQ.1).OR.(MPITCH.EQ.2)) GO TO 110
WRITE (6,460)
WRITE (6,490) MPITCH
WRITE (6,510)
MPITCH=1
IF ((IWALL.EQ.1).OR.(IWALL.EQ.2)) GO TO 120
WRITE (6,460)
WRITE (6,500) IWALL
WRITE (6,510)
IWALL=1
C$-----
C INP 4
C
C LOCATE NON-RECOVERABLE INPUT ERRORS
C
C ZZZ INP-4
C-----
IERR=0
IF ((ISEC.GT.0).AND.(ISEC.LE.15)) GO TO 130
IERR=1
WRITE (6,520)
WRITE (6,530) ISEC
X2=SECWID*FLOAT(ISEC)
IF ((X2.LE.360.).AND.(X2.GT.0.)) GO TO 140
WRITE (6,520) X2
WRITE (6,540) ISEC,SECWID
IERR=1
IF ((IOPT.EQ.2) GO TO 150
IF ((NROWS.GT.0).AND.(NROWS.LE.99)) GO TO 160
IERR=1
WRITE (6,520)
CON01450
CON01460
CON01470
CON01480
CON01490
CON01500
CON01510
CON01520
CON01530
CON01540
CON01550
CON01560
CON01570
CON01580
CON01590
CON01600
CON01610
CON01620
CON01630
CON01640
CON01650
CON01660
CON01670
CON01680
CON01690
CON01700
CON01710
CON01720
CON01730
CON01740
CON01750
CON01760
CON01770
CON01780
CON01790
CON01800
CON01810
CON01820
CON01830
CON01840
CON01850
CON01860
CON01870
CON01880
CON01890
CON01900
CON01910
CON01920

```


C	150	WRITE (6,560) NCROWS	CON01930
		GO TO 160	CON01940
		IF (TNOTOT.GT.0.) GO TO 160	CON01950
		IERR=1	CON01960
		WRITE (6,520)	CON01970
		WRITE (6,570) TNOTOT	CON01980
C	160	IF (RADINS.GT.0.) GO TO 170	CON01990
		IERR=1	CON02000
		WRITE (6,520)	CON02010
		WRITE (6,580) RADINS	CON02020
		IF (ALST.GT.0.) GO TO 180	CON02030
170		IERR=1	CON02040
		WRITE (6,520)	CON02050
		WRITE (6,590) ALST	CON02060
		IF (SIDO.GT.0.) GO TO 190	CON02070
180		IERR=1	CON02080
		WRITE (6,520)	CON02090
		WRITE (6,600) SIDO	CON02100
		IF ((MDIAM.EQ.1).OR.(SIDI.GT.0.)) GO TO 200	CON02110
190		IERR=1	CON02120
		WRITE (6,520)	CON02130
		WRITE (6,610) SIDI	CON02140
		IF (SDDO.GT.1.) GO TO 210	CON02150
200		IERR=1	CON02160
		WRITE (6,520)	CON02170
		WRITE (6,620) SDDO	CON02180
		IF ((MPITCH.EQ.1).OR.(SDDI.GT.1.)) GO TO 220	CON02190
210		IERR=1	CON02200
		WRITE (6,520)	CON02210
		WRITE (6,630) SDDI	CON02220
		IF (IW.EQ.2) GO TO 230	CON02230
		IF (XW1.GT.0.) GO TO 240	CON02240
220		IERR=1	CON02250
		WRITE (6,520)	CON02260
		WRITE (6,640) XW1	CON02270
		GO TO 240	CON02280
230		X=XW2*2.	CON02290
		IF (X.GT.0.) GO TO 240	CON02300
		IERR=1	CON02310
		WRITE (6,520)	CON02320
		WRITE (6,650) XW2	CON02330
		IF ((WNCIR.GE.0.).AND.(WNCIR.LT.1.)) GO TO 250	CON02340
240		IERR=1	CON02350
		WRITE (6,520)	CON02360
		WRITE (6,660) WNCIR	CON02370
		IF ((STBI.GT.-32.).AND.(STBI.LT.212)) GO TO 260	CON02380
250			CON02390
			CON02400


```

260      IERR=1
      WRITE (6,520) STBI
      IF (STBI.LT.STSAT1) GO TO 270
      IERR=1
      WRITE (6,520) STBI,STSAT1
      IF ((IFLOW.GT.0).OR.(IFLOW.LE.3)) GO TO 280
      IERR=1
      WRITE (6,520) IFLOW
      WRITE (6,690) IFLOW
      IF (NEI.GE.0).AND.(NEI.LE.6)) GO TO 290
      IERR=1
      WRITE (6,520) NEI
      WRITE (6,700) NEI
      IF ((NEE.GE.0).AND.(NEE.LE.6)) GO TO 300
      IERR=1
      WRITE (6,520) NEE
      WRITE (6,710) NEE
      IF ((IBAF.GE.-1).AND.(IBAF.LE.ISEC)) GO TO 310
      IERR=1
      WRITE (6,520) IBAF,ISEC
      WRITE (6,720) IBAF,ISEC
      WRITE (6,730)
      IF (IERR.EQ.0) GO TO 320
310      STOP
      RETURN
320      FORMAT (15,3F10.5)
330      FORMAT (15,F10.5,F10.7)
340      FORMAT (4F10.5)
350      FORMAT (15,F10.5,F10.5)
360      FORMAT (15)
370      FORMAT (F12.5)
380      FORMAT (15,4F10.5)
390      FORMAT (2F10.5)
400      FORMAT (15,F10.5)
410      FORMAT (15)
420      FORMAT (2I5,3F10.5)
430      FORMAT (2I5,2F10.5)
440      FORMAT (1H0,31H**)
450      FORMAT (1H,22H INPUT VALUE OF OPT =,15,27H OUT OF RANGE, OPT SET
460      FORMAT (1H,21H INPUT VALUE OF MDIAM=,15,29H OUT OF RANGE, MDIAM SE
470      1 TO 2)
480      1 TO 1)
490      1 TO 1)
500      1 SET TO 1)
      FORMAT (1H,22H INPUT VALUE OF MPITCH=,15,30H OUT OF RANGE, MPITCH
      1)
      FORMAT (1H,21H INPUT VALUE OF IWALL=,15,29H OUT OF RANGE, IWALL SE
      1)

```



```

C-----
COMMON /GLOBCOM/ ALST, DELWP, DELWPC, EXITQA, GFLOW, SIDI, SIDD, PHP, RSPA,
1RADINS, REWI, REMO, SDDI, SDDO, SLDI, SLDI, SLDI, VOL1, VOL2, TNOTO,
2T, BNDRAD, ARATIO, ODI, ODI, SDDC, VLCMAX, PRCCLR
COMMON /INPT/ BC(6), BE(6), ENH(6), ENI(6), ENO(6), FOUL, PHI, SKW, STBI, S
1TSATL, TUBESW, WNCIR, IOPT, PST(15), SECWID, IBAF, ISEC, JBAF(16), JGAS, MD
2IAM, MPITCH, NEE, NETE(6), NETI(6), NRNE(6), NRNI(6), NOROWS, WTST(15)
3, WSI, PFILL
COMMON /INPTL/ IFLOW, IWALL
WRITE (6,60)
WRITE (6,70)
WRITE (6,80)
WRITE (6,90)
WRITE (6,100)
WRITE (6,90)
WRITE (6,80)
WRITE (6,110)
WRITE (6,120)
WRITE (6,130)
WRITE (6,140)
WRITE (6,150)
PRCCLR (6,170) TNOTOT
IF (IOPT.EQ.2) GO TO 10
IF (IOPT.EQ.2)
WRITE (6,160) NOROWS
WRITE (6,180) RADINS
WRITE (6,190) ALST
WRITE (6,200) MDIAM
IF (MDIAM.EQ.0) GO TO 20
WRITE (6,210) SIDD
WRITE (6,240)
GO TO 30
WRITE (6,220) SIDD, SIDI
WRITE (6,230) MPITCH
IF (MPITCH.EQ.0) GO TO 40
WRITE (6,260) SDDO
WRITE (6,280)
GO TO 50
WRITE (6,270) SDDO, SDDI
WRITE (6,250) IWALL
IF (IWALL.EQ.1) WRITE (6,300) XW1
IF (IWALL.EQ.2) WRITE (6,290) XW2
WRITE (6,310) TUBESW
WRITE (6,320) SKW
WRITE (6,330) FGUL
IF (JGAS.EQ.1) WRITE (6,360)
IF (JGAS.EQ.2) WRITE (6,370)
IF (JGAS.EQ.3) WRITE (6,380)
WRITE (6,340) WSI
CONO3370
CONO3380
CONO3390
CONO3400
CONO3410
CONO3420
CONO3430
CONO3440
CONO3450
CONO3460
CONO3470
CONO3480
CONO3490
CONO3500
CONO3510
CONO3520
CONO3530
CONO3540
CONO3550
CONO3560
CONO3570
CONO3580
CONO3590
CONO3600
CONO3610
CONO3620
CONO3630
CONO3640
CONO3650
CONO3660
CONO3670
CONO3680
CONO3690
CONO3700
CONO3710
CONO3720
CONO3730
CONO3740
CONO3750
CONO3760
CONO3770
CONO3780
CONO3790
CONO3800
CONO3810
CONO3820
CONO3830
CONO3840

```


C

```

AH OLES=0.
ROWS=FLOAT(NROWS)
SIDOF=SIDO/12.
SIDIF=SIDI/12.
DELSDF=(SIDOF-SIDIF)/(ROWS-1.)
IF (XW2.GT.0) ODOF=SIDOF+XW2/6.
IF (XW2.GT.0) OODIF=SIDIF+XW2/6.
IF (XW2.LE.0) ODOF=SIDOF+(2*SIDOF*XW1)
IF (XW2.LE.0) OODIF=SIDIF+(2*SIDIF*XW1)
ODOI=ODOCF*12.
ODOI=ODCIF*12.
DELODF=(ODOF-ODOIF)/(ROWS-1.)
DELOD=DELODF*12.
CSPQ=SDCI*ODOF
CSPI=SDCI*OODIF
DELCSP=(CSPQ-CSPI)/(ROWS-1.)
RSPF=.866*(CSPQ+CSPI)/2.
RSP=RSPF*12.
IF (RSPA.NE.0.) RSP=RSPA
RSPF=RSP/12.
BNDRAD=RADINS+RSPF*(ROWS-1.)
DO 40 I=1,NROWS
RAD=BNDRAD-RSPF*FLOAT(I-1)
CSP=CSPC-DELCSP*FLOAT(I-1)
TPAR(I)=1./(CSP*RSPF)
TNR=ARCPR*RAD/CSP
OD=ODOF-DELODF*FLOAT(I-1)
SID(I)=SIDOF-DELSDF*FLOAT(I-1)
ADFLW(I)=(CSP-OD)*TNR
ADTFLW(I)=ADFLW(I)*ALST
TNOS=TNOS+TNR
RADIUS(I)=RAD
AH OLES=AH OLES+PI*(ODOF**2)*TNR/4.
TBNPR(I)=TNR
TN0=TNOS*SECFLG
AH OLES=AH OLES*SECFLG
TN0TOT=TN0*100./(100.-PRCCLR)
GO TO 100

```

40

C

CONTINUE

COMPUTATION OF BUNDLE GEOMETRY IF IOPT = 2

SIDI=SIDO

SDDI=SDDO

ERR1=0.

50

C

C

C

C

C

```

CON06250
CON06260
CON06270
CON06280
CON06290
CON06300
CON06310
CON06320
CON06330
CON06340
CON06350
CON06360
CON06370
CON06380
CON06390
CON06400
CON06410
CON06420
CON06430
CON06440
CON06450
CON06460
CON06470
CON06480
CON06490
CON06500
CON06510
CON06520
CON06530
CON06540
CON06550
CON06560
CON06570
CON06580
CON06590
CON06600
CON06610
CON06620
CON06630
CON06640
CON06650
CON06660
CON06670
CON06680
CON06690
CON06700
CON06710
CON06720

```



```

TNO C=0.
TNOS=0.
TNR=0.
TDIFF=0.
SIDOF=SIDO/12.
SIDIF=SIDI/12.
IF (XW2.GT.0) ODOF=SIDOF+XW2/6.
IF (XW2.GT.0) ODI=SIDIF+XW2/6.
IF (XW2.LE.0) ODOF=SIDOF+(2*SIDOF*XW1)
IF (XW2.LE.0) ODI=SIDIF+(2*SIDIF*XW1)
ODQI=ODQF*12.
ODII=ODIF*12.
DELODF=0.
DELOD=CELODF
CSPO=SDDO*ODOF
CSPI=SDDI*ODIF
DELCSP=0.
RSPF=.866*(CSPO+CSPI)/2.
RSP=RSPF*12.
IF (RSPA.NE.0.) RSP=RSPA
RSPF=RSP/12.

        DETERMINE THE NUMBER OF TUBES IN THE CONDENSOR
        ALL SECTORS ARE IDENTICAL IN GEOMETRY
        BEGINNING WITH THE INNERMOST ROW, CALCULATE THE NUMBER
        OF RCWS IN THE CONDENSOR

I=1
NROWS=1
TNO=TNOTOT*(100.-PRCCLR)/100.
RADIUS(I)=RADINS

SID(I)=SIDOF
TNR=(ARCPR*RADIUS(I))/CSPO
TNOS=TNOS+TNR

        CHECK TO ENSURE THAT THE NUMBER OF TUBES IN THE SECTOR
        HAS NOT EXCEEDED THE NUMBER OF TUBES AVAILABLE

IF (TNOS.LT.(TNO/SECFLG)) GO TO 70
TNOS=TNOS-TNR
TNO C=TNOS*SECFLG
TDIFF=TNO-TNOC
TBNPR(I)=TDIFF/SECFLG
PFILL=TBNPR(I)/TNR
GO TO 80

TPAR(I)=1./(CSPO*RSPF)

```

C C C C C

C 60

C C C

C 70


```
C
C      AOTFLW(I)=(CSPO-ODOF)*TNR
C      AOTFLW(I)=AOTFLW(I)*ALST
C      TBNPR(I)=TNR
C      RADIUS(I+1)=RADIUS(I)+RSPF
C      I=I+1
C      NOROWS=NOROWS+1
C
C      CHECK TO SEE THAT THE NUMBER OF ROWS DOESNOT EXCEED 199
C      BECAUSE 200 IS THE LIMIT FOR THE DIMENSIONS OF NUMEROUS ARRAYS
C
C      IF (NOROWS.LT.100) GO TO 60
C      ERR1=TNO-(TNOS*SECFLG)
C      TNO=TNO+SECFLG
C      WRITE(6,640)
C      NOROWS=NOROWS-1
C      GO TO 90
C
C      IF LAST ROW IS INCOMPLETE, FIND ITS PERTINENT GEOMETRY
C
C      SDDL=(RADIUS(I)*ARCPR)/(ODOF*TBNPR(I))
C      CSPL=SDDL*ODOF
C      TPAR(I)=1./(CSPL*RSPF)
C      AOTFLW(I)=(CSPL-ODOF)*TBNPR(I)
C      AOTFLW(I)=AOTFLW(I)*ALST
C      TNOS=TNOS+TBNPR(I)
C      CONTINUE
C      AHOLE=PI*(ODOF**2)*TNO/4.
C      ROWS=FLCAT(NOROWS)
C      BNDRAD=RADINS+RSPF*(ROWS-1.)
C
C      THIS NEXT SUBROUTINE SWITCHES THE ARRAYS SO THE ARRAY VALUES
C      START WITH THE OUTSIDE ROW AND WORK TOWARDS THE CENTER.
C      THIS WILL MAKE THE GEOMETRY VALUES COMPATIBLE WITH THE REST
C      OF THE PROGRAM.
C
C      CALL SWITCH (RADIUS,NOROWS)
C      CALL SWITCH (SID,NOROWS)
C      CALL SWITCH (TBNPR,NOROWS)
C      CALL SWITCH (TPAR,NOROWS)
C      CALL SWITCH (AOTFLW,NOROWS)
C      CALL SWITCH (AOTFLW,NOROWS)
C
C      CONTINUE
C
C      AOTFLW(NOROWS+1)=ALST*(RADINS-SDI*ODII/24.)*PI/6.
C
C      IF (ISEC.EQ.1) SECANG(1)=PHI
```



```

CON07690
CON07700
CON07710
CON07720
CON07730
CON07740
CON07750
CON07760
CON07770
CON07780
CON07790
CON07800
CON07810
CON07820
CON07830
CON07840
CON07850
CON07860
CON07870
CON07880
CON07890
CON07900
CON07910
CON07920
CON07930
CON07940
CON07950
CON07960
CON07970
CON07980
CON07990
CON08000
CON08010
CON08020
CON08030
CON08040
CON08050
CON08060
CON08070
CON08080
CON08090
CON08100
CON08110
CON08120
CON08130
CON08140
CON08150
CON08160

IF (ISEC.EQ.1) GO TO 110
FCWID=SECWID*( SECFLG-1.)
DO 110 I=1,ISEC
SECA=(PHI-FCWID/2.)*SECWID*(1-1)
IF (SECA.GT.360.) SECA=SECA-360.
SECANG(I)=SECA

DUMMY=ALST*PI*SECWID*SECFLG/360.
BNDRA1=BNDRAD
VOL1=DUMMY*BNDRA1**2
VOLIC=VCL1

IF TNOTOT IS A DESIGN VARIABLE THEN THE INCREASE OF VOLUME AND
OUTSIDE BUNDLE RADIUS MUST BE A GRADUAL PROCESS AND NOT A STEP
INCREASE OR DECREASE

IF (IOPT.EQ.1) GO TO 120
RCORR=RSPF
RCORR=RCORR*((1.-PFILL))
BNDRAD=BNDRAD-RCORR
VOL1=DUMMY*BNDRAD**2
VOLIC=VCL1

VOL2=DUMMY*(BNDRAD**2-RADINS**2)

COMPUTE THE AREA RATIO WHICH IS THE AREA OF TUBE HOLES DIVIDED
BY THE TUBE SHEET AREA

TSAREA=PI*(BNDRAD**2)*SECFLG*SECWID/360.
ARATIO=AHOLES/TSAREA

CC$-----
CC ORC A4
CC TEMP OUTPUT SECTION
CC THIS SECTION FOR DEBUGGING AID
CC ZZZ ORC-A4
CC-----
OR CA4=0.
IF (ORCA4.NE.1.) GO TO 150
WR ITE (6,650)
WR ITE (6,660)
WR ITE (6,700) ODDF,ODIF,DELODF
WR ITE (6,670)
WR ITE (6,700) CSPO,CSPI,DEL CSP
WR ITE (6,680)
WR ITE (6,700) RSPF,RSP,INOS
```



```

WR ITE (6,690) AHOLE,TSAREA,ARATIO
WR ITE (6,700)
WR ITE (6,710)
DO 130 I=1,NROWS
WR ITE (6,740) AUTFLW(I),RADIUS(I),TBNPR(I),TPAR(I)
WR ITE (6,720)
DO 140 I=1,ISEC
WR ITE (6,730) SECANG(I)
130 CONTINUE
C$-----
C ORC 5
C
C LOCATE BAFFLES
C
C THIS SECTION CALCULATES BAFFLE ANGULAR LOCATION
C BASED ON THE INPUT VALUES OF IBAF AND JBAF.
C
C BAFA = BAFFLE LOCATIONS IN RADIAN MEASURED C-C WISE FROM VERT.
C FCWI = ONE-HALF OF THE BUNDLE ANGULAR WIDTH IN RADIAN
C FCWID = FCWI EXPRESSED IN DEGREES OF ARC
C IBAF = BAFFLE CONTROL VARIABLE
C IBAF = -1 IMPLIES BAFFLES BETWEEN EACH SECTOR
C IBAF = 0 IMPLIES NO BAFFLES (SEE NOTE)
C IBAF = N IMPLIES N BAFFLES
C IBAFT = NUMBER OF BAFFLES IN CONDENSER
C IBAFT = ISEC + 1 FOR IBAF = -1
C IBAFT = 2 FOR IBAF = 0
C IBAFT = N FOR IBAF = N
C ISEC = NUMBER OF SECTORS IN THE CONDENSER
C JBAF = SECTOR NUMBERS OF THOSE SECTORS WHOSE COUNTER-CLOCKWISE
C SIDE HAS A BAFFLE ATTACHED.
C PHI = BUNDLE ORIENTATION ANGLE IN DEGREES OF ARC CLOCK-WISE
C FROM VERTICLE.
C PHI I = PHI EXPRESSED IN RADIAN
C SECI = SECTOR ARC WIDTH IN RADIAN
C SECWID = SECTOR ARC WIDTH IN DEGREES OF ARC
C
C ZZZ ORC-5
C-----
C IBAFT=IBAF
C SECI=SECWID*PI/180.
C FCWID=SECWID*SECFLG/2.
C FCWI=FCWID*PI/180.
C PHI I=PHI*PI/180.
C IF (IBAF) 160,200,180
C IBAFT=ISEC+1
C DO 170 I=1,IBAF
C JBAF(I)=I
C DO 190 I=1,IBAF
160
170
180

```



```

190 IT=JBAF(I)
    BAFA(I)=PHI I-FCWI+SECI*FLOAT(IT-1)
    GO TO 210
200 BAFA(I)=PHI I-FCWI
    BAFA(2)=PHI I+FCWI
    IBAFT=2
    JBAF(1)=1
    JBAF(2)=ISEC+1
    CONTINUE
210 C-----
C$ C ORC A5
C TEMP OUTPUT SECTION
C ZZZ ORC-A5
C-----

```

```

    ORCA5=0.
    IF (ORCA5.NE.1) GO TO 240
    WRITE(6,750) IBAF, IBAFT
    WRITE(6,760) IBAF, IBAFT
    WRITE(6,770)
    DO 220 I=1, IBAFT
    WRITE(6,780) I, JBAF(I)
    WRITE(6,790)
    DO 230 I=1, IBAFT
    X=BAFA(I)*180./PI
    WRITE(6,800) X, BAFA(I)
    CONTINUE
220 C$-----
C ORC 6
C
C ZZZ ORC-6
C-----

```

LOCATE ACTIVE BAFFLE FOR EACH SECTOR

```

DO 430 IS=1, ISEC
SECA=SECANG(IS)*PI/180.
SEC=PI/2.-SECA
IF ((SECA.GT.0.) .AND. (SECA.LT.PI)) GO TO 270
DO 250 I=1, IBAFT
BAFAT=BAFA(I)
IF (BAFAT.GT.SECA) GO TO 260
CONTINUE
BAF=PI/2.-BAFAT
IF ((BAFAT.GT.0.) .AND. (SECA.LT.0.)) BAF=PI/2.
GO TO 300
DO 280 I=1, IBAFT
J=IBAFT+1-I
BAFAT=BAFA(J)
IF (BAFAT.LT.SECA) GO TO 290
CONTINUE
250 C$-----
C ORC 6
C
C ZZZ ORC-6
C-----

```

CON08650
 CON08660
 CON08670
 CON08680
 CON08690
 CON08700
 CON08710
 CON08720
 CON08730
 CON08740
 CON08750
 CON08760
 CON08770
 CON08780
 CON08790
 CON08800
 CON08810
 CON08820
 CON08830
 CON08840
 CON08850
 CON08860
 CON08870
 CON08880
 CON08890
 CON08900
 CON08910
 CON08920
 CON08930
 CON08940
 CON08950
 CON08960
 CON08970
 CON08980
 CON08990
 CON09000
 CON09010
 CON09020
 CON09030
 CON09040
 CON09050
 CON09060
 CON09070
 CON09080
 CON09090
 CON09100
 CON09110
 CON09120


```

C      ZT = VERT DIST FROM TARGET TUBE TO BAFFLE INTERSECTION - FT
C      ZZZ ORC-7
C-----
ORCA7=0.
IF (ORCA7.EQ.1.) WRITE (6,860)
DO 420 IR=1,NROWS
IF (IS.EQ.ISEC.AND.IR.GE.20)ORCA7 = 1.
ITR=IR
AN=0.
OD=ODOF-DELODF*FLOAT(IR-1)
R1=RADIUS(IR)
YC=R1*SIN(SEC)
XC=R1*CCS(SEC)
XC S=XC*XC
TEST=ABS(ABS(BAF)-PI/2.)
X=OD*12.
IF (ORCA7.EQ.1.) WRITE (6,870) IR,IS
IF (ORCA7.EQ.1.) WRITE (6,880) X,R1,YC,XC,TEST
IF (TEST.GT.1E-2) GO TO 320
ZT=1E10
GO TO 330
320  ZT=XC*TAN(BAF)
330  XC=ABS(XC)
IF (YC.GT.0.) GO TO 370
IF (ORCA7.EQ.1.) WRITE (6,890) ZT
D1=YC
DO 340 I1=IR,NROWS
R1=RADIUS(I1)-RSPF/2.
IF (R1.LE.XC) GO TO 350
D2=-1.*SQRT(R1*R1-XC*XC)
IF (ORCA7.EQ.1.) WRITE (6,900) I1,R1,D2
IF (D2.GT.ZT) GO TO 360
IF (ORCA7.EQ.1.) WRITE (6,910)
AN=AN+(D2-D1)/(RSPF*2.)
IF (ORCA7.EQ.1.) WRITE (6,920) D1,D2,AN
D1=D2
CONTINUE
D2=-D2
IF (ORCA7.EQ.1.) WRITE (6,940)
IF (D2.GT.ZT) GO TO 410
ITR=NORCWS
D1=D2
IF (ORCA7.EQ.1.) WRITE (6,950)
GO TO 380
AN=AN-D1/(RSPF*2.)
D1=0.
ITR=I1
340
350

```

```

CON09610
CON09620
CON09630
CON09640
CON09650
CON09660
CON09670
CON09680
CON09690
CON09700
CON09710
CON09720
CON09730
CON09740
CON09750
CON09760
CON09770
CON09780
CON09790
CON09800
CON09810
CON09820
CON09830
CON09840
CON09850
CON09860
CON09870
CON09880
CON09890
CON09900
CON09910
CON09920
CON09930
CON09940
CON09950
CON09960
CON09970
CON09980
CON09990
CON10000
CON10010
CON10020
CON10030
CON10040
CON10050
CON10060
CON10070
CON10080

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```

360 IF (ORCA7.EQ.1.) WRITE (6,930) AN, ITR
    GO TO 380
    AN=AN+(ZT-D1)/(RSPF*2.)
    IF (ORCA7.EQ.1.) WRITE (6,960)
    GO TO 410
370 D1=YC
    IF (ORCA7.EQ.1.) WRITE (6,970) D1
380 D0 390 I2=1, ITR
    I1= ITR- I2+1
    R1=RADIUS(I1)+RSPF/2.
    D2=SQR( R1*R1-XCS)
    IF (ORCA7.EQ.1.) WRITE (6,980) IR, I1, R1
    IF (ORCA7.EQ.1.) WRITE (6,990) D1, D2, ZT
    IF (D2.GT.ZT) GO TO 400
    AN=AN+(D2-D1)/(RSPF*2.)
    IF (ORCA7.EQ.1.) WRITE (6,1000) AN
390 D1=D2
    GO TO 410
400 AN=AN+(ZT-D1)/(RSPF*2.)
    IF (ORCA7.EQ.1.) WRITE (6,1010)
410 NF=AN+1.0
    RNF=NF
    ANT(IR, IS)=RNF
    IF (ORCA7.EQ.1.) WRITE (6,1020) IS, IR, AN
    CONTINUE
420 CONTINUE
430 ORCA7=0.
    IF (ORCA7.NE.1.) GO TO 450
    DO 440 I=1, NROWS
    WRITE (6,850) I, (ANT(I,J), J=1, ISEC)
440 CONTINUE
450 C$-----
    C ORC 9
                                     PUMPING POWER CALCULATION
    ALST = TUBE LENGTH FT
    A1 = COEF IN FF CALCULATION
    B1 = EXP IN FF CALCULATION
    CBI = COOLANT SALT CONCENTRATION IN WT PRCNT
    CMDQT = COOLANT MASS FLOW/TUBE IN EACH ROW LBM/SEC
    CMTOT = TOTAL COOLANT MASS FLOW THROUGH CONDENSER LBM/SEC
    DELWP = HEADER PRESSURE DIFFERENCE PSI
    FF = TUBE FRICTION FACTORS
    HEAD = FT*2/SEC**
    SID = STORED INNER TUBE DIAMETER
    TID = STORED INNER VALUE FOR TUBE INNER DIAMETER
    SLD = STORED (LENGTH/TUBE INNER DIAMETER) RATIO
    SLD0 = SLD RATIO OF TUBES IN OUTER ROW

```

```

CON10090
CON10100
CON10110
CON10120
CON10130
CON10140
CON10150
CON10160
CON10170
CON10180
CON10190
CON10200
CON10210
CON10220
CON10230
CON10240
CON10250
CON10260
CON10270
CON10280
CON10290
CON10300
CON10310
CON10320
CON10330
CON10340
CON10350
CON10360
CON10370
CON10380
CON10390
CON10400
CON10410
CON10420
CON10430
CON10440
CON10450
CON10460
CON10470
CON10480
CON10490
CON10500
CON10510
CON10520
CON10530
CON10540
CON10550
CON10560

```



```

C          SLDI = SLD RATIO OF TUBES IN INNER ROW
C          SLD = THE AVERAGE OF SLD0 AND SLDI
C          HLOSS = HEAD LOSS
C          REW = REYNOLDS # OF COOLANT
C          REW0 = REYNOLDS # OF TUBES IN THE OUTER ROW
C          REWI = REYNOLDS # OF TUBES IN INNER ROW
C          PHP = PUMPING POWER IN HP
C          PHPCCN = PUMPING POWER OF THE CONDENSER BUNDLE ONLY
C          DELWPC = PRESSURE LOSS THRU TUBES IN PSI
C          GFLOW = OVERALL COOLANT FLOW IN LBM/HR
C          RHOW = DENSITY OF COOLANT
C          XMUW = VISCOSITY OF COOLANT
C          VELBI = COOLING WATER VELOCITY IN FT/SEC
C
C          ZZZ ORC-9
C-----
C          ORCA9=0.
C          RHOW=RDEFN(CBI,STBI)
C          XMUW=BMUFN(CBI,STBI)/3600.
C          XNUW=XMUW/RHOW
C          CMTOT=0.
C          IF ((GFLOW.GT.0.) .OR. (VELBI.GT.0.)) GO TO 530
C          HEAD=2.*DELP*144.*SG/RHOW
C          DO 520 IR=1,NROWS
C          IF (ORCA9.EQ.1.) WRITE (6,1030) IR
C          ITEM=IR-1
C          IF (ITEM.EQ.0) ITEM=1
C          VG=VW(ITEM)
C          IF (ORCA9.EQ.1.) WRITE (6,1040) IR,ITEM,VG
C          IENHAN=0
C          IF (NEI.EQ.0) GO TO 480
C          DO 460 IF=1,NEI
C          NRNI1=NRNI(IT)
C          NRNI2=NRNI1+NEI(IT)-1
C          IF (IR.GE.NRNI1.AND.IR.LE.NRNI2) GO TO 470
C          CONTINUE
C          GO TO 480
C          IENHAN=1
C          AL=BC(IT)
C          BI=BE(IT)
C          TID=SID(IR)
C          SLD=ALST/TID
C          IF (IR.EQ.1) SLD0=SLD
C          IF (IR.EQ.NROWS) SLDI=SLD
C          IF (ORCA9.EQ.1.) WRITE (6,1050) TID,SLD,IR
C          REW=VG*TID/XNUW
C          IF (ORCA9.EQ.1.) WRITE (6,1060) REW,TID,XNUW
C          IF (IENHAN.EQ.1) GO TO 510

```

```

CON10570
CON10580
CON10590
CON10600
CON10610
CON10620
CON10630
CON10640
CON10650
CON10660
CON10670
CON10680
CON10690
CON10700
CON10710
CON10720
CON10730
CON10740
CON10750
CON10760
CON10770
CON10780
CON10790
CON10800
CON10810
CON10820
CON10830
CON10840
CON10850
CON10860
CON10870
CON10880
CON10890
CON10900
CON10910
CON10920
CON10930
CON10940
CON10950
CON10960
CON10970
CON10980
CON10990
CON11000
CON11010
CON11020
CON11030
CON11040

```



```

500 IF (REW,GE.51904.4) GO TO 500
    AL=.3164
    BL=-.25
    GO TO 510
510 AL=.184
    BL=-.2
    CONTINUE
    IF (ORCA9.EQ.1.) WRITE (6,1070) AL,B1,REW
    FF=AL*REW**(.B1)
    HLOSS=(1.1+FF*SLD)
    IF (ORCA9.EQ.1.) WRITE (6,1080) FF,SLD,HLOSS,HEAD
    VG1=SQRT(HEAD/HLOSS)
    TEST=ABS(VG-VG1)/VG
    IF (ORCA9.EQ.1.) WRITE (6,1090) TEST,VG,VG1
    VG=VG1
    IF (TEST.GT.0.01) GO TO 490
    IF (IR.EQ.1) REW=REW
    IF (IR.EQ.NOROWS) REW=REW
    CMDOT(IR)=PI*TIID**2*VG*RHOW/4.
    VW(IR)=VG
520 CMTOT=SECFLG*CMDOT(IR)*TBNPR(IR)+CMTOT
    PHPCON=PHP
    DELWPC=DELWP
    GO TO 600
530 A=0.
    DO 540 IR=1,NROWS
    TID=STC(IR)
    IF (IR.EQ.1) SLD=ALST/TID
    IF (IR.EQ.NOROWS) SLD=ALST/TID
    A=A+TID*TIID*PI*TBNPR(IR)/4.
    SLD=(SLD+SLD1)/2.
    TID=ALST/SLD
    IF (GFLCW.LE.0.) GO TO 550
    VX=GFLCW/((RHOW*A*SECFLG)*3600.)
    CMTOT=GFLOW/3600.
    GO TO 560
550 CMTOT=VELBI*RHOW*A*SECFLG
    VX=VELBI
    REW=VX*TIID/XNJW
560 IF (REW.GT.51904.4) GO TO 570
    FF=.3164*REW**(-.25)
    GO TO 580
570 FF=.184*REW**(-.2)
580 DELWPC=(1.1+FF*SLD)*VX**2*RHOW/(2.*SG*144.)
    DO 590 IR=1,NROWS
    CMDOT(IR)=PI*SID(IR)**2*VX*RHOW/4.
    VW(IR)=VX

```

```

CON11050
CON11060
CON11070
CON11080
CON11090
CON11100
CON11110
CON11120
CON11130
CON11140
CON11150
CON11160
CON11170
CON11180
CON11190
CON11200
CON11210
CON11220
CON11230
CON11240
CON11250
CON11260
CON11270
CON11280
CON11290
CON11300
CON11310
CON11320
CON11330
CON11340
CON11350
CON11360
CON11370
CON11380
CON11390
CON11400
CON11410
CON11420
CON11430
CON11440
CON11450
CON11460
CON11470
CON11480
CON11490
CON11500
CON11510
CON11520

```



```

REWO=VX*SID(1)/XNUW
REW1=VX*SID(NROWS)/XNUW
HEAD=2.*DELWPC*144.*SG/RHOW
PHP=CMCT*DELWPC*144./(RHOW*550.)
PHPCON=PHP
DELWP=DELWPC
CONTINUE
HFG=HFGFN(STSAT1)
STB2ES=(WSI*HFG)/(CPB*CMUT*3600.)*STBI

```

600

C\$-----

C UR C A9

TEMP OUTPUT SECTION

C ZZZ ORC-A9

C-----

```

ORCA9=0.
IF (ORCA9.NE.1.) GO TO 630
WRITE (6,1100)
WRITE (6,1110) RHOW
WRITE (6,1120) XMUW
WRITE (6,1130) XNUW
WRITE (6,1140) HEAD
WRITE (6,1150) REW,FF
WRITE (6,1160) DELWP
WRITE (6,1170) SLDI,SLDO
WRITE (6,1180) REWI,REWO
WRITE (6,1190) CMTO
WRITE (6,1200) PHP
WRITE (6,1210)
DO 610 I=1,NROWS
WRITE (6,1220) I,VW(I)
DO 620 I=1,NROWS
WRITE (6,1220) I,CMDOT(I)
CONTINUE

```

610

620

630

C\$-----

C UR C 10

CALL TO SECALC

C ZZZ ORC-10

C-----

```

CALL SECALC (ARCPR,DELODF,ODIF,ODOF)
RETURN
FORMAT (1H0,46HTHE NUMBER OF ROWS EXCEEDS THE MAXIMUM ALLOWED)
FORMAT (1H1,24HOUTPUT SECTION FOR ORC-4)
FORMAT (1H0,10HDOOF ,10HCELODF )
FORMAT (1H0,10HCSP ,10HCELCSP )
FORMAT (1H0,10HRSPP ,10HTNOS )
FORMAT (1H0,10HHAHOLES ,10HARATIO )
FORMAT (1H ,3F10.5)
FORMAT (1H0,10HAOTFLW ,10HBNPR ,10HTPAR )

```

640

650

660

670

680

690

700

710

CON11530
CON11540
CON11550
CON11560
CON11570
CON11580
CON11590
CON11600
CON11610
CON11620
CON11630
CON11640
CON11650
CON11660
CON11670
CON11680
CON11690
CON11700
CON11710
CON11720
CON11730
CON11740
CON11750
CON11760
CON11770
CON11780
CON11790
CON11800
CON11810
CON11820
CON11830
CON11840
CON11850
CON11860
CON11870
CON11880
CON11890
CON11900
CON11910
CON11920
CON11930
CON11940
CON11950
CON11960
CON11970
CON11980
CON11990
CON12000


```

720 1) FORMAT (1H0,10HSECCANG , )
730 FORMAT (1H ,F10.6)
740 FORMAT (1H ,4F10.5)
750 FORMAT (1H1,24HOUTPUT FOR SECTION ORC-5)
760 FORMAT (1H0,7HIBAF = ,I4,5X,8HIBAF = ,I4)
770 FORMAT (1H0,22HFOR I = : JBAF(I) = )
780 FORMAT (1H ,3X, I4, 8X, I4)
790 FORMAT (1H0,36HBAFFLE ANGLES IN DEGREES AND RADIAN)
800 FORMAT (1H ,3X, F6.1, 3X, F6.4)
810 FORMAT (1H1,24HOUTPUT FOR SECTION ORC-6)
820 FORMAT (1H0,50HANGLES ARE MEASURED CNTR-CLOCK WISE FROM HORIZONTAL)
830 FORMAT (1H0,16HCURRENT SECTOR = ,I3,16H ANGLE IN DEG = ,F6.1, 21H AN
IN RADIANS = ,F6.4)
840 FORMAT (1H0,32HANGLE OF ACTIVE BAFFLE IN DEG. = ,F6.1, 14H IN RADIA
INS = ,F6.4)
850 FORMAT (1H ,I3, 2X, 6(F5.2, 2X))
860 FORMAT (1H0,28H*** OUTPUT FOR ORC-7 *** )
870 FORMAT (1H0,5HROW = ,I4,11H SECTOR = ,I4)
880 FORMAT (1H ,4HOD = ,F6.4, 10H RADIUS = ,F6.2, 6H YC = ,F6.2, 6H XC = ,
F6.2, 8H TEST = ,E10.3)
890 FORMAT (1H0,36HTARGET TUBE IN LOWER QUADRANT, ZT = ,F20.4)
900 FORMAT (1H ,15HLOWR QUAD - ROW = ,I3, 5H RI = ,F7.3, 5H D2 = ,F7.3)
910 FORMAT (1H ,18HBAFFLE WAS NOT HIT)
920 FORMAT (1H ,15HSTILL LOWR - D1 = ,F7.3, 5H D2 = ,F7.3, 5H AN = ,F6.2)
930 FORMAT (1H ,34HPASSING INTO UPPER QUADRANT - AN = ,F6.2, 15H CURREN
T ROW = ,I4)
940 FORMAT (1H ,30HMOVING THRU VOID TO UPPER QUAD)
950 FORMAT (1H ,22HBAFFLE NOT HIT IN VOID)
960 FORMAT (1H ,28HBAFFLE HIT IN LOWER QUADRANT)
970 FORMAT (1H ,35HTARGET TUBE IS IN UPPER QUAD - D1 = ,F8.4)
980 FORMAT (1H ,37HCALCULATING IN UPPER QUAD - TRGET RW = ,I4, 12H CALC
IROW = ,I4, 11H CORR RAD = ,F8.4)
990 FORMAT (1H ,4HD1 = ,F6.3, 6H D2 = ,F6.3, 6H ZT = ,E10.3)
1000 FORMAT (1H ,34HUPPER QUAD - BAFFLE NOT HIT - AN = ,F6.2)
1010 FORMAT (1H ,23HUPPER QUAD - BAFFLE HIT)
1020 FORMAT (1H0,26H** FINAL FOR TARGET ** IS = ,I4, 4H IR = ,I4, 5H ANT = ,F8.
12)
1030 FORMAT (1H ,16HAT DO LOOP - IR = ,I4)
1040 FORMAT (1H ,9HTEMP IR = ,I4, 7H ITEM = ,I4, 5H VG = ,E10.3)
1050 FORMAT (1H ,16HABOVE 380 - TID = ,E10.3, 6H SLD = ,E10.3, 5H IR = ,I4)
1060 FORMAT (1H ,16HBELOW 380 - REW = ,E10.3, 6H TID = ,E10.3, 7H XNUM = ,E10
1.3)
1070 FORMAT (1H ,12HAT 400 - A1 = ,E10.3, 5H B1 = ,E10.3, 6H REW = ,E10.3)
1080 FORMAT (1H ,3HFF = ,E10.3, 5H SLD = ,E10.3, 7H HLCSS = ,E10.3, 6H HEAD = ,E10
1.3)
1090 FORMAT (1H ,5HTEST = ,E10.3, 5H VG = ,E10.3, 6H VG1 = ,E10.3)
1100 FORMAT (1H1,16HOUTPUT FOR ORC-9)
CON12010
CON12020
CON12030
CON12040
CON12050
CON12060
CON12070
CON12080
CON12090
CON12100
CON12110
CON12120
CON12130
CON12140
CON12150
CON12160
CON12170
CON12180
CON12190
CON12200
CON12210
CON12220
CON12230
CON12240
CON12250
CON12260
CON12270
CON12280
CON12290
CON12300
CON12310
CON12320
CON12330
CON12340
CON12350
CON12360
CON12370
CON12380
CON12390
CON12400
CON12410
CON12420
CON12430
CON12440
CON12450
CON12460
CON12470
CON12480

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```

4),WSP(15)      AMWNC,CBI,CPB,PI,SG,IFIRST      CON12970
COMMON /CONST/  AMWNC,CBI,CPB,PI,SG,IFIRST      CON12980
COMMON /COOL1/  VELC(100),PMIXC(100),PSATC(100),WSC(100)  CON12990
1),ALMTDC(100),SHIC(100),SHNFC(100),RCC(100),ROUTC(100),VNREC(100),  CON13000
2VP,SHHC(100),HEFFC(100),UNC(100),WCNDC(100),GFLOWC(100),QUAC(100),C  CON13010
3UMDPC(100)      CON13020
COMMON /COOL/   IVNOC,HNOC,INOC,TSATEX,PMXEXT,AMLSEX,WSEXIT,VELEXIT,T  CON13030
1B2C,ENHIC,ENHOC,ENHFC,SHWINC,WBC      CON13040
DIMENSION C(15)      CON13050
-----      CON13060
C$  SEC 2      CON13070
C      INITIALIZE : OLD ORC 8,9      CON13080
C      PART - 1      CON13090
C      CON13100
ALST = TUBE LENGTH FT      CON13110
AMOLNC = TOTAL MOLES OF N/C GAS ENTERING CONDENSER      CON13120
AMOLSS = MOLES OF STM ENTERING COND      CON13130
AMOLST = SUM OF MOLES OF STM AND N/C ENTERING COND      CON13140
AMWNC = MOLECULAR WEIGHT OF N/C      CON13150
ANRAT = TOTAL NO OF TUBE IN 360 DEG COND      CON13160
ARAT = RATIO OF OUTSIDE ROW MIN FLOW AREA TO SECTOR FACE AREA      CON13170
ARCPR = CIRCUMFERENTIAL ARC LENGTH PER UNIT RADIUS      CON13180
AOTFLW = AREA OPEN TO FLOW IN A ROW OF TUBES IN A SECTOR FT**2      CON13190
BNDRAD = TUBE BUNDLE OD FT      CON13200
DELODF = CHANGE IN TUBE OD PER ROW FT      CON13210
DELOP12 = ENTRANCE STM PRESS LOSS PSI      CON13220
DELOP23 = STM PRESS LOSS AS STM IS ACCEL INTO FIRST ROW PSI      CON13230
G2 =      CON13240
HFG = STEAM LATENT HEAT OF VAPORIZATION BTU/LBM      CON13250
NOROWS = NO. OF ROWS IN THE CONDENSER      CON13260
ODOF = TUBE OD, INSIDE ROW FT      CON13270
ODIF = TUBE OD, INSIDE ROW FT      CON13280
ODOI = TUBE OD, INSIDE ROW IN      CON13290
ODII = TUBE OD, INSIDE ROW IN      CON13300
PMIX1 = PRESS OF STM - N/C MIXTURE ENTERING COND PSIA      CON13310
PSAT1 = ENTERING STEAM SAT PRESSURE PSI      CON13320
PTLIM = LOWEST SATURATION PRESSURE THE STEAM CAN GO BEFORE THE      CON13330
CORRESPONDING SATURATION TEMPERATURE FALLS BELOW INLET      CON13340
COOLANT TEMPERATURE. THE RESULT IS 0 HEAT TRANSFER      CON13350
RADINS = BUNDLE VOID RADIUS FT      CON13360
ROWS = NO OF ROWS IN THE CONDENSER      CON13370
SECFAR = FACE AREA FOR ONE SECTOR      CON13380
SECFLG = NO. OF SECTORS IN THE CONDENSER MODEL * FLOW RATE)      CON13390
SMTBI = ROW BY ROW SUM OF (COOLANT FLOW      CON13400
SMTWB = ROW BY ROW SUM OF (COOLANT FLOW      CON13410
SMTB2 = ROW BY ROW SUM OF (COOLANT OUTLET TEMP * FLOW RATE)      CON13420
SUMQ = ROW BY ROW SUM OF HEAT DUTY ON TUBES      CON13430
SG = GRAVITY FACTOR 32.2      CON13440

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C STSAT1 = ENTERING STM SAT TEMP F FT**3/LBM CON13450
C SVMIX1 = ENTERING MIXTURE SPECIFIC VOL FT**3/LBM CON13460
C SVNCI = ENTERING N/C GAS SATURATION TEMP R CON13470
C TSAT1 = ENTERING STEAM SATURATION TEMP CON13480
C WGAS = N/C FLOW TO EACH SECTOR LBM/HR CON13490
C WNCIR = N/C GAS FLOW TO CONDENSER LBM/HR CON13500
C WNCIR = RATIO OF MASS FLOW OF N/C TO MASS FLOW OF STEAM CON13510
C WS = STEAM FLOW TO EACH SECTOR LBM/HR CON13520
C WSI = TOTAL STM FLOW TO CONDENSER LBM/HR CON13530
C VEL1 = MIXTURE VELOCITY PRIOR TO ENTERING FIRST ROW FT/SEC CON13540
C VEL2 = MIXTURE VELOCITY IN FIRST ROW OF TUBES FT/SEC CON13550
C IVNOC = THE NUMBER OF VERTICAL TUBES IN THE COOLER CON13560
C IHNOC = THE NUMBER OF HORIZONTAL TUBES IN THE COOLER CON13570
C HTCLR = THE HEIGHT OF THE COOLER CON13580
C VGI = ENTERING STM SPECIFIC VOL FT**3/LBM CON13590
C CON13600
C ZZZ SEC-2 CON13610
C----- CON13620
C ROWS=FLOAT(NROWS) CON13630
C TNOC=0. CON13640
C IVNOC=1. CON13650
C IF (PRCLR.EQ.0.0) GO TO 10 CON13660
C CON13670
C COMPUTE THE NUMBER OF VERTICAL AND HORIZONTAL ROWS IN THE COOLER CON13680
C IF THERE IS ONE CON13690
C CON13700
C SDC=SDDI CON13710
C TNOC=(PRCLR/(100.-PRCLR))*TNO CON13720
C HTCLR=BNDRAD-RADINS CON13730
C VNOC=(HTCLR/(SDC*ODIF*0.886))+1. CON13740
C IHNOC=TNOC/VNOC CON13750
C HNOC=FLOAT(IHNOC) CON13760
C VNOC=TNOC/HNOC CON13770
C IVNOC=VNOC+.5 CON13780
C VNOC=FLOAT(IVNOC) CON13790
C CON13800
C KSTOP=NCROWS+1 CON13810
C SECFLG=FLOAT(1 SEC) CON13820
C WNCI=WSI*WNCIR CON13830
C ILOOP=0 CON13840
C WSAVE=(WSI+WNCI)/SECFLG CON13850
C CON13860
C INITIALIZE TEST VARIABLES TO 0 CON13870
C CON13880
C CON13890
C CONTINUE CON13900
C STMSUM=0. CON13910
C WTSUM=0. CON13920

```



```

STBIR=STBI+459.69
PTLIM=PSATFN(STBIR)
TSAT1=STSAT1+459.69
PSAT1=PSATFN(TSAT1)
AMOLNC=WNCI/AMWNC
HF G=HF GFN(STSAT1)
AMOLSS=WSI/18.015
AMOLST=AMOLSS+AMOLNC
PMIX1=PSAT1*(AMOLST/AMOLSS)
VGI=VG FN(TSAT1,PMIX1)
SVNCL=10.729*TSAT1/(AMWNC*PMIX1)
SVMIX1=1.07*(AMOLSS/(AMOLST*VGI)+AMOLNC/(SVNCL*AMOLST))
G2=288.0*SG*SVMIX1
SM TBI=0.0
SM WB=0.0
WB=0.
SM TB2=0.0
SUMQ=0.0
CUMDRY=0.
SUBFLD=0.
NBAD=0
WCNDAY=0.
SUBAVE=0.

```

COMPUTE THE PRESSURE LOSSES AT THE ENTRANCE OF THE SECTOR

```

SECFAR=ARCPR*(BNDRAD+ODOF/2.)*ALST
ARAT=AOTFLW(1)/SECFAR
VEL1=(WSI+WNCI)*SVMIX1/(3600.0*SECFAR*SECFLG)
VEL2=VEL1/ARAT
IF (ARAT-0.715) 30,30,40
DELP12=0.4*(1.25-ARAT)*VEL2**2/G2
GO TO 50
DELP12=0.75*(1.0-ARAT)*VEL2**2/G2
DELP23=(VEL2**2-VEL1**2)/G2

```

SEC A2

TEMP OUTPUT SECTION

ZZZ SEC-A2

```

-----
SECA2=0.
IF (SECA2.EQ.0.) GO TO 70
DO 60 I=1,I SEC
WRITE (6,610)
WRITE (6,620) WSI,WNCI

```

CON13530
CON13940
CON13950
CON13560
CON13970
CON13980
CON13590
CON14000
CON14010
CON14020
CON14030
CON14040
CON14050
CON14060
CON14070
CON14080
CON14090
CON14100
CON14110
CON14120
CON14130
CON14140
CON14150
CON14160
CON14170
CON14180
CON14190
CON14200
CON14210
CON14220
CON14230
CON14240
CON14250
CON14260
CON14270
CON14280
CON14290
CON14300
CON14310
CON14320
CON14330
CON14340
CON14350
CON14360
CON14370
CON14380
CON14390
CON14400


```

C CON15850
C CON15860
C CON15870
C CON15880
C CON15890
C CON15900
C CON15910
C CON15920
C CON15930
C CON15940
C CON15950
C CON15960
C CON15970
C CON15980
C CON15990
C CON16000
C CON16010
C CON16020
C CON16030
C CON16040
C CON16050
C CON16060
C CON16070
C CON16080
C CON16090
C CON16100
C CON16110
C CON16120
C CON16130
C CON16140
C CON16150
C CON16160
C CON16170
C CON16180
C CON16190
C CON16200
C CON16210
C CON16220
C CON16230
C CON16240
C CON16250
C CON16260
C CON16270
C CON16280
C CON16290
C CON16300
C CON16310
C CON16320

OD = TUBE OD FT
ODOF = TUBE CD, OUTSIDE ROW FT
XDW = TUBE WALL THICKNESS FT
AXI = TUBE INNER CROSS-SECTIONAL AREA FT**2
AO = TUBE INNER SURFACE AREA FT**2
AW = TUBE OUTER SURFACE AREA FT**2
SHMINV = LOG MEAN TUBE SURFACE AREA FT**2
PTLIM = INVERSE OF WALL HEAT X-FER COEF HR-FT**2-F/BTU THE
C CORRESPONDING SATURATION PRESSURE THE STEAM CAN GO BEFORE
C COOLANT TEMPERATURE. THE RESULT IS 0 HEAT TRANSFER
SKW = TUBE THERMAL CONDUCTIVITY BTU-FT/HR-FT**2-F
TID = TUBE INNER DIAMETER FT
WST = STEAM FLOW TO A ROW IN A SECTOR LBM/HR

C ZZZ SEC-4
C-----
DO 380 IL=1,NROWS
I=IL
L=J
IJX=0
LQ=0
WB=CMDOGT(I)*3600.
AXO=AO*FLW(I)
OD=ODOF-DELODF*FLOAT(IL-1)
TID=SID(IL)/2.
XW=(OD-TID)/2.
AI=ALST*PI*TID
AO=ALST*PI*OD
AXI=PI*(TID/2. )**2
SDW=(OD-TID)/A*LOG(OD/TID)
AW=ALST*PI*SDW
SHMINV=(AO*XW)/(SKW*AW)
WST=WS(IL,J)
C**** IF (WST.LE.0) WRITE (6,600) J,IL
C$-----
C SEC A4
C TEMP OUTPUT SECTION
C ZZZ SEC-A4
C-----
SECA4=0.
IF (SECA4.NE.1) GO TO 110
WRITE (6,810) IL
WRITE (6,820) AI,AO
WRITE (6,830) TID,AXI

```


WRITE (6,840) SDW,AW
 WRITE (6,850) SHWINV
 CONTINUE

110

SEC 5

CALL TO HETTRN AND PRSDRP

THIS SECTION INSIDE BOTH MAIN ROW AND SECTOR LOUPS

STSAT = STEAM SATURATION TEMP F
 ANST = TARGET TUBE LOCATION IN VERTICAL ROW
 WS = STEAM FLOW TO A ROW OF A SECTOR LBM/HR
 WNC = N/C FLOW TO A ROW OF A SECTOR LBM/HR
 UN = OVERALL HEAT TRANSFER COEF FOR CNE ROW OF TUBES IN A SECT.
 ALMTD = LMTD FOR A ROW IN A SECTOR F
 HOMCI = STSAT - COOLANT INLET TEMP
 HOMCO = STSAT - COOLANT OUTLET TEMP
 STFO = AVE TEMP OF OUTER TUBE FILM
 SHI = INTERNAL TUBE FILM HEAT TRANSFER COEF.
 SHN = EXTERNAL TUBE FILM HEAT TRANSFER COEF
 RCUT = RATIO OF N/C GAS TO STM AT ONE ROW OF TUBES
 AXO = EXTERNAL HEAT TRANSFER COEF
 VNRE = AREA OPEN TO FLOW IN A ROW OF TUBES FT*2
 DUMMY = REYNOLDS NUMBER
 VPSHH = OUTLET COOLANT TEMP F
 HEFF = N/C GAS FILM HEAT TRANSFER COEF
 L = SECTOR NUMBER
 HFG = STEAM LATENT HEAT OF VAPORIZATION BTU/LBM
 WCND = COND ENSATE FROM ONE ROW OF TUBES IN A SECTOR LBM/HR
 AO = OUTER SURFACE AREA OF A TUBE FT*2
 PT LIM = LOWEST SATURATION PRESSURE THE STEAM CAN GO BEFORE THE
 CORRESPONDING SATURATION TEMPERATURE FALLS BELOW INLET
 COOLANT TEMPERATURE. THE RESULT IS 0 HEAT TRANSFER
 TBNPR = NUMBER OF TUBES IN A ROW
 TAL = SATURATED STEAM TEMP
 VMIX = SPECIFIC VOLUME OF STM-N/C MIXTURE FT*3/LBM
 SDO = TUBE OD FT
 VPSH = FRICTION FACTOR USED IN PRESSURE DROP CALC
 DELPTP = PRESS DROP ACROSS A ROW OF TUBES PSI
 ENHFP = PRESSURE DROP FRICTION FACTOR ENHANCEMENT
 ENHF = PRESS DROP FRICTION FACTOR ENHANCEMENT FACTOR, STM SIDE.

ZZZ SEC-5

CHECK TO ENSURE THAT THE SATURATION PRESSURE IN AN EARLIER ROW

CON16330
 CON16340
 CON16350
 CON16360
 CON16370
 CON16380
 CON16390
 CON16400
 CON16410
 CON16420
 CON16430
 CON16440
 CON16450
 CON16460
 CON16470
 CON16480
 CON16490
 CON16500
 CON16510
 CON16520
 CON16530
 CON16540
 CON16550
 CON16560
 CON16570
 CON16580
 CON16590
 CON16600
 CON16610
 CON16620
 CON16630
 CON16640
 CON16650
 CON16660
 CON16670
 CON16680
 CON16690
 CON16700
 CON16710
 CON16720
 CON16730
 CON16740
 CON16750
 CON16760
 CON16770
 CON16780
 CON16790
 CON16800


```

C      IF CHECK2 < 0, THAT INDICATES THAT THE SATURATION STEAM PRESSURE
C      IS BELOW PTLIM AND THUS THE STEAM TEMPERATURE IS BELOW COOLANT
C      INLET TEMPERATURE. THIS MEANS THAT NO HEAT IS REMOVED FROM THE
C      STEAM AND THE ALMTD IS 0. THE COOLANT OUTLET TEMP EQUALS THE
C      COOLANT INLET TEMP AND THE TUBE FILM TEMP IS THE SAME AS THE
C      COOLANT INLET TEMP. THE REMAINING VARIABLES ARE GIVEN THE LAST
C      GOOD VALUES RETURNED BY HETTRN.
C      ALMTD(I,J)=0.
C      HOMCI=0.
C      HOMCO=0.
C      VP SHH(I,J)=STBI
C      HEFF(I,J)=HEFF(I-1,J)
C      RC(I,J)=RC(I-1,J)
C      VNRE(I,J)=VNRE(I-1,J)
C      SHI(I,J)=SHI(I-1,J)
C      SHN(I,J)=SHN(I-1,J)
C      UN(I,J)=UN(I-1,J)
C      ROUT(I,J)=ROUT(I-1,J)
C      STFO=STBI
C      ENHF=ENHFSV
C
C      TB2=VP SHH(I,L)
C      HFG=HFGFN(SISAT(I,J))
C
C      CHECK TO SEE IF STEAM FLOW IN A SECTOR HAS ALREADY GONE TO 0
C      IF IT HAS CHECK1<0 AND DUMMY VALUES HAVE ALREADY BEEN ASSIGNED
C      TO WCND.
C
C      IF (CHECK1.LT.0.) GO TO 160
C
C      WCND(I,J)=UN(I,J)*AO*ALMTD(I,J)/HFG*TBNPR(I)
C
C      CHECK TC SEE IF SATURATION PRESSURE HAS ALREADY GONE BELOW PTLIM
C      AND DUMMY VARIABLES HAVE BEEN ASSIGNED TO DELPTP
C
C      IF (CHECK2.LT.0) GO TO 170
C
C      CALL PRSDRP (TAL,VMIX,WS(I,J),WNC,AXO,OD,VPSH,DELPTP,ENHF)
C      CONTINUE
C      JX=0
C
C      -----
C      SEC A5
C
C      TEMP OUTPUT SECTION
C
C      -----
C      ZZ ZSEC-A5
C      -----
C      SECA5=0.

```

```

CON17290
CON17300
CON17310
CON17320
CON17330
CON17340
CON17350
CON17360
CON17370
CON17380
CON17390
CON17400
CON17410
CON17420
CON17430
CON17440
CON17450
CON17460
CON17470
CON17480
CON17490
CON17500
CON17510
CON17520
CON17530
CON17540
CON17550
CON17560
CON17570
CON17580
CON17590
CON17600
CON17610
CON17620
CON17630
CON17640
CON17650
CON17660
CON17670
CON17680
CON17690
CON17700
CON17710
CON17720
CON17730
CON17740
CON17750
CON17760

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IF (SECA5.NE.1) GO TO 170
WRITE (6,860) I,J
WRITE (6,910) TB2, UN(I,L), VNRE(I,L)
WRITE (6,870) ALMD(I,L), HEFF(I,L)
WRITE (6,880) HOMC I, HOMCO
WRITE (6,890) RC(I,L), ROUT(I,L)
WRITE (6,900) SHI(I,L), SHN(I,L)
WRITE (6,920) WNC, WS(I,L)
WRITE (6,930) DELPTP, VPSH
WRITE (6,940) WCND(I,J), HFG
WRITE (6,950) AD, TBNPR(I)
C$-----
C SEC 6
C
C TEST FOR POSITIVE STM FLOW AND SAT PRESS
C
C WS = STM FLOW TO ONE ROW IN A SECTOR LB/HR
C WNCD = CONDENSATE FROM ONE ROW OF TUBES IN A SECTOR LB/HR
C WNC = N/C GAS FLOW TO ONE ROW OF TUBES IN A SECTOR LB/HR
C AXO = AREA OPEN TO FLOW IN A ROW OF A SECTOR
C GFLW = MASS FLW RATE OF MIXTURE IN A TUBE BUNDLE LB/(HR-FT**2)
C I = ROW NUMBER
C AMOLS = MOLES OF STEAM IN A ROW OF A SECTOR
C PMIX = PRESS OF STM - N/C MIXTURE
C DELPTP = PRESS DROP ACROSS A ROW OF TUBES
C AMWNC = MOLECULAR WEIGHT OF N/C
C PSAT = SATURATED STEAM PRESS
C J = SECTOR NUMBER
C JSAT = SATURATED STEAM TEMP - F
C TSAT = SATURATED STEAM TEMP - R
C PAL = DUMMY FOR PSAT
C CHECK1 = INDICATOR THAT NEG. STEAM FLOW HAS BEEN ENCOUNTERED
C CHECK2 = INDICATOR ANALYSIS
C          SSSKP SATFN(STBI) HAS BEEN
C          ENCOUNTERED IN SECTOR ANALYSIS
C PTEST = NEXT ROW PRESSURE VALUE TO CHECK FOR A BAD VALUE
C WTST = NEXT ROW STEAM FLOW TO CHECK FOR A NEGATIVE VALUE
C WTST = TEST VARIABLE WHICH IS NEGATIVE WHEN NEGATIVE STEAM FLOW
C          IS ENCOUNTERED IN A SECTOR
C PTST = TEST VARIABLE WHICH IS A NEGATIVE WHEN PSAT DROPS BELOW
C          PTLIM IN A SECTOR
C
C ZZZ SEC-6
C-----
C CHECK TO SEE IF STEAM FLOW HAS ALREADY GONE TO 0 IN AN EARLIER
C ROW CALCULATIO AND FIXUP HAS BEEN MADE TO WS AND WCND
C
C 170 IF (CHECK1.LT.0) GO TO 180
C      WS(I+1,J)=WS(I,J)-WCND(I,J)
C

```

CON1 7770
 CON1 7780
 CON1 7790
 CON1 7800
 CON1 7810
 CON1 7820
 CON1 7830
 CON1 7840
 CON1 7850
 CON1 7860
 CON1 7870
 CON1 7880
 CON1 7890
 CON1 7900
 CON1 7910
 CON1 7920
 CON1 7930
 CON1 7940
 CON1 7950
 CON1 7960
 CON1 7970
 CON1 7980
 CON1 7990
 CON1 8000
 CON1 8010
 CON1 8020
 CON1 8030
 CON1 8040
 CON1 8050
 CON1 8060
 CON1 8070
 CON1 8080
 CON1 8090
 CON1 8100
 CON1 8110
 CON1 8120
 CON1 8130
 CON1 8140
 CON1 8150
 CON1 8160
 CON1 8170
 CON1 8180
 CON1 8190
 CON1 8200
 CON1 8210
 CON1 8220
 CON1 8230
 CON1 8240


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C      DELPP IS AN ARBITRARY INCREMENTAL PRESS. DROP OVER REMAINING ROWS
C
C      DO 220 IK=IK2,NRWSP2
C      PSAT(IK,J)=PSAT(IK-1,J)-DELPP
C      CONTINUE
C      X1=PTST-PTLIM
C      X2=PSAT(I,J)
C
C      CALCULATE PTST WHICH IS A TEST VALUE INDICATING THAT SATURATION
C      PRESSURES WERE REACHED IN THE ANALYSIS FROM WHICH THE
C      CORRESPONDING TSAT IS EXCEEDED BY THE INLET COOLANT TEMP (IE PTLIM
C      WAS VIOLATED) A LARGE NEGATIVE VALUE FOR PTST INDICATES A BAD
C      CONDENSOR DESIGN IN WHICH PTLIM WAS VIOLATED EARLY IN THE SECTOR
C      ANALYSIS. AS THE CONDENSOR MODEL IMPROVES AND SATURATION PRESSURE
C      STAYS ABOVE PTLIM THEN PTST CONVERGES TO 0.
C
C      PTST(J)=(X1/(X2-X1))-FLOAT(NOROWS-I)
C      CHECK2=-2
C      PAL=PSAT(I+1,J)
C      PMIX(I+1,J)=PMIX(I,J)-DELPTP
C      IF (PMIX(I+1,J).LE..001) PMIX(I+1,J)=.001
C
C      IF (SECA6.EQ.1) WRITE (6,990) I,J
C      IF (SECA6.EQ.1) WRITE (6,1000) PSAT(I,J),PSAT(I+1,J)
C      IF (SECA6.EQ.1) WRITE (6,1010) PTST(J)
C
C      GO TO 230
C$-----
C      SEC 7
C
C      CORRECTION TO CONDENSATE RATE FOR
C      SENSIBLE HEAT
C
C      PAL = SAT STM PRESS
C      STFO = AVE TEMP OF OUTER TUBE FILM
C      UN = OVERALL HEAT X-FER CUEF FOR A ROW OF TUBES
C      AO = TUBE OUTER SURFACE AREA
C      ALMTD = LMTD
C      TBNPR = NUMBER OF TUBES IN A ROW OF A SECTOR
C      WS = STEAM FLOW TO A ROW OF A SECTOR
C      WNC = N/C FLOW TO A ROW OF A SECTOR
C      JGAS = FLAG FOR TYPE OF N/C
C      AMWNC = MOLECULAR WEIGHT OF N/C
C      TAL = SAT STM TEMP ENTERING SECTOR R
C      WCNDP = CONDENSATE FROM ONE ROW CORRECTED FOR SENSIBLE
C      HEAT CHANGE LB/HR
C      WCND = CONDENSATE FROM ONE ROW
C      JX =
C      IDENT = USER INPUT IDENTIFICATION STRING FOR RUN

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```

C      CHECK TO SEE IF STEAM FLOW HAS GONE TO 1 IN AN EARLIER ROW
C      CALCULATION AND SUBSEQUENT ROWS HAVE BEEN FIXED UP WITH DUMMY
C      VALUES. A NEGATIVE CHECK1 INDICATES FIXUP HAS BEEN MADE FOR
C      THAT SECTOR
C      IF (CHECK1.GE.0) GO TO 300
C      GO TO 350
C      WTEST=WS(I+1,J)
C      CHECK THE NEXT ROW TO SEE IF STEAM FLOW < 1
C      IF (WTEST-1.) 310,310,340
C      IF STEAM FLOW HAS GONE TO 1 THEN A STANDARD FIXUP IS MADE TO
C      ALLOW THE PROGRAM TO CONTINUE AND DUMMY VALUES FOR STEAM FLOW
C      AND CONDENSATE ARE ENTERED INTO THE REMAINING ROWS. IF STEAM FLOW
C      HAS GONE TO 1 THAT INDICATES ZERO STEAM FLOW. A VERY SMALL NUMBER
C      WILL BE USED TO SIMULATE 0 STEAM FLOW IN THE REMAINING ROWS.
C      IK1=I+1
C      RSTOGO=FLOAT(NROWS-I)
C      WSEND=(WTEST-1.)/(WS(I,J)-WTEST+1.)
C      WTEST(J)=WSEND-RSTOGO
C      IF (SECA6.EQ.1) WRITE (6,960) I,J
C      IF (SECA6.EQ.1) WRITE (6,970) WS(I,J),WTEST
C      IF (SECA6.EQ.1) WRITE (6,980) WTEST(J)
C      CALCULATE NEGATIVE FLOW RATE FOR USE IN CALCULATING EXITIFR
C      DO 320 KK=IK1,NCROWS
C      TUBDRY=TUBDRY+TBNPR(KK)
C      CONTINUE
C      TUBDRY=TUBDRY-WSEND*TBNPR(I)
C      TUBDRY=CUMDRY+TUBDRY
C      SUBRAT=WCND(I-1,J)/TBNPR(I)
C      SUBAVE=SUBAVE+SUBRAT
C      NBAD=NBAD+1
C      SUBFLO=SUBFLO+SUBRAT*TUBDRY
C      NRWSP1=NROWS+1
C      RSTOGO=FLOAT(NRWSP1-I)
C      WS(I,J)=(1.0)*(RSTOGO+1)
C      WCND(I,J)=WS(I,J)
C      DELWS=WS(I,J)/(RSTOGO+1)
C      DELWC=WCND(I,J)/(RSTOGO+1)

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CON15690
CON15700
CON15710
CON15720
CON15730
CON15740
CON15750
CON15760
CON15770
CON15780
CON15790
CON15800
CON15810
CON15820
CON15830
CON15840
CON15850
CON15860
CON15870
CON15880
CON15890
CON15900
CON15910
CON15920
CON15930
CON15940
CON15950
CON15960
CON15970
CON15980
CON15990
CON20000
CON20010
CON20020
CON20030
CON20040
CON20050
CON20060
CON20070
CON20080
CON20090
CON20100
CON20110
CON20120
CON20130
CON20140
CON20150
CON20160

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C      AMOLS = MOLES OF STM TO A ROW
C      TSAT = STM SATURATION TEMP R
C      PMIX = ST -N/C MIXTURE PRESSURE
C      AMWNC = MOLECULAR WEIGHT OF N/C GASSES
C      VNC = SPECIFIC VOLUME OF N/C GASSES
C      WNC = N/C FLOW TO A ROW OF A SECTOR
C      AMOLSC = MOLES OF A N/C TO A SECTOR
C      VEL = STM-N/C MIXTURE VELOCITY IN A ROW OF TUBES
C      WS = STM FLOW TO A ROW IN A SECTOR LB/HR
C      AOTFLW = AREA OPEN TO STM FLOW IN A ROW OF TUBES
C      SMTBI = SUM OF (COOLANT INLET TEMP*COOLANT FLOW RATE)
C      WB = COLLANT FLOW TO ONE TUBE IN THE CONDENSER
C      STBI = COOLANT INLET TEMP
C      TBNPR = NUMBER OF TUBES IN A ROW OF A SECTOR
C      SMTWB = ROW BY ROW SUM OF COOLANT FLOW
C      SMTB2 = ROW BY ROW SUM OF (COOLANT OUTLET TEMP*FLOW RATE)
C      TB2 = COOLANT OUTLET TEMP FOR A TUBE
C      SUMQ = SUM OF HEAT DUTY ON ROWS OF TUBES
C      UN = OVERALL HEAT X-FER COEF FOR A ROW
C      AO = TUBE OUTER SURFACE AREA
C      ALMTC = LMTD
C      QOA = HEAT FLUX ACROSS TUBE WALL
C      CUMDP = ACCUMULATED PRESS DROP FROM INLET TO CURRENT ROW
C      PMIX = PRESS OF STM-N/C MIXTURE
C      PHP = OVERALL POWER TO DRIVE THE COOLANT THROUGH THE CONDENSER
C      AND THE COOLER TUBES IN HP.
C      DELP = PRESSURE DROP ACROSS A SECTOR
C      DELPTP = PRESSURE DROP ACROSS A ROW
C      TAL = SAT STEAM TEMP R
C      I = ROW NUMBER
C      IT = DUMMY LOOP VARIABLE (IT = 1)
C      SVMIX1 = SPECIFIC VOLUME OF STM-N/C MIXTURE ENTERING CONDENSER
C      WGAS = FLOW OF N/C GAS TO A SECTOR
C      AOTFLW = AREA OPEN TO STEAM FLOW IN A ROW OF TUBES
C      C = USED TO ADJUST STEAM FLOW TO MATCH PRESS DROPS IN SEC-10
C      ZZZ SEC-8
C-----
370  VG=VGFN(TSAT,PMIX(I+1,J))
      AMOLSC=WNC/AMWNC
      VMIX=(10.729*TSAT)/(AMWNC*PMIX(I+1,J))
      VMIX=1.0/((AMOLSC/(VG*(AMOLSC+AMOLSC)))+(AMOLSC/(VNC*(AMOLSC+AMOLSC))
      1))
C      IF (I.EQ.NBROWS) GOTO 190
C      VEL(I+1,J)=(WS(I+1,J)+WNC)/(AOTFLW(I+1,J)*VMIX/3600.0
C      SMTBI=SMTBI+WB*STBI*TBNPR(I)
C      SMTWB=SMTWB+WB*TBNPR(I)
C      SMTB2=SMTB2+WB*TBNPR(I)*TB2
C      SUMQ=SUMQ+UN(I,J)*AO*ALMTD(I,J)*TBNPR(I)

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CON20650
CON20660
CON20670
CON20680
CON20690
CON20700
CON20710
CON20720
CON20730
CON20740
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CON20780
CON20790
CON20800
CON20810
CON20820
CON20830
CON20840
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CON20870
CON20880
CON20890
CON20900
CON20910
CON20920
CON20930
CON20940
CON20950
CON20960
CON20970
CON20980
CON20990
CON21000
CON21010
CON21020
CON21030
CON21040
CON21050
CON21060
CON21070
CON21080
CON21090
CON21100
CON21110
CON21120

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380      QQA(I,J)=UN(I,J)*ALMTD(I,J)
      CUMDP(I,J)=(PMIX(I)-PMIX(I+1,J)
      DELP(J)=DELP(J)+DELP(T)
      TAL=TSAT
      CONTINUE
      C(J)=(WS(1,J)+WGAS(J))/SQRT(DELP(J))
C
C      INCLUDE ONLY THOSE SECTORS WHERE STEAM FLOW HAS NOT GONE TO 0
C      IN STMSUM SO AS NOT TO INCLUDE DUMMY VALUES IN THE CALCULATION
C      OF EXIT STEAM FRACTION
C
390      IF (CHECK1.GE.0) STMSUM=STMSUM+WS(NOROWS+1,J)
      CONTINUE
C$-----
C      SEC A8
C      TEMP OUTPUT SECTION
C      ZZZ SEC-A8
C-----
      SECA8=0.
      IF (SECA8.NE.1.) GO TO 410
      WR ITE (6,1050)
      DO 400 I=1,ISEC
      WR ITE (6,1100) WTST(I),PTST(I)
400
410      CONTINUE
C$-----
C      SEC 9
C      EXIT FRACTION AND PRESS ADJUST
C      ZZZ SEC-9
C-----
      PMXEXT=0.
      DELPVE=0.
      PSUM=0.
      VELEXT=0.
      DO 420 I=1,ISEC
      PAL=PSAT(KSTOP,I)
      PSUM=PSUM+PAL/SECF LG
      VELEXT=VELEXT+VEL(NOROWS+1,I)
      DELPVE=DELPVE+DELP(I)
      PMXEXT=PMXEXT+PMIX(NOROWS+1,I)
420
C      MAKE PTST AND EXIT STEAM FRACTION VARIABLES CONTINUOUS
C      THIS IS DONE TO SATISFY AND ENHANCE OPTIMIZER CALCULATIONS. IF
C      THE VARIABLE PTST IS GIVEN THE LAST ROW +1 PRESSURE VALUES. IF
C      IT IS NON-NEGATIVE. IN ORDER TO MAKE EXIT STEAM FRACTION
C      (EXITFR) CONTINUOUS. THE VARIABLE FR IS CREATED WHICH INCORPORATES
C      NUMBERS OF ALL THE SECTORS. IF THE WTST VALUES ARE LARGE NEGATIVE
C      THROUGHOUT THE CONDENSOR THEN FR-->-1. AS WTST VALUES APPROACH 0 INDICATING STEAM
C      THEN FR IS THEN ADDED TO EXITFR

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C      TO MAKE IT CONTINUOUS FROM 1-->-1 REFLECTING THE CONDENSOR IN
C      NO STEAM IS EXTRACTED AND THE CONDENSOR WHERE ALL THE TUBES ARE
C      DRY.
C
C      430  I=1, ISEC
C      IF (PTST(I).GE.0.) PTST(I)=PSAT(NOROWS+1,I)-PTLIM
C      WT SAV=WT SAV+WTST(I)
C      FR =-(SUBFLO)/W SI
C      DELPVE=DELPVE/SECFLG
C      DO 450 J=1, ISEC
C
C      IF (ABS(DELP(J))/DELPVE-1.0)-.01) 450,450,440
C      IF (ABS(DELPVE).GT.0.001) GO TO 460
C      GO TO 580
C      CONTINUE
C      DLPVES=DELPVE
C      CONSM=0.0
C      WNCFR=WNCI/(WNCI+W SI)
C      DO 470 J=1, ISEC
C      CONSM=CCNSM+C(J)
C      CONTINUE
C      DELPVE=(WSAVE*SECFLG)**2/CONSM**2
C      DELPVE=(DELPVE+DLPVES)/2.0
C      DMAX=ABS(DELPVE-DELP(1))
C      JMAX=1
C      DO 480 J=2, ISEC
C      DHOLD=ABS(DELPVE-DELP(J))
C      IF (DHOLD.LE.DMAX) GO TO 480
C      DMAX=DHOLD
C      JMAX=J
C      CONTINUE
C      WSPT=0.0
C      WPT=0.0
C      DO 500 J=1, ISEC
C      WP(J)=C(J)*SQRT(DELPVE)
C      WSP(J)=WP(J)*W SI/(WNCI+W SI)
C      IF (J.EQ.JMAX) GO TO 500
C      FAC=WSP(J)-WS(1,J)
C      AFAC=ABS(FAC)
C      IF (AFAC.LT.1.E-3) GO TO 490
C      CHGSC=0.5*WS(1,J)*.26
C      IF (IPLCCP.GT.3) CHGSC=CHGSC/IPLCCP*2.0
C      WSP(J)=WS(1,J)+FAC/AFAC*SQRT(AFAC)*CHGSC
C      CONTINUE
C      WSPNT=WSP+WSPT+WSP(J)
C      WGAS(J)=WSP(J)*WNCFR
C      WP(J)=WSP(J)+WGAS(J)

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CON21610
CON21620
CON21630
CON21640
CON21650
CON21660
CON21670
CON21680
CON21690
CON21700
CON21710
CON21720
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CON21740
CON21750
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CON21770
CON21780
CON21790
CON21800
CON21810
CON21820
CON21830
CON21840
CON21850
CON21860
CON21870
CON21880
CON21890
CON21900
CON21910
CON21920
CON21930
CON21940
CON21950
CON21960
CON21970
CON21980
CON21990
CON22000
CON22010
CON22020
CON22030
CON22040
CON22050
CON22060
CON22070
CON22080

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CON2 2090
CON2 2100
CON2 2110
CON2 2120
CON2 2130
CON2 2140
CON2 2150
CON2 2160
CON2 2170
CON2 2180
CON2 2190
CON2 2200
CON2 2210
CON2 2220
CON2 2230
CON2 2240
CON2 2250
CON2 2260
CON2 2270
CON2 2280
CON2 2290
CON2 2300
CON2 2310
CON2 2320
CON2 2330
CON2 2340
CON2 2350
CON2 2360
CON2 2370
CON2 2380
CON2 2390
CON2 2400
CON2 2410
CON2 2420
CON2 2430
CON2 2440
CON2 2450
CON2 2460
CON2 2470
CON2 2480
CON2 2490
CON2 2500
CON2 2510
CON2 2520
CON2 2530
CON2 2540
CON2 2550
CON2 2560

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500      WPT=WPT+WP(J)
        CONTINUE
        WSP(JMAX)=WSI-WSPT
        WSPT=WSPT+WSP(JMAX)
        WGAS(JMAX)=WSP(JMAX)*WNCFR
        WP(JMAX)=WSP(JMAX)+WGAS(JMAX)
        WPT=WPT+WP(JMAX)
        ILOOP=ILOOP+1

        CHECK FOR STEAMM ADJUSTMENTS WHICH TRY TO PUT TO LOW A STEAM
        VALUE INTO A SECTOR

510      IFAIL=0
        DO 520 K=1,ISEC
        COMP=WSI/200.
        IF (WSP(K).LT. COMP) IFAIL=K
        IF (WSP(K).LT. COMP) WRITE (6,999)
        FORMAT (1X,'STEAM FLOW ADJUSTMENT WAS REQUIRED.')
        IF (WSP(K).LT. COMP) GO TO 530
        CONTINUE
        GO TO 550
520      DIFF=2.*COMP-WSP(IFAIL)
        WSP(IFAIL)=2.*COMP
        ADJ=DIFF/FLOAT(ISEC-1)
        DO 540 J=1,ISEC
        IF (J.EQ. IFAIL) GO TO 540
        WSP(J)=WSP(J)-ADJ
        CONTINUE
        GO TO 510
530      CONTINUE

540      DO 560 J=1,ISEC
        WS(1,J)=WSP(J)
        IF (ILOOP-50) 20,20,570
        WRITE (6,1110) IDENT
550      INCORPORATE FR INTO EXITFR TO MAKE IT CONTINUOUS FROM 1-->-1

560      EXITFR=STMSUM/WSI+FR
        WSEXIT=STMSUM
        VELEXT=VELEXT/SECFLG
        DELPVE = DELPVE/SECFLG
        PMXEXT=PMXEXT/SECFLG
        TSATEX=TSATFN(PSUM)
        AMLSEX=STMSUM/18.015
        IF ((STMSUM.LE.0.).AND.(PRCCLR.GT.0.)) AMLSEX=1.0/18.015
        EXITFC=1.
        IF (PRCCLR.EQ.0.0) EXITOA=EXITFR

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C          IF (PRCCLR.EQ.0.0) GO TO 590
C          CALL TO COOLER, IF THERE IS ONE
C          ENHIC=ENHI
C          ENHOC=ENHO
C          ENHFC=ENHF
C          SHWINC=SHWINV
C          TB2C=TB2
C          WBC=WB
C          IF (NBAD.GT.0) WCNDAY=SUBAVE/FLOAT(NBAD)
C          CALL COCLEX(TBDRYC,WCNDAY,SBFLOC,WSOUT)
C          CALCULATE OVERALL POWER REQUIRE TO DRIVE THE COOLANT
C          PHP=PHP*(SMWBC/(SMWBC+SMWB)+1.)
C          CALCULATE THE OVERALL COOLER/CONDENSER VOLUME
C          VOL1=(VOL1+VOLC)
C          CALCULATE OVERALL EXIT FRACTION
C          IF ((EXITFR.GT.0.) .AND. (EXITFC.GT.0.)) EXITOA=EXITFR*EXITFC
C          IF ((EXITFR.GT.0.) .AND. (EXITFC.GT.0.)) RETURN
C          EXITOA=WSOUT/WSI+(-(SUBFLO+SBFLOC)/WSI)
C          RETURN
590  FORMAT (1X,7HSECTOR ,I3,2X,5HROW ,I4)
600  FORMAT (1H,37H ***** OUTPUT FOR SEC-2 ***** )
610  FORMAT (1H,10HWSI ,10HWNCI ,10HWGAS(I) ,10HAOTFLW(I) )
620  FORMAT (1H,10HWS(1,I) ,10HSTSAT1 ,10HPMIX1 ,10HAMOLST ,10HVMIX1 ,10HG2 )
630  FORMAT (1H,10HHFG ,10HTSAT1 ,10HPSAT1 ,10HAMOLNC ,10HVMIX1 ,10HVG1 )
640  FORMAT (1H,10HTSAT1 ,10HPSAT1 ,10HAMOLNC ,10HVMIX1 ,10HVG1 ,1H,2F10.3)
650  FORMAT (1H,10HTSAT1 ,10HPSAT1 ,10HAMOLNC ,10HVMIX1 ,10HVG1 ,1H,2F10.3)
660  FORMAT (1H,10HTSAT1 ,10HPSAT1 ,10HAMOLNC ,10HVMIX1 ,10HVG1 ,1H,3F10.3)
670  FORMAT (1H,10HTSAT1 ,10HPSAT1 ,10HAMOLNC ,10HVMIX1 ,10HVG1 ,1H,10HSECFAR )
680  FORMAT (1H,10HTSAT1 ,10HPSAT1 ,10HAMOLNC ,10HVMIX1 ,10HVG1 ,1H,10HVEL1 )
690  FORMAT (1H,10HTSAT1 ,10HPSAT1 ,10HAMOLNC ,10HVMIX1 ,10HVG1 ,1H,10HDELPI )
700  FORMAT (1H,10HTSAT1 ,10HPSAT1 ,10HAMOLNC ,10HVMIX1 ,10HVG1 ,1H,2F10.4)
710  FORMAT (1H,10HTSAT1 ,10HPSAT1 ,10HAMOLNC ,10HVMIX1 ,10HVG1 ,1H,10HDELPI )
720  FORMAT (1H,10HTSAT1 ,10HPSAT1 ,10HAMOLNC ,10HVMIX1 ,10HVG1 ,1H,10HDELPI )
730  FORMAT (1H,10HTSAT1 ,10HPSAT1 ,10HAMOLNC ,10HVMIX1 ,10HVG1 ,1H,10HDELPI )
740  FORMAT (1H,10HTSAT1 ,10HPSAT1 ,10HAMOLNC ,10HVMIX1 ,10HVG1 ,1H,10HDELPI )
750  FORMAT (1H,10HTSAT1 ,10HPSAT1 ,10HAMOLNC ,10HVMIX1 ,10HVG1 ,1H,10HDELPI )
760  FORMAT (1H,10HTSAT1 ,10HPSAT1 ,10HAMOLNC ,10HVMIX1 ,10HVG1 ,1H,10HDELPI )
770  1RATON= ,E10.3)
780  1RATON= (1H,21HENTERING STM TEMPS - ,F6.2,4F ,F6.2,2H R)

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790  FORMAT (1H,32HENT SP. VOL. IN FT**3/LBM - STM=E10.3,9H N/C GAS=,CON23050
      1E10.3,5H MIXTURE=E10.3)CON23060
800  FORMAT (1H,23HENTERING STM VELOCITY =,E10.3,8H FT/SEC)CON23070
810  FORMAT (1H0,33H** OUTPUT FOR SEC-4 - ROW NUMBER,14)CON23080
820  FORMAT (1H0,33HTUBE WALL AREA - SQ.FT. - INSIDE=E10.3,10H OUTSIDECON23090
      1E=E10.3)CON23100
830  FORMAT (1H,9HTUBE ID=,F6.3,30H FT - TUBE X-SECTIONAL AREA =,E10CON23110
      1.3,6H FT**2)CON23120
840  FORMAT (1H,24HLOG MEAN TUBE THICKNESS=E10.3,25H FT - LOG MEAN TUCON23130
      1BE AREA=E10.3,6H FT**2)CON23140
850  FORMAT (1H,30HINVERSE WALL HEAT X-FER COEF =,E10.3,17H HR-FT**2-FCON23150
      1/B TU)CON23160
860  FORMAT (1H0,22HIN SECALC, SEC-5, ROW,,14,9H, SECTOR,,13)CON23170
870  FORMAT (1H,7HALMTD =,E10.3,8H HEFF =,E10.3)CON23180
880  FORMAT (1H,7HOMCI =,E10.3,8H HOMCO =,E10.3)CON23190
890  FORMAT (1H,7HRC =,E10.3,8H KOUT =,E10.3)CON23200
900  FORMAT (1H,7HSHI =,E10.3,8H SHN =,E10.3)CON23210
910  FORMAT (1H,7HTB2 =,E10.3,8H UN =,E10.3,8H VNRE =,E10.3)CON23220
920  FORMAT (1H,7HWC =,E10.3,8H WS =,E10.3)CON23230
930  FORMAT (1H,7HDELP TP=,E10.3,8H VPSH =,E10.3)CON23240
940  FORMAT (1H,7HWCND =,E10.3,8H HFG =,E10.3)CON23250
950  FORMAT (1H,7HAO =,E10.3,8H TBNPR =,E10.3)CON23260
960  FORMAT (1H0,32HNEG STEAM MASS FLOW IN SEC-6 ROW,I4,10H OF SECTOR,I1CON23270
      14)CON23280
970  FORMAT (1H,18HCURRENT FLOW RATE=,F12.6,17H NEXT FLOW RATE=,F12.6CON23290
      1)CON23300
980  FORMAT (1H,8HWIST(I)=,F12.6)CON23310
990  FORMAT (1H0,30HNEGATIVE SAT PRESS IN SEC-6 ROW,I4,9H SECTOR,14)CON23320
1000  FORMAT (1H,18HCURRENT SAT PRESS=,E10.3,11H NEXT ROW=,E10.3)CON23330
1010  FORMAT (1H,8HPIST(I)=,E10.3)CON23340
1020  FORMAT (1H,25HNO PROBLEMS IN SE-6, ROW=,I4,8H SECTOR=,I4)CON23350
1030  FORMAT (1H0,23HOUTPUT FROM SEC-7, ROW,I4,11H OF SECTOR,I4)CON23360
1040  FORMAT (1H,9HWS(I,J)=,E10.3,13H WCND(I,J)=,E10.3)CON23370
1050  FORMAT (1H,6HTSAT =,E10.3,14H STSAT(I,J)=,E10.3)CON23380
1060  FORMAT (1H,11HGFLW(I,J)=,E10.3,13H PSAT(I,J)=,E10.3)CON23390
1070  FORMAT (1H,28HSEC-7, BELOW 130, WCND(I,J)=,E10.3,8H WCNUP=,E10.3CON23400
      1)CON23410
1080  FORMAT (1H,22HSEC-7, BELOW 130, TAL=,E10.3,6H TSAT=,E10.3)CON23420
1090  FORMAT (1H1,3X,4HWTST,8X,4HPIST,)CON23430
1100  FORMAT (1H,E10.3,2X,E10.3,2X,E10.3)CON23440
1110  FORMAT (1H1,19A4,A3,40HNO CONVERGENCE IN COND. DELTA P (IPL00P))CON23450
1120  FORMAT (1H1,19A4,A3/1H0,5X,2H1=,I2,5X,2HJ=,I1/1H0,24HNO CONVERGENCCON23460
      1E FOR WCND(I,I2,1H,,11,1H))CON23470
      1ENDCON23480
      *****CON23490
      *****CON23500
      *****CON23510
      *****CON23520

```



```

SUBROUTINE HETTRN (ALST,AXO,IR,LJ,CD,WB,SHW,INV,STSAT,TID,WNC,WST,A
1LMTD,ENHF,HEFF,HOMCI,HOMCD,RC,ROUT,SHI,SHNF,STFO,TB2,VNF,XNRE,ENHI
2,ENHO,SDD)
-----
C  HET 1
C  SUBROUTINE HETTRN
C
C  HETTRN CALLED BY: SECALC
C  SUBROUTINES CALLED BY HETTRN:
C    , DSVTY
C  FUNCTION ROUTINES CALLED BY HETTRN:
C    , PSATFN , SMUFN , AMUFN , HFCFN , SKBFN ,
C    , BMUFN , CPFN , ROEFN ,
C
C  DEC 69 REV. COLBJ,FDAVE
C  COMMON /INPT/ BC(6),BE(6),ENH(6),ENI(6),ENO(6),FOUL,PHI,SKW,STBI,S
1TSA1,TUBESW,NCIR,IOP1,PIST(15),SECWID,JBFAF,ISEC,JBAF(16),JGAS,MD
2IAM,MPITCH,NEE,NEI,NETE(6),NETI(6),NRNE(6),NRNI(6),NOROWS,WTST(15)
3,WST,PFILL
C  COMMON /ORC2/ ANT(100,15),STB2ES,VW(100)
C  COMMON /CONST/ AMWNC,CBI,CPB,PI,SG,IFIRST
C  DIMENSION R2(3),E2OK(3)
C  HETON=0
C  IF (HETCN.NE.1.) GO TO 10
C  M5=1
C  WRITE (6,310) M5
C  WRITE (6,320) IR,LJ
C  WRITE (6,330) OD,TID
C  WRITE (6,340) SHW,INV,STSAT
C  WRITE (6,350) WST,WNC
C
C  DATA INITIALIZATION:
C
C  BB = COOLANT FLOW IN TUBE LBM/HR-FT**2
C  DGFLM = CALCULATED TEMP DROP ACROSS FILM???
C  DLFLM = CALCULATED TEMP DROP ACROSS LIQUID FILM???
C  HEFF = NONCONDENSIBLE FILM HEAT TRANSFER COEF, BTU/HR-FT**2-F
C  UEST = ESTIMATED OVERALL HEAT TRANSFER COEF
C
C  VALUES FOR USE IN SUBROUTINE DSVTY
C  E2OK = FORCE CONSTANT FOR NONCOND GAS
C  R2 = GAS COLLISION DIAMETER IN ANGSTROMS
C  ZZZ HET-2
C
C  IF (WNC.EQ.0.) DGFLM=0.
C  IF ((LJ.EQ.1).AND.(IR.EQ.1)) GO TO 20
C
C  CON23530
C  CON23540
C  CON23550
C  CON23560
C  CON23570
C  CON23580
C  CON23590
C  CON23600
C  CON23610
C  CON23620
C  CON23630
C  CON23640
C  CON23650
C  CON23660
C  CON23670
C  CON23680
C  CON23690
C  CON23700
C  CON23710
C  CON23720
C  CON23730
C  CON23740
C  CON23750
C  CON23760
C  CON23770
C  CON23780
C  CON23790
C  CON23800
C  CON23810
C  CON23820
C  CON23830
C  CON23840
C  CON23850
C  CON23860
C  CON23870
C  CON23880
C  CON23890
C  CON23900
C  CON23910
C  CON23920
C  CON23930
C  CON23940
C  CON23950
C  CON23960
C  CON23970
C  CON23980
C  CON23990
C  CON24000

```


CON24010
CON24020
CON24030
CON24040
CON24050
CON24060
CON24070
CON24080
CON24090
CON24100
CON24110
CON24120
CON24130
CON24140
CON24150
CON24160
CON24170
CON24180
CON24190
CON24200
CON24210
CON24220
CON24230
CON24240
CON24250
CON24260
CON24270
CON24280
CON24290
CON24300
CON24310
CON24320
CON24330
CON24340
CON24350
CON24360
CON24370
CON24380
CON24390
CON24400
CON24410
CON24420
CON24430
CON24440
CON24450
CON24460
CON24470
CON24480

```
IF (IR.EQ.1) GO TO 30
UEST=UESTSV
HEFF=HEFFSV
DGFLM=DGFLMS
DLFLM=DLFLMS
TB2=TB2SV
XNRET=XNRES
GO TO 40
IF (IFIRST.EQ.0) GO TO 30
HEFF=2100.
DATA UEST,DGFLM,DLFLM,XNRET/700.,0.05,2.0,100000./
DATA R2,E2OK/3.617,3.996,3.838,97.0,190.0,140.75/
TB2=STB2ES
IFIRST=0
GO TO 40
UEST=UEST1
HEFF=HEFF1
DGFLM=DGFLM1
DLFLM=DLFLM1
TB2=TB21
XNRET=XNRE1
CONTINUE
```

20

30

40

FDAVE IS AN INPUT ITEM WITH A VALUE BETWEEN 0 AND 1 RELATED TO TUBE
SPACING AND ORIENTATION.
FDAVE=C.5

```
BNF=ANT(IR,LJ)
IF (BNF.LT.1.) BNF=1.
BB=W8/(PI*TID*TID/4.)
AI=PI*TID*ALST
AO=PI*CC*ALST
ENHI=1.
ENHF=1.
ENHO=1.
IF (NEI.EQ.0) GO TO 60
DO 50 I=1,NEI
ITEM1=NRNI(I)
ITEM2=ITEM1+NETI(I)-1
IF (I.GE.ITEM1).AND.(I.LE.ITEM2) ENHI=ENI(I)
IF (NEI.EQ.0) GO TO 80
DO 70 I=1,NEE
ITEM1=NRNE(I)
ITEM2=ITEM1+NETE(I)-1
IF (I.LT.ITEM.OR.I.GT.ITEM2) GO TO 70
ENNO=ENO(I)
```

50

60


```

C ZZZ HET-3
C-----
90 K=0
    WBOOTH=WST+WNC
    RC=WNC/(WBOOTH)
    TSAT=TSAT+459.69
    GMAX=(WBOOTH)/AXO
    PSAT=PSATFN(TSAT)
    AMOLSS=WST/18.015
    AMOLSC=WNC/AMWNC
    AMOLT=AMOLSS+AMOLSC
    AMWAV=(WBOOTH)/AMOLT
    RMWNC=SQRT((AMWNC)*AMOLSC/AMOLT)
    RMWSC=4.24444*AMOLSS/AMOLT
    AVIS=(SMJFN(TSAT)*RMWSC+AMUFN(TSAT,JGAS)*RMWNC)/(RMWSC+RMWNC)
    HSTO=HFGFN(TSAT)
    STBAVE=(STBI+TB2)/2.0
    DELFLM=DLFLM
    DLGFLM=DGFLM
    STWO=TSAT-DELFLM-DLGFLM
    STFO=(TSAT+STWO)/2.0
    XNRE=OD*(WBOOTH)/(AXO*3600.*AVIS)
    IF (HETCN.NE.1.) GO TO 100
M5=3
    WRITE (6,310) M5
    WRITE (6,410) GMAX,PSAT
    WRITE (6,420) AMOLSS,AMOLSC,AMOLT
    WRITE (6,430) RMWNC,RMWSC
    WRITE (6,440) AVIS,HSTO,XNRE
    WRITE (6,450) STBAVE,STFO,STWO
    WRITE (6,460) DELFLM,DLGFLM
C$-----
C HET 5
C-----
C COLBY J FACTOR CALCULATION AND BRANCH
C WNCI = TOTAL WEIGHT OF NONCOND ENTERING CONDENSER
C IF NO NONCOND, THEN SKIP DIFFUSIVITY CALCULATION.
C ZZZ HET-5
C-----
100 CONTINUE
    COLBJ=EXP(0.53883-0.544*ALOG(XNRE))
    IF (HETCN.NE.1.) GO TO 110
M5=5
    WRITE (6,310) M5
    WRITE (6,470) COLBJ,WNC
    IF (WNC.EQ.0.) GO TO 120
110

```

```

CON24970
CON24980
CON24990
CON25000
CON25010
CON25020
CON25030
CON25040
CON25050
CON25060
CON25070
CON25080
CON25090
CON25100
CON25110
CON25120
CON25130
CON25140
CON25150
CON25160
CON25170
CON25180
CON25190
CON25200
CON25210
CON25220
CON25230
CON25240
CON25250
CON25260
CON25270
CON25280
CON25290
CON25300
CON25310
CON25320
CON25330
CON25340
CON25350
CON25360
CON25370
CON25380
CON25390
CON25400
CON25410
CON25420
CON25430
CON25440

```


C\$-----

HET 6

PREPARATION FOR CALL TO SUBROUTINE DFSVTY;
CALCULATE SCHMIDT NUMBER AND PARAMETER CJ.

THIS SECTION IS BRANCHED AROUND IF NO NONCOND GASSES ARE
PRESENT.

AMWAV = STEAM-N/C GAS MIXTURE MOLECULAR WEIGHT
AVIS = AVG VISCOSITY OF STEAM-N/C GAS MIXTURE
CJ = PARAMETER USED IN CALCULATION OF THE HEAT TRANSFER COEF.
SEE EQUATION 11 IN APP B
DD = DIFFUSIVITY (SEE NOTE)
DDG = DUMMY PASSING PARAMETER FOR SUB DFSVTY
E2OK = N/C GAS FORCE CONSTANT (SEE SUB DFSVTY)
PATM = SAT PRESS IN ATM
PGB = NONCOND PARTIAL PRESSURE
PSAT = STEAM SATURATION PRESS PSI
R = N/C GAS COLLISION DIAMETER (SEE SUB DFSVTY)
TSAT = STEAM SATURATION TEMP
TSATK = SATURATION TEMP IN DEGREES KELVIN
XNSCH = SCHMIDT NUMBER

NOTE: PATM, PGB, AND TSATK ARE USED FOR CALL TO DFSVTY
SEE NOTE IN HET-3, BUT REM PGB IS USED AGAIN
IN HET-12.

NOTE: 18.015 IS THE MOLECULAR WEIGHT OF WATER
2.655 IS THE STEAM COLLISION DIAMETER IN ANGSTROMS
363. IS THE STEAM FORCE CONSTANT USED IN DFSVTY

NOTE: ONLY DG (DIFFUSIVITY IN CM**2/SEC) IS MODIFIED IN
SUBROUTINE DFSVTY.

NOTE: 0.258 CONVERTS UNITS OF DIFFUSIVITY FROM
CM**2/SEC TO FT**2/HR, AND 3600 CONVERTS
VISCOSITY FROM LB(F)-HR TO LB(F)-SEC.
IN XNSCH CALC, .258*3600 = 928.8

ZZZ HET-6

PATM=PSAT/14.7
PGB=PSAT*(AMOL SC/AMOLSS)
TSATK=TSAT/1.8
CALL DFSVTY (TSATK,PATM,18.015,AMWNC,2.655,R2(JGAS),363.,E2OK(JGAS
1),0.,0.,DG,DGG)
XNSCH=(AVIS*TSAT*10.73/((PSAT+PGB)*AMWAV*DG))*928.8

CON255450
CON255460
CON255470
CON255480
CON255490
CON255500
CON255510
CON255520
CON255530
CON255540
CON255550
CON255560
CON255570
CON255580
CON255590
CON255600
CON255610
CON255620
CON255630
CON255640
CON255650
CON255660
CON255670
CON255680
CON255690
CON255700
CON255710
CON255720
CON255730
CON255740
CON255750
CON255760
CON255770
CON255780
CON255790
CON255800
CON255810
CON255820
CON255830
CON255840
CON255850
CON255860
CON255870
CON255880
CON255890
CON255900
CON255910
CON255920


```

CJ=COLBJ*GMAX/(XNSCH**0.066667)
IF (HETCN.NE.1.) GO TO 120
M5=6
WRITE (6,310) M5

```

```

C$-----

```

```

C HET 7

```

```

      BEGIN ITERATIVE CALCULATIONS FOR:
      HEAT TRANSFER COEF
      TEMP DROP ACROSS LIQUID FILM
      TEMP DROP ACROSS N/C FILM (IF ANY)

```

```

      THE PROGRAM CONTINUES TO RETURN HERE UNTIL ONE
      OR MORE OF THE FOLLOWING CONDITIONS ARE MET:
      1) CHANGE IN THE CALCULATED HEAT TRANSFER COEF IS
      LESS THAN OF THE PREVIOUS VALUE AND AT LEAST
      FOUR ITERATIONS HAVE BEEN MADE.
      2) CHANGE IN THE CALCULATED TEMP ACROSS THE LIQUID
      FILM IS LESS THAN .01 DEGREES AND THE CHANGE
      IN CALCULATED TEMP DIFF ACROSS THE GAS FILM
      IS LESS THEN .001 DEGREES.
      3) TEN ITERATIONS HAVE BEEN COMPLETED.

```

```

      AI = INTERNAL SURFACE AREA OF ONE TUBE FT**2
      AO = EXTERNAL SURFACE AREA OF ONE TUBE FT**2
      BB = COOLANT FLOW IN TUBE LBM/HR-FT**2
      BMU = COOLANT VISCOSITY LBM/HR-FT
      BNF = TUBE FLOODING FACTOR
      CBI = COOLANT CONCENTRATION
      ENHI = INTERNAL FILM COEF ENHANCEMENT FACTOR
      ENHO = EXTERNAL FILM COEF ENHANCEMENT FACTOR
      RIN = RESISTANCE TO HEAT TRANSFER DUE INNER FILM
      TID = TUBE ID FT
      OD = TUBE OD FT
      SHBI = SPECIFIC HEAT OF COOLANT BTU/LBM-F
      SHI = INTERNAL FILM HEAT TRANSFER COEF BTU/HR-FT**2-F
      SHMK = EXTERNAL FILM HEAT TRANSFER COEF BTU/HR-FT**2-F
      SKB = THERMAL CONDUCTIVITY OF THE COOLANT BTU-FT/HR-FT**2-F
      SKBO = THERMAL CONDUCTIVITY OF OUTER LIQUID FILM
      STBAVE = AVE COOLANT TEMP.
      STFO = AVE TEMP OF OUTER LIQUID FILM (SEE NOTE)
      STSAT = LOCAL SAT STM TEMP
      TB2 = OUTLET COOLANT TEMP. (SEE NOTE ON TB2 IN HET-3)
      WB = COOLANT FLOW RATE PER TUBE LBM/HR
      XNRB = REYNOLDS NUMBER OF COOLANT
      XNPRB = PRANDTL NUMBER OF COOLANT

```

```

      NOTE: SHMK IS CALCULATED FIRST WITHOUT TAKING FLOODING
      INTO ACCOUNT. SEE NUSSELT EQN PG. 16.

```

```

CON25930
CON25940
CON25950
CON25960
CON25970
CON25980
CON25990
CON26000
CON26010
CON26020
CON26030
CON26040
CON26050
CON26060
CON26070
CON26080
CON26090
CON26100
CON26110
CON26120
CON26130
CON26140
CON26150
CON26160
CON26170
CON26180
CON26190
CON26200
CON26210
CON26220
CON26230
CON26240
CON26250
CON26260
CON26270
CON26280
CON26290
CON26300
CON26310
CON26320
CON26330
CON26340
CON26350
CON26360
CON26370
CON26380
CON26390
CON26400

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```

C      NOTE: STFO IS INITIALLY GUESSED BASED ON STWO WHICH
C      IS IN TURN BASED ON DLFLM AND DGFLM. DGFLM AND
C      DLFLM ARE INITIALIZED BY DATA STATEMENT UPON
C      ENTERING THIS SUBROUTINE. ALL FOUR OF
C      THESE VARIABLES ARE UPDATED BELOW.
C
C      ZZZ HET-7
C-----
120  TB2=STSAT-(STSAT-STBI)/EXP(UEST*AO/(WB*CPFN(CBI,STBAVE)))
      STBAVE=(STBI+TB2)/2.
      SKBO=SKBFN(0.,STFO)
      SKB=SKBFN(CBI,STBAVE)
      BMU=BMUFN(CBI,STBAVE)
      SHBI=CPFN(CBI,STBAVE)
      XNREB=TBID*BB/BMU
      XNPRB=SHBI*BMU/SKB
      SHI=0.023*(XNREB)**0.8*XNPRB**0.33333*(SKB/TID)*ENHI
      RIN=AO/(SHI*AI)
      SHMK=0.725*(SKBO**3*ROEFN(0.0,STFO)**2*HFGFN(STFO)*416975040.0/(BM
      1UFN(0.,STFO)*OD*DELFLM))**.25*ENHO
C
C      THIS IS THE PLACE TO ENTER CORRECTIONS TO THE EXTERNAL FILM COEFF. THE
C      FOLLOWING IS A CORRECTION ACCORDING TO FUJII, ET AL. TO ACCOUNT FOR
C      THE EFFECTS OF VAPOR SHEAR AT HIGH VAPOR REYNOLDS NUMBERS.
C
      SHMK1=SHMK
      ANUS=SHMK*OD/SKB0
      EMUL=BMUFN(0.,STSAT)
      EMUS=SMUFN(TSAT)*3600.
      RHOL=ROEFN(0.,STSAT)
      RHOS=1./VGFN(TSAT,PSAT)
      VISRAT=(EMUS/RHOS)/(EMUL/RHOL)
      RETP=XNRE*VISRAT
      TERM=SQRT(RETP)/ANUS
      IF (TERM.LT..278) GO TO 130
      ANUSP=1.24*(ANUS**0.8)*(RETP**0.1)
      GO TO 140
130  ANUSP=ANUS*0.70/0.725
140  SHMKP=ANUSP*SKBQ/OD
      SHMK=SHMKP
C
      IF (HETCN.NE.1.) GO TO 150
M5=7
WRITE (6,310) M5
WRITE (6,480) TB2,STBAVE

```

CON26410
 CON26420
 CON26430
 CON26440
 CON26450
 CON26460
 CON26470
 CON26480
 CON26490
 CON26500
 CON26510
 CON26520
 CON26530
 CON26540
 CON26550
 CON26560
 CON26570
 CON26580
 CON26590
 CON26600
 CON26610
 CON26620
 CON26630
 CON26640
 CON26650
 CON26660
 CON26670
 CON26680
 CON26690
 CON26700
 CON26710
 CON26720
 CON26730
 CON26740
 CON26750
 CON26760
 CON26770
 CON26780
 CON26790
 CON26800
 CON26810
 CON26820
 CON26830
 CON26840
 CON26850
 CON26860
 CON26870
 CON26880


```

C$-----
C HET 9
WR ITE (6,490) SKB0,SKB
WR ITE (6,500) BMU,SHBI
WR ITE (6,510) XNREB,XNPRB
WR ITE (6,520) SHI,RIN,SHMK
TEST FOR TUBE FLOODING, MODIFY
SHMK AS NEEDED
THIS SECTION MODIFIES THE EXTERNAL HEAT TRANSFER
COEF GIVEN BY NUSSELT EQN. FOR TUBE FLOODING.
BNF = CALCULATED TUBE FLOODING FACTOR (SEE NOTE BELOW)
FDAVE = INPUT VALUE OF TUBE FLOODING FACTOR (SEE NOTE BELOW)
FDAVN & FDAVM ARE INTERMEDIATE VALUES USED LATER.
FOUL = TUBE FOULING FACTOR INPUT BY USER
RFACT = SUM OF THERMAL RESISTANCES NEGLECTING GAS FILM
RIN = RESISTANCE TO HEAT X-FER DUE TO INTERNAL FILM
SHNF = EXTERIOR HEAT X-FER COEF, SHMK MODIFIED FOR RAIN
SHWINV = INVERSE OF WALL HEAT TRANSFER CCOEF
NOTE:ADD CALC FOR FDAVE CURRENTLY UNDEF
NOTE:CALC FOR FDAVM AND FDAVN SHOULD BE MOVED
ABOVE LINE 40.
INUNDATION EFFECT MAY BE BYPASSED BY SETTING BNF = 1.
BNF = 1.
IF (FDAVE.LT.0..OR..BNF.LT.2.) GO TO 160
FDAVN=0.6*FDAVE+(1.-0.5647*FDAVE)/BNF**0.20
FDAVM=0.6*FDAVE+(1.-0.5647*FDAVE)/(BNF-1.)**0.20
SHNF=SHMK*(BNF*FDAVN-(BNF-1.)*FDAVM)
GO TO 170
SHNF=0.95*(BNF**0.9-(BNF-1.)***0.9)*SHMK
RFACT=RIN+SHWINV+FOUL+1./SHNF
IF (HETCN.NE.1.) GO TO 180
M5=9
WR ITE (6,310) M5
WR ITE (6,530) FDAVN,FDAVM
WR ITE (6,540) SHNF,RFACT
C$-----
C HET 10

```



```

      CALCULATE LOG-MEAN TEMP-DIFFERENCE
      THIS CALCULATION IS COMPLETED ONCE DURING EACH
      ITERATION. UPON COMPLETION, A TEST IS MADE
      FOR THE PRESENCE OF N/C GASSES. IF ANY
      GAS IS PRESENT, PROGRAM BRANCHES TO LINE 110 (HET-12).
      IF NOT, THEN PROGRAM FLOWS TO HET-11, BELOW.

      ALMDT = LMTD      DIFF BETWEEN STEAM AND COOLANT OUTLET
      HOMCO = TEMP      DIFF BETWEEN STEAM AND COOLANT AT INLET
      HOMCI = TEMP
      JRC = PRINT CONTROL FLAG
      RODT = RATIO OF HOMCI TO HOMCO
      STBI = COOLANT INLET TEMP
      STSAT = STEAM SATURATION TEMP
      TB2 = COOLANT OUTLET TEMP
      WNCI = TOTAL N/C GAS FLOW TO CONDENSER

      ZZZ HET-10
      -----
180  HOMCI=STSAT-STBI
      HOMCO=STSAT-TB2
      IF (HOMCO.LT.1.E-15) WRITE(6,20600) HOMCI,HOMCO
      IF (HOMCI.LT.1.E-15) WRITE(6,20600) HOMCI,HOMCO
      IF (HOMCO.LT.1.E-15) HOMCO=1.E-15
      RODT=HOMCI/HOMCO
      IF INLET/OUTLET DELTA T RATIO IS NEAR 1.0, ALMTD = ARITH. AVG.
      IF (RODT.GT.1.1) GO TO 190
      ALMTD=0.5*(HOMCI+HOMCO)
      IF (HETCN.EQ.1.) WRITE (6,670) STSAT,STBI,TB2,LJ,IR,II
      GO TO 200
190  ALMTD=(HOMCI-HOMCO)/ALOG(RODT)
200  CONTINUE
      IF (HETCN.NE.1.) GO TO 210
      M5=10
      WRITE (6,310) M5
      WRITE (6,550) HOMCI,HOMCO
      WRITE (6,560) RODT,ALMTD
      IF (WNC.NE.0.) GO TO 230
210  -----
      HET 11
      CALCULATE FINAL HEAT TRANSFER PARAMETERS
      FOR NO N/C GASSES ENTERING THE CONDENSER.

```

CON27370
 CON27380
 CON27390
 CON27400
 CON27410
 CON27420
 CON27430
 CON27440
 CON27450
 CON27460
 CON27470
 CON27480
 CON27490
 CON27500
 CON27510
 CON27520
 CON27530
 CON27540
 CON27550
 CON27560
 CON27570
 CON27580
 CON27590
 CON27600
 CON27610
 CON27620
 CON27630
 CON27640
 CON27650
 CON27660
 CON27670
 CON27680
 CON27690
 CON27700
 CON27710
 CON27720
 CON27730
 CON27740
 CON27750
 CON27760
 CON27770
 CON27780
 CON27790
 CON27800
 CON27810
 CON27820
 CON27830
 CON27840


```

CON27850
CON27860
CON27870
CON27880
CON27890
CON27900
CON27910
CON27920
CON27930
CON27940
CON27950
CON27960
CON27970
CON27980
CON27990
CON28000
CON28010
CON28020
CON28030
CON28040
CON28050
CON28060
CON28070
CON28080
CON28090
CON28100
CON28110
CON28120
CON28130
CON28140
CON28150
CON28160
CON28170
CON28180
CON28190
CON28200
CON28210
CON28220
CON28230
CON28240
CON28250
CON28260
CON28270
CON28280
CON28290
CON28300
CON28310
CON28320

CONTROL PASSES FROM THIS ROUTINE TO ONE OF TWO PLACES,
LINE 40 (HET-7) TO START A NEW ITERATION, OR TO
LINE 170 (HET-13) TO TERMINATE THIS HETTRN CALL
WHEN ONE OF THE FOLLOWING CRITERIA ARE MET:
1) 10 ITERATIONS COMPLETED WITHOUT CONDITION 2 BEING
MET.
2) CHANGE IN HEAT TRANSFER COEF. LT. 1 PERCENT
BETWEEN SUCCESSIVE ITERATIONS.

DELFLM = SAVED VALUE OF DLFLM, USED ONLY FOR PRINT
DGFLM = TEMP DROP ACROSS GAS FILM, SET TO 0 HERE
DLFLM = TEMP DROP ACROSS LIQUID CONDENSATE FILM
K = COUNTER FOR NUMBER OF ITERATIONS
RFACT = SUM OF THERMAL RESISTANCES
ROUT = EXTERNAL FILM RESISTANCE. (SEE NOTE)
OD = TUBE OD
SHNF = EXTERNAL FILM HEAT X-FER COEF
SKBO = THERMAL CONDUCTIVITY OF EXTERNAL LIQUID FILM
STCO = TEMP AT SURFACE OF OUTER LIQUID FILM
STFO = AVE TEMP OF OUTER LIQUID FILM
STBAVE = AVE COOLANT TEMP
STSAT = STEAM SATURATION TEMP
STW0 = TUBE OUTER WALL TEMP
UEST = VALUE OF UEST FROM LAST ITERATION
UESTS = SAVED VALUE OF UEST, USED ONLY FOR PRINT
UESTT = SAVED HEAT X-FER COEF.
XNU = NUSSELT NUMBER

NOTE: CONSIDER REQUIREMENT FOR MINIMUM OF
FOUR ITERATIONS PRIOR TO UPDATING UEST. COULD
BE AN ALLOWANCE FOR DELFLM TO CATCH UP IN
CONVERGENCE. IF SO, TEST DELFLM FOR ITS
OWN CONVERGENCE, ELIMINATE MIN OF 4 ITERATIONS.

NOTE: ROUT WILL BE REDEFINED AS 1/ROUT
IN HET-13 PRIOR TO EXIT FROM HETTRN

ZZZ HET-11
-----
ROUT=1./SHNF
UEST=1./RFACT
XNU=SHNF*OD/SKBO
IF (ABS(UEST-UESTT)/UEST.LT..01) GO TO 290
UESTSV=UEST
UEST=UESTT
IF (K.GT.4) UEST=0.5*(UEST+UESTSV)
STW0=STSAT-(STSAT-STBAVE)*UEST/SHNF

```


STCO=STSAT
STFO=(STCO+STWO)/2.

DGFLM=0.
DLFLM=STCO-STWO
DELFM=DLFLM

K=K+1
IF (HETCN.NE.1.) GO TO 220

M5=11
WR ITE (6,310) M5
WR ITE (6,570) UTEST,UESTSV
WR ITE (6,580) XNU,ROUT

WR ITE (6,590) STWO,STCO,STFO
WR ITE (6,600) DGFLM,DLFLM
IF (K.LT.20) GO TO 120
WR ITE (6,640) UTESTSV,UTEST
GO TO 290

220

C\$-----
C HET 12

CALCULATE FINAL HEAT TRANSFER PARAMETERS
FOR N/C GASSES ENTERING CONDENSER

PROGRAM CONTROL COMES HERE FROM HET-10. CONTROL PASSES
FROM HERE TO ONE OF TWO PLACES, EITHER LINE 40 (HET-7) TO
BEGIN THE NEXT ITERATION, OR TO LINE 170 (HET-13) WHEN
ONE OF THE FOLLOWING CONDITIONS ARE MET:
1) 10 ITERATIONS HAVE BEEN COMPLETED PRIOR TO MEETING
CONDITION 2.
2) CHANGES IN UEST.LT..001, IN DLFLM.LT.
.01 AND IN DGFLM.LT..001

ALMTD = LMTD
BONE = CONSTANT IN SOLUTION OF EQN 11A, APP A.

CJ - SEE BONE
CONE - SEE BONE
DELFM = DLFLM FROM PREVIOUS ITERATION
DELTAU = HEAT DUTY ON HEAT X-FER SURFACE
DENOM =
DGFLM = TEMP DIFF ACROSS N/C GAS FILM
DLFLM = TEMP DIFF ACROSS CONDENSATE FILM
DLGFLM = DGFLM FROM PREVIOUS ITERATION
DTDP = INVERSE OF DPDT (SEE NOTE)
HEFF = N/C GAS HEAT X-FER COEF
HSTO = STEAM LATENT HEAT OF VAPORIZATION
IBT = PRINT CONTROL
IR = CURRENT ROW NUMBER
K = ITERATION COUNTER
LJ = CURRENT SECTOR NUMBER
PGB = PARTIAL PRESSURE OF N/C GASSES

CON2 8330
CON2 8340
CON2 8350
CON2 8360
CON2 8370
CON2 8380
CON2 8390
CON2 8400
CON2 8410
CON2 8420
CON2 8430
CON2 8440
CON2 8450
CON2 8460
CON2 8470
CON2 8480
CON2 8490
CON2 8500
CON2 8510
CON2 8520
CON2 8530
CON2 8540
CON2 8550
CON2 8560
CON2 8570
CON2 8580
CON2 8590
CON2 8600
CON2 8610
CON2 8620
CON2 8630
CON2 8640
CON2 8650
CON2 8660
CON2 8670
CON2 8680
CON2 8690
CON2 8700
CON2 8710
CON2 8720
CON2 8730
CON2 8740
CON2 8750
CON2 8760
CON2 8770
CON2 8780
CON2 8790
CON2 8800


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CUN2 8810
CUN2 8820
CUN2 8830
CUN2 8840
CUN2 8850
CUN2 8860
CUN2 8870
CUN2 8880
CUN2 8890
CUN2 8900
CUN2 8910
CUN2 8920
CUN2 8930
CUN2 8940
CUN2 8950
CUN2 8960
CUN2 8970
CUN2 8980
CUN2 8990
CUN2 9000
CUN2 9010
CUN2 9020
CUN2 9030
CUN2 9040
CUN2 9050
CUN2 9060
CUN2 9070
CUN2 9080
CUN2 9090
CUN2 9100
CUN2 9110
CUN2 9120
CUN2 9130
CUN2 9140
CUN2 9150
CUN2 9160
CUN2 9170
CUN2 9180
CUN2 9190
CUN2 9200
CUN2 9210
CUN2 9220
CUN2 9230
CUN2 9240
CUN2 9250
CUN2 9260
CUN2 9270
CUN2 9280

RFACT = SUM CF THERMAL RESISTANCES NEGLECTING N/C GASSES
ROUT = EXTERNAL HEAT X-FER COEF
OD = TUBE OD
SHNF = LIQUID FILM HEAT X-FER COEF
SKBO = THERMAL CONDUCTIVITY OF OUTER LIQUID
STBAVE = AVG COOLANT TEMP
STCO = TEMP AT SURFACE OF OUTER LIQUID FILM
STFC = OUTER LIQUID FILM AVG TEMP
STSAT = SIM SATURATION TEMP
STWO = TUBE OUTER WALL TEMP
T = AVG LIQUID FILM TEMP IN RANKIN
TDIF = TEMP DIFF BETWEEN SAT STM AND AVE COOLANT TEMP
TSAT = SIM SAT TEMP IN RANKIN
UEST & UESTSV - VALUE OF UTEST AND UEST FROM PREVIOUS ITERATION
UTEST = OVERALL HEAT X-FER COEF.
XNU = NUSSELT NUMBER

ZZZ HET-12
-----
230 CONTINUE
M5=12
IF (HEICN.EQ.1.) WRITE (6,310) M5
STCO=STSAT-DLGF LM
T=(TSAT+STCO+459.69)/2.0
DTDP=1./(((6452.562+2.*837533.2/T)/(T**2)*EXP(14.15012--(6452.562+83
17533.2/T)/T))
BONE=-1.*(CJ*HSTO/ALMTD+(PGB*UTDP/ALMTD+1.)/RFACT)
CONE=CJ*HSTO/(ALMTD*RFACT)
UTEST=(-SQRT(BONE**2-4.*CONE)-BONE)/2.
DENOM=1.-UTEST*RFACT
IF (DENOM.GT.1.E-7) GO TO 240
HEFF=1./CONE
GO TO 250
240 HEFF=UTEST/DENOM
250 CONTINUE
ROUT=(HEFF+SHNF)/(HEFF*SHNF)
TDIF=STSAT-STBAVE
IF (TDIF.GT.0) GO TO 260
IF (IBT.EQ.0) WRITE (6,660) STSAT,STBAVE,LJ,IR
IBT=1
TDIF=.1
CONTINUE
260 DELTAU=TDIF*UTEST
DLFLM=DELTAU/SHNF
STCO=STSAT-DGFLM
STWO=STCO-DLFLM

```



```

TB2=TB2C
AOC=PI*ALST*SDO
AO=AOC
WB=WBC
FDAVE=0.5
SDIC=SID(NROWS)
SDI=SDIC
AI=PI*ALST*SDI
SHWINV=SHWINC
ENHI=ENHIC
ENHO=ENHOC
ENHF=ENHFC
STBIR=STBI+459.69
PTLIM=PSATFN(STBIR)
SMTBIC=0.0
SMWBC=0.0
SMTB2C=0.0
SUMQC=0.0
WTSTC=0.
IKOWC=0.
WSOUT=0.
SBFLOC=0.
SBRATC=0.
TBDRYC=0.
AMLSCC=WNCC/AMWNC
AXOC=SDC*HNOC*ALST*(SDDMIN-1.0)
SVGEXT=VGFN(TSATX,PMXEXT)
SVNCEX=10.729*TSATX/(AMWNC*(PMXEXT))
AMLT=AMLSEX+AMLSCC
SVMXEX=1.0/(AMLSEX/(AMLT*SVGEXT)+AMLSCC/(AMLT*SVNCEX))
VELC(1)=(WSEXT+WNCC)*SVMXEX/(AXOC*3600.0)

VL CMAX=VELC(1)
G2=288.0*SG*SVMXEX
IF (VELC(1)-VELEXT) 10,10,20
A2DA1=VELC(1)/VELEXT
DELPCT=((1.0-A2DA1)*VELEXT**2)/G2
GO TO 50
A2DA1=VELEXT/VELC(1)
IF (A2DA1-0.715) 40,40,30
DELPCT=(0.75*(1.0-A2DA1)*VELC(1)**2)/G2
GO TO 50
DELPCT=(0.4*(1.25-A2DA1)*VELC(1)**2)/G2
PMIXC(1)=PMXEXT-DELPCT-(VELC(1)**2-VELEXT**2)/G2
PSATC(1)=PMIXC(1)*AMLSEX/(AMLSEX+AMLSCC)

RECALCULATE THE AREA RATIO (THE RATIO OF TUBE X-SECTIONAL
AREA TO TTUBE SHEET AREA) TO INCLUDE THE COOLER

```

C

10

20

30

40

50

C

C

C


```

C          SECFLG=FLOAT(I SEC)
C          TSAREA=PI*(BNDRAD**2)*SECFLG*SECWID/360.
C          AHOLESA=ARATIO*TSAREEA
C          AHOLESA=AHOLESA+(PI*SDO**2)*TNOC/4.
C          TSAREA=TSAREEA+HTCLR*HNOC*SDC*SDOC
C          ARATIO=AHOLESA/TSAREEA
C
C          CALCULATE THE COOLER VOLUME
C          VOLC=HTCLR*SDDC*SDOC*HNOC*ALST
C
C          CHECK TO SEE IF INITIAL PRESSURE LOSSES DROP THE PRESSURE
C          BELOW PTLIM
C          IF (PSATC(1).GT.PTLIM) GO TO 80
C          SUMQC=0.
C          NRWSP4=IVNOC+1
C          CONSLA=PTLIM/FLOAT(NRWSP4)
C          CONSLB=PMXEXT/FLOAT(NRWSP4)
C          CUMDPC(1)=PMIX1-PMXEXT
C          PSATC(1)=PTLIM
C          PMIXC(1)=PMXEXT
C          QDAC(1)=0.
C          DELPC=PMXEXT
C          UNC(1)=700.
C          WSC(1)=WSEXT
C          WCNDC(1)=0.
C
C          DO 60 M=2,NRWSP4
C          WSC(M)=WSEXT
C          WCNDC(M)=0.
C          PMIXC(M)=PMIXC(M-1)-CONSLB
C          PSATC(M)=PSATC(M-1)-CONSLA
C          CUMDPC(M)=PMIX1-PMIXC(M)
C          CONTINUE
C
C          DO 70 M=2,IVNOC
C          UNC(M)=700.
C          QDAC(M)=0.
C          ALMTDC(M)=0.
C          CONTINUE
C          SMWBC=SMWBC+WB*TNOC
C          SMTBIC=WB*TNOC*STBI
C          SMTB2C=SMTBIC
C          EXITFC=1.
C          RETURN
C

```



```

80      PAL=PSATC(1)
      TAL=TSATFN(PAL)
      STSATC(1)=TAL-459.69
      WSC(1)=WSEXT
      CHECK TO SEE IF INLET STEAM TO COOLER IS 0
      IF (WSC(1).GT.1.) GO TO 90
      ENTER DUMMY VALUES IN COOLER VARIABLES
      CREATE NEGATIVE FLOW PARAMETER FOR USE IN COMPUTING EXITFRC
      TDRYC=TNOC
      SBRATC=WCNDAY
      SBFLOC=SBRATC*TDRYC
      WSC(1)=1.
      IROWC=3
      VG=VGFN(TAL,PMIXC(1))
      AMLSSC=WSC(1)/18.015
      VNC=10.729*TAL/(AMWNC*PMIXC(1))
      VMIX=1.0/((AMLSSC/(VG*(AMLSSC+AMLSCC)))+(AMLSSC/(VNC*(AMLSSC+AMLSCC+AMLSCC+1C))))
      VELC(1)=(WSC(1)+WNC)*VMIX/(AXOC*3600.0)
      NFC=(VNC+1.)/2.0
      ANFC=NFC
      IDT=0
      JRC=0
      IBT=0
      DO 450 IK=1,IVNOC
      I=IK
      LJ=7
      ROW BY ROW CALCULATION A LA HETTRN
      STSAT=STSATC(1)
      ANF=ANFC
      WS=WSC(1)
      WNC=WNC
      AXO=AXOC
      IR=I
      DATA HEFF,UEST,DGFLM,DLFLM/2100.,700.,0.05,2.0/
      DATA R2,E2OK/ 3.617,3.996,3.838,97.0,109.0,140.75/
      DATA R2/3.617,3.996,3.838/
      K=0
      RC=WNC/(WS+WNC)

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CON32170
CON32180
CON32190
CON32200
CON32210
CON32220
CON32230
CON32240
CON32250
CON32260
CON32270
CON32280
CON32290
CON32300
CON32310
CON32320
CON32330
CON32340
CON32350
CON32360
CON32370
CON32380
CON32390
CON32400
CON32410
CON32420
CON32430
CON32440
CON32450
CON32460
CON32470
CON32480
CON32490
CON32500
CON32510
CON32520
CON32530
CON32540
CON32550
CON32560
CON32570
CON32580
CON32590
CON32600
CON32610
CON32620
CON32630
CON32640

```



```

TSAT=TSAT+459.69
GMAX=(WS+WNC)/AXO
PSAT=PSATFN(TSAT)
AMOLSS=WS/18.015
AMOLSC=AMOLSS+AMOLSC
AMOLT=AMOLSS+AMOLSC
AMWAV=(WS+WNC)/AMOLT
RMWNC=SQRT(AMWNC)*AMOLSC/AMOLT
RMWSC=4.2444*AMOLSS/AMOLT
AVIS=(SMUFN(TSAT)*RMWSC+AMUFN(TSAT,JGAS)*RMWNC)/(RMWSC+RMWNC)
HSIS=(HFGFN(TSAT))
BB=WB/(PI*SDI*SDI/4.)
STBAVE=(STBI+TB2)/2.0
DELFLLM=DLFLLM
DLGFLM=DLGFLM
STWO=STSAT-DELFLM-DLGFLM
STFO=(TSAT+STWO)/2.0
XNRE=SDO*(WS+WNC)/(AXO*3600.*AVIS)
IF (XNRE.GE.100.) GO TO 100
CONTINUE
COLBJ=EXP(0.53883-0.544*ALOG(XNRE))
IF (WNCIR.EQ.0.) GO TO 110
PATM=PSAT/14.7
PGB=PSAT*(AMOLSC/AMOLSS)
TSATK=TSAT/1.8
CALL DFSVTY (TSATK,PATM,18.015,AMWNC,2.655,R2(JGAS),363.,E2OK(JGAS
1),0.,0.,DG,DGG)
C****
C 0.258 CONVERTS CM2/S TO FT2/HR AND 3600. CONVERTS VISCOSITY TO
C LB/F-HR FROM LB-F-S. 0.258 * 3600. = 928.8
C****
XNSCH=(AVIS*TSAT*10.73/((PSAT+PGB)*AMWAV*DG))*928.8
CJ=COLBJ*GMAX/(XNSCH*0.666667)
TB2=STSAT-(TSAT-STBI)/EXP(UEST*AO/(WB*CPFN(CBI,STBAVE)))
STBAVE=(STBI+TB2)/2.
SKBO=SKBFN(0.,STFO)
SKB=SKBFN(CBI,STBAVE)
BMU=BMUFN(CBI,STBAVE)
SHBI=CPFN(CBI,STBAVE)
XNREB=SDI*BB/BMU
XNPRB=SHBI*BMU/SKB
SHI=0.023*(XNREB)*0.8*XNPRB**0.33333*(SKB/SDI)*EIH
RIN=AG/(SHI*AI)
BNF=ANF
IF (ANF.LE.1.0) BNF=1.0
SHMK=0.725*(SKBO**3*ROEFN(0.0,STFO)**2*HFGFN(STFO)*41.975040.0/(BM
1UFN(0,STFO)*SDO*DELFLLM))**0.25*ENHO

```



```

C*****
HOMCI=STSAT-STBI
HIMCO=STSAT-TB2
C
C IF (HIMCO.LT.1.E-15) WRITE (6,530) HOMCI,HIMCO,LJ,IR
C IF (HIMCO.LT.1.E-15) HIMCO=1.E-15
C
C RODT=HCMCI/HIMCO
C
C IF INLET/OUTLET DELTA T RATIO IS NEAR 1.0, ALMTD = ARITH AVE
C IF (RODT.GT.1.1) GO TO 170
C ALMTD=0.5*(HOMCI+HIMCO)
C IF (JRC.EQ.0) WRITE (6,540) STSAT,STBI,TB2,LJ,IR
C JRC=1
C GO TO 180
170 ALMTD=(HOMCI-HIMCO)/ALOG(RODT)
180 CONTINUE
C IF (WNCIR.NE.0.) GO TO 190
C ROUT=1./SHNF
C UTEST=1./RFACT
C XNU=SHNF*SDO/SKBO
C IF (ABS(UEST-UTEST)/UTEST.LT..001) GO TC 280
C UESTSV=UEST
C UEST=UTEST
C IF (K.GT.4) UEST=0.5*(UTEST+UESTSV)
C STWO=STSAT-(STSAT-STBAVE)*UTEST/SHNF
C STCO=STSAT
C STFO=(STCO+STWO)/2.
C DGFLM=0.
C DLFLM=STCO-STWO
C DELFLM=DLFLM
C K=K+1
C IF (K.LT.20) GO TO 110
C WRITE (6,510) UESTSV,UTEST
C GO TO 280
190 CONTINUE
C STCO=STSAT-DLGFLM
C T=(TSAT+STCO+455.69)/2.0
C DTP=1./(((6452.562+2.*837533.2/T)/T**2)*EXP(14.15012-(6452.562+83
17533.2/T)/T))
C BONE=-1.*(CJ#HSTO/ALMTD+(PGB*DTDP/ALMTD+1.)/RFACT)
C CONE=CJ#HSTO/(ALMTD*RFACT)
C TTERM1=BCONE**2
C TTERM2=4.*CONE
C TTERM=TTERM2/TTERM1
C IF (TTERM.LT.0.999999) GO TO 200
C UEST=-BONE/2.
C GO TO 220

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```

CON33610
CON33620
CON33630
CON33640
CON33650
CON33660
CON33670
CON33680
CON33690
CON33700
CON33710
CON33720
CON33730
CON33740
CON33750
CON33760
CON33770
CON33780
CON33790
CON33800
CON33810
CON33820
CON33830
CON33840
CON33850
CON33860
CON33870
CON33880
CON33890
CON33900
CON33910
CON33920
CON33930
CON33940
CON33950
CON33960
CON33970
CON33980
CON33990
CON34000
CON34010
CON34020
CON34030
CON34040
CON34050
CON34060
CON34070
CON34080

```


200	IF ((4.*CONE).GT.(BONE**2)) UTEST=0.	CON34090
	IF ((4.*CONE).GT.(BONE**2)) GO TO 210	CON34100
210	UTEST=(-SQRT(BONE**2-4.*CONE))/2.	CON34110
220	DENOM=1.-UTEST*RFAC	CON34120
	CONTINUE	CON34130
	IF (DENOM.GT.1.E-7) GO TO 230	CON34140
	HEFF=1.0E8	CON34150
230	GO TO 240	CON34160
240	HEFF=UTEST/DENOM	CON34170
	CONTINUE	CON34180
	ROUT=(HEFF+SHNF)/(HEFF*SHNF)	CON34190
	TDIF=STSAT-STBAVE	CON34200
	IF (TDIF.GT.0) GO TO 250	CON34210
	IF (IBT.EQ.0) WRITE (6,530) STSAT,STBAVE,LJ,IR	CON34220
	IBT=1	CON34230
	TDIF=.1	CON34240
250	CONTINUE	CON34250
	DELTAU=TDIF*UTEST	CON34260
	DGFLM=DELTAU/HEFF	CON34270
	DLFLM=DELTAU/SHNF	CON34280
	STCO=STSAT-DGFLM	CON34290
	STWO=STCO-DLFLM	CON34300
	STFO=(STCO+STWO)/2.	CON34310
	XNU=SDO/(SKB0*ROUT)	CON34320
	IF (ABS(UTEST-UEST)/UTEST.LT..001) GO TO 260	CON34330
	DELFLM=DLFLM	CON34340
	DLGFLM=DGFLM	CON34350
	UESTSV=UEST	CON34360
	UEST=UTEST	CON34370
	K=K+1	CON34380
	IF (K.GT.4) UEST=0.5*(UTEST+UESTSV)	CON34390
	IF (K.LT.20) GO TO 110	CON34400
	WRITE (6,510) UESTSV,UTEST	CON34410
	GO TO 280	CON34420
260	IF (ABS(DELFLM-DLFLM).LT..01) GO TO 270	CON34430
	DELFLM=DLFLM	CON34440
	DLGFLM=DGFLM	CON34450
	K=K+1	CON34460
	IF (K.LT.20) GO TO 110	CON34470
	WRITE (6,520) DELFLM,LJ,IR	CON34480
270	IF (ABS(DLGFLM-DGFLM).LT..001) GO TO 280	CON34490
	DLGFLM=DGFLM	CON34500
	K=K+1	CON34510
	IF (K.LT.20) GO TO 110	CON34520
	WRITE (6,500) DLGFLM,UEST,UTEST,LJ,IR	CON34530
280	VNF=UTEST	CON34540
	VP SHH=TB2	CON34550
	XHEFF=HEFF	CON34560


```

310 STSATC(I+1)=STFC
TSATMN=STSATC(I+1)+459.69
PMIN=PSATFN(TSATMN)
C
C ENTER DUMMY VALUE FOR NEXT ROW WSC
C WSC(I+1)=1.
C QGC=(WSC(I)*CPSFN(TAL)/18.015+WNCC*CPAFN(TAL,JGAS)/AMWNC)*(TAL-TSA
1TMN)
C QTC=QGC+WCNDC(I)*HFG
320 IF (STSATC(I)-TB2) 320,320,330
330 TB2=STSATC(I)-0.1
C CONTINUE
C IF (STSATC(I+1).GT.STBI) GO TO 340
340 ALMTDC(I)=ALMTDC(I-1)
C GO TO 350
C CONTINUE
350 ALMTDC(I)=(STSATC(I+1)-STBI)-(STSATC(I)-TB2))/ALOG((STSATC(I+1))-S
1TB I)/(STSATC(I)-TB2))
C CONTINUE
C UNC(I)=QTC/(AO*HNOC*ALMTDC(I))
C AMLSSC=WSC(I+1)/18.015
C PMIXC(I+1)=PMIXC(I)-DELTPC
C PSATC(I+1)=(PMIXC(I+1)*AMLSSC)/(AMLSSC+AMLSSC)
C CHECK TO SEE IF STEAM CORRECTIONS FOR DRY TUBES DROVE PSATC TOO LO
C IF ((IRGWC.GT.0).AND.(PSATC(I+1).LE.PTLIM)) WRITE (6,560) I
C IF ((IROWC.GT.0).AND.(PSATC(I+1).LE.PTLIM)) PSATC(I+1)=PSATC(I)
C CHECK FOR PRESSURE DROPPING BELOW PTLIM
C IF (PSATC(I+1).GT.PTLIM) GO TO 390
C CALCULATE PARAMETERS FOR CURRENT ROW
C PSATC(I)=PTLIM
C SUMQPC(I)=PMIX1-PMIXC(I)
C SUMQC=SUMQC+UNC(I)*AO*ALMTDC(I)*HNOC
C QOAC(I)=UNC(I)*ALMTDC(I)
CUN35050
CUN35060
CUN35070
CUN35080
CUN35090
CUN35100
CUN35110
CUN35120
CUN35130
CUN35140
CUN35150
CUN35160
CUN35170
CUN35180
CUN35190
CUN35200
CUN35210
CUN35220
CUN35230
CUN35240
CUN35250
CUN35260
CUN35270
CUN35280
CUN35290
CUN35300
CUN35310
CUN35320
CUN35330
CUN35340
CUN35350
CUN35360
CUN35370
CUN35380
CUN35390
CUN35400
CUN35410
CUN35420
CUN35430
CUN35440
CUN35450
CUN35460
CUN35470
CUN35480
CUN35490
CUN35500
CUN35510
CUN35520

```



```

C
SMWBC=SMWBC+WB*HNOC
SMTBIC=SMTBIC+WB*HNOC*STBI
SMTB2C=SMTB2C+WB*HNOC*TB2
CON35530
CON35540
CON35550
CON35560
CON35570
CON35580
CON35590
CON35600
CON35610
CON35620
CON35630
CON35640
CON35650
CON35660
CON35670
CON35680
CON35690
CON35700
CON35710
CON35720
CON35730
CON35740
CON35750
CON35760
CON35770
CON35780
CON35790
CON35800
CON35810
CON35820
CON35830
CON35840
CON35850
CON35860
CON35870
CON35880
CON35890
CON35900
CON35910
CON35920
CON35930
CON35940
CON35950
CON35960
CON35970
CON35980
CON35990
CON36000

C
NRWSP4=IVNOC+1
RTG=FLOAT(NRWS P4-I)
NRWSP5=I+1
CONST=PMIXC(I)/RTG
CONST2=PTLIM/RTG
CON35530
CON35540
CON35550
CON35560
CON35570
CON35580
CON35590
CON35600
CON35610
CON35620
CON35630
CON35640
CON35650
CON35660
CON35670
CON35680
CON35690
CON35700
CON35710
CON35720
CON35730
CON35740
CON35750
CON35760
CON35770
CON35780
CON35790
CON35800
CON35810
CON35820
CON35830
CON35840
CON35850
CON35860
CON35870
CON35880
CON35890
CON35900
CON35910
CON35920
CON35930
CON35940
CON35950
CON35960
CON35970
CON35980
CON35990
CON36000

C
CALCULATE PARAMETERS FOR THE REMAINING ROWS
DO 370 M=NRWSP5,NRWSP4
  QQAC(M)=0.
  WSC(M)=WSC(M-1)
  WCNDC(M)=0.
  STSATC(M)=STSATC(M-1)
  PSATC(M)=PSATC(M-1)-CONST2
  PMIXC(M)=PMIXC(M-1)-CONST
  VELC(M)=VELC(M-1)
  CONTINUE
370
C
DO 380 IFX=NRWSP5, IVNOC
  UNC(IFX)=UNC(IFX-1)
  CUMDPC(IFX)=PMIX1-PMIXC(IFX-1)
  SMTBIC=SMTBIC+WB*HNOC*STBI
  SMTB2C=SMTBIC
  QQAC(IFX)=0.
  SMWBC=SMWBC+WB*HNOC
  ALMTDC(IFX)=0.
380
C
DE LPC=PMIXC(1)-PMIXC(NRWSP4)
GO TO 460
C
C
C
390
C
PAL=PSATC(I+1)
IF (PAL.GT.0.) GO TO 400
WRITE (6,550) PMIXC(I),I
CONTINUE
400
C
TSATC=TSATFN(PAL)
STSATC(I+1)=TSATC-459.69
IF (ABS(TSATC-TAL).LT.0.001) TSATC=TAL-0.001
WCNDP=(UNC(I)*AD*HNOC*ALMTDC(I)-(WSC(I)*CPSFN(TSATC)/18.015+WNCC*C
1PAFN(TSATC,JGAS)/AMWNC)*(TAL-TSATC))/HFG
IF (ABS(WCNDP/WCNDC(I)-1.0)-.005) 430,410,410
WCNDC(I)=WCNDP
JY=JY+1
410
C
IF (JY-50) 290,290,420

```


420	WRITE (6,540) I	CUN36010
430	CONTINUE	CUN36020
	WCNDC(I)=WGNDP	CUN36030
	WSC(I+1)=WSC(I)-WCNDC(I)	CUN36040
	AMLSSC=WSC(I+1)/18.015	CUN36050
	PSATC(I+1)=(PMIXC(I+1)*AMLSSC)/(AMLSSC+AMLSSC)	CUN36060
	PAL=PSATC(I+1)	CUN36070
	TSATC=TSATFN(PAL)	CUN36080
440	STSATC(I+1)=TSATC-459.69	CUN36090
	VG=VGFN(TSATC,PMIXC(I+1))	CUN36100
	VN C=10.729*TSATC/(AMWNC*PMIXC(I+1))	CUN36110
	VMIX=1.0/((AMLSSC/(VG*(AMLSSC+AMLSSC)))+(AMLSSC/(VNC*(AMLSSC+AMLSSC	CUN36120
	1C)))	CUN36130
	VELC(I+1)=(WSC(I+1)+WNCC)*VMIX/(AXDC*3600.0)	CUN36140
	SMTBIC=SMTBIC+WB*HNOC*STBI	CUN36150
	SMWBC=SMWBC+WB*HNOC	CUN36160
	SMTB2C=SMTB2C+WB*HNOC*TB2	CUN36170
	SUMQC=SUMQC+UNC(I)*AO*ALMTDC(I)*HNOC	CUN36180
	QOAC(I)=UNC(I)*ALMTDC(I)	CUN36190
	CUMDPC(I)=PMIX1-PMIXC(I)	CUN36200
	TAL=TSATC	CUN36210
	ANFC=ANFC-1.0	CUN36220
450	CONTINUE	CUN36230
460	DELP C=PMIXC(1)-PMIXC(IVNOC+1)	CUN36240
C	CONTINUE	CUN36250
C	CHECK TG SEE IF INLET STEAM FLOW WAS 0	CUN36260
	IF (IROWC*LT.2) GO TO 480	CUN36270
	NRWSP4=IVNOC+1	CUN36280
	SUMQC=0.	CUN36300
	DO 470 M=1,NRWSP4	CUN36310
	QOAC(M)=0.	CUN36320
	ALMTDC(M)=0.	CUN36330
	WSC(M)=0.	CUN36340
470	WCNDC(M)=0.	CUN36350
	EXITFC=-999	CUN36360
	WTSTC=FLOAT(IVNOC)	CUN36370
	RETURN	CUN36380
C	IF COOLER TUBES WENT DRY CREATING A NEGATIVE EXITFC	CUN36390
C		CUN36400
C		CUN36410
480	IF (WTSTC*GE.0.) GO TO 490	CUN36420
	EXITFC=-(SBFLOC)/WSC(1)	CUN36430
	RETURN	CUN36440
C	EXITFC=WSC(I+1)/WSC(1)	CUN36450
490	WSOUT=WSC(I+1)	CUN36460
		CUN36470
		CUN36480


```

2VP SHHC(100),HEFFC(100),UNC(100),WCNDC(100),GFLWC(100),QDAC(100),C CON3 6970
3UMDPC(100) CON3 6980
COMMON /COOL/ IVNOC,HNOC,TNOC,TSATEX,PMXEXT,AMLSEX,WSEXIT,VELEXIT,T CON3 6990
182C,ENHIC,ENH3C,ENHFC,SHWINC,WBC CON3 7000
COMMON /OUT2A/ RCONRR,RSPP,PHPCON,VOLIC CON3 7010
DIMENSION AWTST(15) CON3 7020
DELPVE=0. CON3 7030
AOT=0. CON3 7040
VEL3=0. CON3 7050
CAWT=0. CON3 7060
UBARW=0. CON3 7070
VWBAR=0. CON3 7080
SECFLG=FLOAT(I SEC) CON3 7090
DO 10 I=1,NROWS CON3 7100
OD=OD0 I-DELOD*FLOAT(I-1) CON3 7110
CAWT=CAWT+(CD*OD/144.-SID(I)*SID(I))*TBNPR(I) CON3 7120
AO=3.14159*OD*ALST*TBNPR(I)/12. CON3 7130
VWBAR=VWBAR+VW(I) CON3 7140
AOT=AOT+AO CON3 7150
BUNWT=PI*CAWT*ALST*TUBESW*SECFLG/4. CON3 7160
AOT=AO T*SECFLG CON3 7170
DO 30 J=1,I SEC CON3 7180
DO 20 I=1,NROWS CON3 7190
UBARW=UBARW+UN(I,J)*TBNPR(I) CON3 7200
VEL3=VEL3+VEL(NROWS,J) CON3 7210
DELPVE=DELPVE+DELP(J) CON3 7220
CONTINUE CON3 7230
UBARW=UBARW/TNO CON3 7240
VWBAR=VWBAR/FLOAT(NROWS) CON3 7250
VEL3=VEL3/SECFLG CON3 7260
DELPVE=DELPVE/SECFLG CON3 7270
ADTCND=SUMQ/(UBARW*AOT) CON3 7280
STSATX=TSATEX-459.69 CON3 7290
TDROP1=STSAT(1,1)-STSATX CON3 7300
RSPI=RSPP*12. CON3 7310
CON3 7320
CON3 7330
BNDIAM=BNDRAD*2. CON3 7340
BL=ALST/2. CON3 7350
VODID=RADINS*2. CON3 7360
AVTBI=SMIBI/SMWB CON3 7370
AVTB2=SMIB2/SMWB CON3 7380
HDLSS=DELP*144./62.366 CON3 7390
CON3 7400
CON3 7410
CON3 7420
CON3 7430
CON3 7440

```

THERE IS THE POSSIBILITY THAT THE AVERAGE INLET COOLANT TEMP
 (AVTBI) EQUALS THE AVERAGE OUTLET COOLANT TEMP (AVTB2). WHEN
 THE CONDENSER IS COMFIGURED SUCH THAT NO HEAT IS REMOVED FROM
 THE STEAM

CON37450
CON37460
CON37470
CON37480
CON37490
CON37500
CON37510
CON37520
CON37530
CON37540
CON37550
CON37560
CON37570
CON37580
CON37590
CON37600
CON37610
CON37620
CON37630
CON37640
CON37650
CON37660
CON37670
CON37680
CON37690
CON37700
CON37710
CON37720
CON37730
CON37740
CON37750
CON37760
CON37770
CON37780
CON37790
CON37800
CON37810
CON37820
CON37830
CON37840
CON37850
CON37860
CON37870
CON37880
CON37890
CON37900
CON37910
CON37920

```

C      IF (AVTB1.EQ.AVTB2) DTCND2=0.
      IF (AVTB1.EQ.AVTB2) UPCOND=UBARW
      IF (AVTB1.EQ.AVTB2) GO TO 40

C      DTCND2=(AVTB2-AVTB1)/ALOG((STSAT1-AVTB1)/(STSAT1-AVTB2))
      UPCOND=SUMQ/(AOT*DTCND2)
      WRITE (6,60)
      WRITE (6,80)
      WRITE (6,90)
      WRITE (6,100)
      WRITE (6,90)
      WRITE (6,80)
      WRITE (6,70)
      WRITE (6,110)
      WRITE (6,120)
      WRITE (6,130)
      WRITE (6,160)
      WRITE (6,140) EXITFR

      UBARW,ADTCND,DELPVE,TDROP1,VEL2,VEL3
      SUMQ
      WSI
      AOT
      EXITFR

C      DO 50 I=1,ISEC
      IF (WTST(I).LT.0.) AWTST(I)=-WTST(I)
      IF (WTST(I).LT.0.) WRITE (6,150) I,AWTST(I)
      CONTINUE

      WRITE (6,170) TNO,NOROWS
      WRITE (6,180) ALST ID,BNDIAM,BL
      WRITE (6,190) VOID ID,BNDIAM,BL
      IF (MDIAM.EQ.1) WRITE (6,200) SIDO,ODOI
      IF (MDIAM.EQ.2) WRITE (6,210) SIDI,ODII
      IF (MDIAM.EQ.2) WRITE (6,220) SIDO,ODOI
      IF (MPITCH.EQ.1) WRITE (6,230) SDDO
      IF (MPITCH.EQ.2) WRITE (6,240) SDDI,SDDO
      IF (XW2.GT.0.) WRITE (6,250) XW2
      IF (XW2.LE.0.) WRITE (6,260) XW1
      WRITE (6,270) RSPI
      WRITE (6,280) VCLLC,VOL2,BUNWT
      WRITE (6,290) AVTB1
      WRITE (6,300) AVTB2
      WRITE (6,320) SMWB
      WRITE (6,330) VWBAR
      WRITE (6,310) HDLOSS,PHPCON
      WRITE (6,340) DTCND2
      WRITE (6,350) UPCOND
      WRITE (6,360) PFILL
      IF (PRCCLR.LE.0) WRITE (6,370) ARATIO
      IF (PRCCLR.GT.0.) CALL OUT2C
      RETURN

```



```

60  FORMAT (1H1)
70  FORMAT (1H0,31X,29H***** CONDENSER *****//)
80  FORMAT (1H,15X,16X,29H*****//)
90  FORMAT (1H,15X,16X,1H*,27X,1H*)
100  FORMAT (1H,15X,16X,29H* CONDIP SUMMARY OF RESULTS *)
110  FORMAT (1H,10X,56H*OVERALL U LOG-MEAN PRESSURE DROP DEG.F INLET
1 VELOCIT/10X,57H-SQ.FT. DEG.F PSIA 2,1P3E10.2//)
2 T/SEC/11X,57H-SQ.FT. DEG.F PSIA 2,1P3E10.2//)
3 TLET//12X,57H-SQ.FT. DEG.F PSIA 2,1P3E10.2//)
120  FORMAT (1H,10X,12HHEAT DUTY ,8X,1PE11.4,2X,6H8TU/HR)
130  FORMAT (1H,10X,18HSTEAM TO BUNDLE ,2X,1PE11.4,2X,5H1B/HR)
140  FORMAT (1H,10X,13HEXIT FRAC. ,7X,1PE11.4//)
150  FORMAT (1H,10X,10HIN SECTOR ,12,2X,29HTHE FRACTION OF DRY ROWS WA
1S ,F10.6)
160  FORMAT (1H,10X,19HHT. TRF. SURF. AREA,1X,1PE11.4,2X,6HSQ.FT.)
170  FORMAT (1H,1X/10X,16HBUNDLE DIAMETER,,3X,2HFT,7X,F10.2,1X,6HTUBES
1,15,1X,4HROWS)
180  FORMAT (1H,12X,6HINSIDE,4X,7HOUTSIDE,13X,10HTUBES ARE ,F5.2,1X,7H
1FT LONG)
190  FORMAT (1H,12X,F5.2,5X,F5.2,14X,11HBUNDLE IS ,F5.2,1X,7HFT LONG/
1)
200  FORMAT (1H,10X,20HTUBE INSIDE DIAMETER,15X,F7.4,2X,6HINCHES/1H,1
10X,21HTUBE OUTSIDE DIAMETER,14X,F7.4,2X,6HINCHES)
210  FORMAT (1H,10X,30HTUBE INSIDE DIAMETER INNER ROW,6X,F7.4,2X,3HIN.
1/11X,31HTUBE OUTSIDE DIAMETER INNER ROW,5X,F7.4,2X,3HIN.//)
220  FORMAT (1H,11X,30HTUBE INSIDE DIAMETER OUTER ROW,5X,F7.4,2X,3HIN.
1/11X,31HTUBE OUTSIDE DIAMETER OUTER ROW,5X,F7.4,2X,3HIN.)
230  FORMAT (1H,10X,26HTUBE CIRCUMFERENTIAL PITCH,9X,F7.4)
240  FORMAT (1H,10X,36HTUBE CIRCUMFERENTIAL PITCH INNER ROW,5X,F7.4/1H
1,10X,36HTUBE CIRCUMFERENTIAL PITCH OUTER ROW,5X,F7.4)
250  FORMAT (1H,10X,14HTUBE THICKNESS,21X,F7.4,2X,6HINCHES)
260  FORMAT (1H,10X,34H RATIO OF TUBE THICKNESS TO TUBE OD,5X,F7.4)
270  FORMAT (1H,10X,28HSPACE BETWEEN COCENTRIC ROWS,7X,F7.4,2X,3HIN.//)
280  FORMAT (1H,10X,13HBUNDLE VOLUME,7X,1PE11.4,2X,6HCU.FT./11X,19HTUBE METAL WEIGHT
1E SPACE VOLUME ,1X,1PE11.4,2X,6HCU.FT./11X,19HTUBE METAL WEIGHT
2,1X,1PE11.4,2X,2H1B//)
290  FORMAT (1H,10X,19HINLET WATER TEMP ,5X,F5.2,1X,5HDEG.F)
300  FORMAT (1H,10X,20HOUTLET WATER TEMP ,4X,F9.2,1X,5HDEG.F)
310  FORMAT (1H,10X,20HHEAD LOSS,15X,F9.2,1X,6HFT H2O/11X,13HPUMPING POW
1WER,11X,F9.2,1X,2HHP//)
320  FORMAT (1H,10X,13HWATER FLOW ,12X,1PE11.4,1X,5H1B/HR)
330  FORMAT (1H,10X,17HWATER VELOCITY ,8X,1PE11.4,1X,6HFT/SEC//)
340  FORMAT (1H,10X,12H AVG. LMTD ,14X,F7.2,1X,5HDEG.F)
350  FORMAT (1H,10X,9H AVG. U ,17X,F7.2,1X,18H8TU/HR-SQ.FT-DEG.F//)
360  FORMAT (1H,10X,46HTHE PERCENTAGE OF THE OUTERMOST ROW FILLED IS ,
1F7.4//)
370  FORMAT (1H,10X,18HTHE AREA RATIO IS ,F7.4//)
END

```


CON38890
CON38900
CON38910
CON38920
CON38930
CON38940
CON38950
CON38960
CON38970
CON38980
CON38990
CON39000
CON39010
CON39020
CON39030
CON39040
CON39050
CON39060
CON39070
CON39080
CON39090
CON39100
CON39110
CON39120
CON39130
CON39140
CON39150
CON39160
CON39170
CON39180
CON39190
CON39200
CON39210
CON39220
CON39230
CON39240
CON39250
CON39260
CON39270
CON39280
CON39290
CON39300
CON39310
CON39320
CON39330
CON39340
CON39350
CON39360

```

VWBARC=VWBAR
IF (AVTBIC.EQ.AVTB2C) DTCOL2=0.
IF (AVTBIC.EQ.AVTB2C) UPCOOL=UBARWC
IF (AVTBIC.EQ.AVTB2C) GO TO 20

DTCOL2=(AVTB2C-AVTBIC)/ALOG((STSATX-AVTBIC)/(STSATX-AVTB2C))
UPCOOL=SUMQC/(AOTC*DTCOL2)

VALUES FOR OVERALL OUTPUT

AOTOT=AOTC+AOT
TNOT=TNC+TNCC
UBARQA=(UBARW*TNQ+UBARWC*TNOC)/TNQ
ADTOA=(SUMQC+SUMQ)/(UBARQA*AOTOT)
DELPQA=PMIXI-PMIXC(IVNOC+1)
TDRPOA=STSAT1-STSATC(IVNOC+1)
SUMQQA=SUMQC+SUMQ
WTQA=BUNWT*COOLWT
AVTB1A=(AVTB1*TNQ+AVTBIC*TNOC)/TNQ
AVTB2A=(AVTB2*TNQ+AVTB2C*TNOC)/TNQ
SMWBQA=SMWBC+SMWB
HDLQA=DELPQA*144./62.366
DTQA=(AVTB2A-AVTB1A)/ALOG((STSAT1-AVTB1A)/(STSAT1-AVTB2A))
UPQA=SUMQQA/(AOTOT*DTQA)
WRITE (6,40)
WRITE (6,50) UBARWC,ADTCLR,DELPVC,TDROP,VELC(1),VELC(IVNOC)
WRITE (6,60) SUMQC
WRITE (6,70) WSC(1)
WRITE (6,110) AOTC

IF (EXITFC.LT.-900.) WRITE (6,100)
IF (EXITFC.LT.-900.) GO TO 30

WRITE (6,80) EXITFC

IF (WTSTC.LT.0) WRITE (6,90) WTSTC

WRITE (6,120) TNOC,IVNOC
WRITE (6,130) HTCLR,WDCLR
WRITE (6,140) VCLC,COOLWT
WRITE (6,150) AVTBIC
WRITE (6,160) AVTB2C
WRITE (6,170) SMWBC
WRITE (6,180) VWBARC
WRITE (6,190) DTCOL2
WRITE (6,210) UPCOOL
WRITE (6,220)

```

C

C

C

C

C

20

C

C

C

30

C


```

10      WRITE (6,160) I,VW(I),A,OD,SDD,TBNPR(I)
      DO 30 J=1,ISEC
      WRITE (6,120) J
      DO 20 I=1,NCROWS
      DD=OD+I-DELOD*FLOAT(I-1)
      AU=3.14159*OD*ALST
      QU=UN(I,J)*ALMTD(I,J)*AQ
      WRITE (6,160) I,Q,CUMDP(I,J),ALMTD(I,J),RC(I,J),VNRE(I,J)
      CONTINUE
      DO 50 J=1,ISEC
      WRITE (6,130) J
      DO 40 I=1,NOROWS
      WRITE (6,160) I,UN(I,J),ROUT(I,J),HEFF(I,J),SHN(I,J),SHI(I,J)
      CONTINUE
      DO 70 J=1,ISEC
      WRITE (6,150) J
      DO 60 I=1,NOROWS
      WRITE (6,160) I,WS(I,J),WCND(I,J),STSAT(I,J),PSAT(I,J),PMIX(I,J)
      CONTINUE
      IF (PRCCCLR.LE.0.) GO TO 110
      WRITE (6,170)
      DO 80 I=1,IVNQC
      QC=UNC(I)*ALMTD(I)*AOC
      WRITE (6,200) I,QC,CUMDPC(I),ALMTDC(I),RCC(I),VNREC(I)
      CONTINUE
      WRITE (6,180)
      DO 90 I=1,IVNQC
      WRITE (6,200) I,UNC(I),ROUTC(I),HEFFC(I),SHNFC(I),SHIC(I)
      CONTINUE
      WRITE (6,190)
      DO 100 I=1,IVNQC
      WRITE (6,200) I,WSC(I),WCNDC(I),STSATC(I),PSATC(I),PMIXC(I)
      CONTINUE
      RETURN
      FORMAT (1H1///,15X,15X,28HROW BY ROW OUTPUT FOR SECTOR,13//,15X,3HCON4,0670
1ROW,4X,4HHEAT,5X,8HPRESSURE,3X,8HLOG MEAN,3X,8HRATIO UF,2X,10HSFEACON4,0680
2M SIDE/15X,3HNO.,2X,8HTRANSFER,5X,4HDROP,5X,9HTEMP DIFF,2X,7HN/C GCON4,0690
3AS,3X,12HREYNOLDS NO./15X,5X,8HPER TUBE,3X,7HPER ROW,14X,10HTO SIMCON4,0700
4 FLW/15X,6X,6HBTU/HR,6X,3HPSI,8X,5HDEG F/)
      FORMAT (1H1///,15X,5X,47HROW BY ROW HEAT TRANSFER COEFFICIENTS FORCON4,0720
1 SECTOR,13//,15X,23X,14HBTU/HR-FT*2-F//15X,3HROW,2X,8HOVER-ALL,3X,8CON4,0730
2HXTURNAL,3X,8HEXTURNAL,3X,8HLIQ FILM,3X,8HLIQ FILM/)
      FORMAT (1H1///,15X,8HGAS FILM,3X,8HLIQ FILM,3X,8HCONDENSER PARAMETERCON4,0750
3HTUBE,7X,4HTUBE,5X,8HGAS FILM,3X,8HCONDENSER PARAMETERCON4,0760
4 HTUBE,7X,4HTUBE,5X,8HGAS FILM,3X,8HCONDENSER PARAMETERCON4,0770
1S//15X,3HROW,3X,7HCOOLANT,2X,12HCOOLANT MASS,3X,4HTUBE,7X,4HTUBE,4CON4,0780
2X,9HNUMBER OF/15X,3HNO.,3X,21HVELOCITY FLOW/TUBE,4X,2HOD,8X,5HPCON4,0790
3ITCH,3X,9HTUBES PER/15X,17X,6HMBM/HR,6X,6HINCHES,14X,11HROW PER SECON4,0800
4C/)

```



```

10 IMIX=1
20 CONTINUE
C INERT GAS IS CO2
C ** HEAT CAPACITY FOR MW = 40.1 BTU/LB-MOL - DEG R
      GAS2=0.209*40.1
      CPAFN=GAS2
      IF (IMIX.EQ.0) RETURN
30 CONTINUE
C INERT GAS IS AIR
C AIR CP, BTU / LB MOLE-DEGREE RANKIN
      GAS1=7.139-0.9884E-3*T+0.1393E-5*T**2-0.3367E-9*T**3
      CPAFN=GAS1
      IF (IMIX.EQ.0) RETURN
C INERT GAS IS MIXTURE OF AIR & CO2
      GAS3=(GAS1+GAS2)/2.0
      CPAFN=GAS3
      RETURN
      END
C
C
C FUNCTION CPFN (C,T)
      IF (C-0.005) 20,20,30
10 IF (C-0.005) 20,20,30
C *** EQUATION SPECIFIC HEAT FOR PURE WATER
20 CP=1.0121559+(-0.24618473E-3+0.1028215E-5*T)*T
      GO TO 40
C *** EQUATION SPECIFIC HEAT FOR BRINE
30 CP=.96946859+(2.*(0.0010404965)*T)-(.91199294*C)+(2.*(-.000648296
      159)*C*T)+(-1.555779*(C**2))+2.*(.0076721465)*(C**2)*T)+(6.7981008
      2*(C**3))+2.*(-.012610354)*(C**3)*T)
40 CPAFN=CP
      RETURN
      END
C
C
C FUNCTION CPSFN (T)
C ***
C CPSFN CALCULATES THE HEAT CAP OF STEAM IN BTU/LB-MOL-R GIVEN T IN
C DEG R. EQUATIONS FROM H. NORITAKE, BASED ON TABULATED VALUES IN
C NASA TR-R-132
C ***
      CP SFN=(7.838-(.2531E-3-(.2892E-6-.7693E-10*T)*T)*T)
      RETURN
      END
C
C
C FUNCTION HFGFN (T)
      HFGFN=1093.88-0.5703*T+.00012819*T**2-.0000008824*T**3
      RETURN

```



```

C
C
CCC
C
CCC
END
SUBROUTINE PRSDRP (TSAT,VMIX,WS,WNC,AXC,SDO,SF,DELPTP,ENHF)
PRSDRP  REBUILT ON 9-16-69 TO USE EQUATION FOR SF
SG=32.174
GSTAR=(WS+WNC)/(AXO*3600.)
ANRE=(SCQ*GSTAR)/SMUFN(TSAT)
SF=(0.102+52.2/ANRE)*ENHF
DELPTP=SF*GSTAR**2*VMIX/(72.0*SG)
RETURN
END
C
C
FUNCTION PSATFN (T)
PSATFN=2.718**((14.150119-(6452.5621/T)-(837533.21/T**2)))
RETURN
END
C
C
FUNCTION ROEFN (C,T)
DENSITY OF SALINE SOLUTION. RANGE OF DATA WAS 0 - 26 PERCENT
CONCENTRATION AND 40 - 300 DEGREES FARENHEIT.
ROEFN=0.62707172E2+0.49364088E2*C-(0.43555304E-2+0.32554667E-1*C+
10.46076521E-4-0.63240299E-4*C)*T)
RETURN
END
C
C
FUNCTION SKBFN (C,T)
THERMAL CONDUCTIVITY OF SALINE SOLUTION. RANGE OF DATA
0 - 24 PERCENT CONCENTRATION AND 40 - 300 DEGREES FARENHEIT
SKBFN=(.30157913+.697989E-3*T-.12506E-5*T**2-.2072E-10*T**3)*(-.16
187109*C+1.)
RETURN
END
C
C
FUNCTION SMUFN (T)
SMUFN=1.0E-5*(0.122+(1.001E-3)*T+(2.892E-7)*T**2-(7.693E-11)*T**3)
RETURN
END
C
C
FUNCTION TSAIFN (P)
AA=(ALOG(P)-14.150119)

```



```

10  DET=6452.5621*2-(4.0*AA*837533.21)
    IF (DET) 10,20,20
    WRITE (6,60)
    CALL EXIT
20  X=(-6452.5621+SQRT(DET))/((2.0*AA)
    Y=(-6452.5621-SQRT(DET))/((2.0*AA)
    IF (X-Y) 30,40,40
30  TSATFN=Y
    GO TO 50
40  TSATFN=X
50  CONTINUE
    RETURN
60  FORMAT (1H1,37H SUBROUTINE TSATFN FINDS COMPLEX ROOTS)
    ENDC
    C
    C
    FUNCTION VGFN (T,P)
    X=ALOG(T/P)
    VGFN=EXP((.103758E-2*X-.0177861)*X+1.10267)*X-.72240)
    RETURN
    ENDC
    C
    C
    SUBROUTINE SWITCH (A,N)
    DIMENSION A(N)
    NN=N/2
    K=N+1
    DO 10 I=1,NN
    T=A(I)
    A(I)=A(K-I)
    A(K-I)=T
    CONTINUE
    RETURN
    ENDC
10

```


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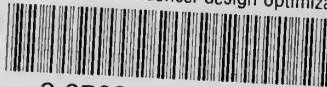
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